Table 1 to paragraph (a)(1)(iv)— Continued

Column A variety	Column B maturity guide
Fayette	1
Fire Red	1
First Lady	D
Flamecrest	1
Flavorcrest	G
Flavor QueenFlavor Red	H G
Franciscan	G
Goldcrest	H
Golden Crest	Н
Golden Lady	F
Honey Red	G
John Henry	J
July Elberta	C G
July Lady June Lady	G
June Pride	J
June Sun	Н
Kearney	1
Kern Sun	H
Kingscrest	Н
Kings Lady	
Kings Red Lacey	i
Mary Anne	G
May Crest	Ğ
May Lady	G
May Sun	I
Merrill Gem	G
Merrill Gemfree O'Henry	G I
Pacifica	G
Parade	ĺ
Pat's Pride	D
Prima Lady	J
Prime Crest	Н
Queencrest	G
Ray CrestRed Cal	G I
Red Dancer (Red Boy)	i
Redglobe	C
Redhaven	G
Red Lady	G
Redtop	G
ReginaRich Lady	G J
Rich May	H
Rio Oso Gem	i
Royal Lady	J
Royal May	G
Ruby May	H
Ryan Sun	Ī
Scarlet Lady	F
September SunSierra Crest	H
Sierra Lady	ï
Sparkle	1
Springcrest	G
Spring Lady	Н
Springold	D
Sugar LadySummer Lady	J L
Summerset	L
Suncrest	G
Topcrest	H
Tra Zee	J
Willie Red	G

Table 1 to paragraph (a)(1)(iv)— Continued

Column A variety	Column B maturity guide
Zee Lady	L

Note: Consult with the Federal or Federal-State Inspection Service Supervisor for the maturity guides applicable to the varieties not listed in this table.

(v) If a grower or handler believes his/ her fruit is meeting the appropriate maturity level but the fruit has not been so graded by the inspector, he/she may appeal the inspection by calling the officer-in-charge of the local Federal-State Inspection Service office to arrange for an on-site examination of the fruit.

(b) During the period April 1 through June 30 of each fiscal period, no handler shall handle any package or container of any variety of peaches not specifically named in paragraphs (a)(2), (a)(3) or (a)(4) of this section unless:

- (c) During the period July 1 through October 31 of each fiscal period, no handler shall handle any package or container of any variety of peaches not specifically named in paragraphs (a)(2), (a)(3) or (a)(4) of this section unless:
- (e) Container tolerances. The contents of individual packages in the lot are subject to the following limitations, provided the averages for the entire lot are within the tolerances specified in this part:
- (1) For packages which contain more than 10 pounds, and a tolerance of 10 percent or more is provided, individual packages shall have not more than one and one-half times the tolerance specified. For packages which contain more than 10 pounds and a tolerance of less than 10 percent is provided, individual packages shall have not more than double the tolerance specified.
- (2) For packages which contain 10 pounds or less, individual packages are not restricted as to the percentage of defects.

Dated: September 18, 1997.

Robert C. Keeney.

Director, Fruit and Vegetable Division. [FR Doc. 97-25410 Filed 9-26-97; 8:45 am] BILLING CODE 3410-02-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-NM-63-AD; Amendment 39-10147; AD 97-20-10]

RIN 2120-AA64

Airworthiness Directives: de Havilland Model DHC-8-100, -200, and -300 Series Airplanes

AGENCY: Federal Aviation Administration, DOT. **ACTION:** Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain de Havilland Model DHC-8-100, -200, and -300 series airplanes, that requires modification of the attitude and heading reference systems (AHRS). This amendment is prompted by a report of loss of power to both AHRS's during flight due to a faulty terminal block to which the signal ground for the AHRS's are connected. The actions specified by this AD are intended to prevent simultaneous power loss to both AHRS's, which could result in reduced controllability of the airplane. **DATES:** Effective November 3, 1997.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of November 3, 1997.

ADDRESSES: The service information referenced in this AD may be obtained from Bombardier, Inc., Bombardier Regional Aircraft Division, Garratt Boulevard, Downsview, Ontario M3K 1Y5, Canada. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, New York Aircraft Certification Office, Engine and Propeller Directorate, 10 Fifth Street, Third Floor, Valley Stream, New York; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Luciano Castracane, Aerospace Engineer, Systems and Equipment Branch, ANE-172, FAA, New York Aircraft Certification Office, Engine and Propeller Directorate, 10 Fifth Street, Third Floor, Valley Stream, New York 11581; telephone (516) 256-7535; fax (516) 568-2716.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to de Havilland Model DHC-8-100, -200, and -300 series airplanes was published in the **Federal Register** on July 22, 1997 (62 FR 39194). That action proposed to require modification of the attitude and heading reference systems (AHRS).

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 173 de Havilland Model DHC-8-100, -200, and -300 series airplanes of U.S. registry will be affected by this AD, that it will take approximately 4 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will cost approximately \$10 per airplane. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$43,250, or \$250 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

Regulatory Impact

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) Is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the rules docket. A copy of

it may be obtained from the rules docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

97–20–10 De Havilland, Inc.: Amendment 39–10147. Docket 97–NM–63–AD.

Applicability: Model DHC-8-100, -200, and -300 series airplanes, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent simultaneous power loss to both attitude and heading reference systems (AHRS), which could result in reduced controllability of the airplane, accomplish the following:

(a) Within 400 hours time-in-service after the effective date of this AD, modify the AHRS's, in accordance with Bombardier Alert Service Bulletin S.B. A8–34–117, Revision 'C', dated February 14, 1997.

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, New York Aircraft Certification Office (ACO), FAA, Engine and Propeller Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, New York ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the New York ACO.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(d) The modification shall be done in accordance with Bombardier Alert Service Bulletin S.B. A8-34-117, Revision 'C', dated February 14, 1997. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Bombardier, Inc., Bombardier Regional Aircraft Division, Garratt Boulevard, Downsview, Ontario M3K 1Y5, Canada. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, New York Aircraft Certification Office, Engine and Propeller Directorate, 10 Fifth Street, Third Floor, Valley Stream, New York; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(e) This amendment becomes effective on November 3, 1997.

Note 3: The subject of this AD is addressed in Canadian airworthiness directive CF-97-01R1, dated February 3, 1997.

Issued in Renton, Washington, on September 19, 1997.

Vi L. Lipski,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 97–25416 Filed 9–26–97; 8:45 am] BILLING CODE 4910–13–U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-ANE-15; Amendment 39-10137; AD 97-19-17]

RIN 2120-AA64

Airworthiness Directives; General Electric Company CT58 Series Turboshaft Engines

AGENCY: Federal Aviation Administration, DOT.
ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to General Electric Company CT58 series turboshaft engines, that requires removal from service of certain compressor rear shafts, initial and repetitive inspections of specific critical rotating parts, and replacement if found cracked, until those parts are removed from service and replaced with improved design parts. This amendment is prompted by a stage 2 turbine wheel