

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

98-06-01 DASSAULT AVIATION:

Amendment 39-10379. Docket 97-NM-190-AD.

Applicability: All Model Mystere-Falcon 50 series airplanes, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent interference between the wiring harnesses and adjacent equipment, support brackets, and structural elements, which could cause an electrical short circuit resulting in fire, and consequent loss of electrical power to essential flight systems; accomplish the following:

(a) Within 6 months or 300 flight hours after the effective date of this AD, whichever occurs first, accomplish the requirements of paragraphs (a)(1), (a)(2), and (a)(3) of this AD in accordance with Dassault Service Bulletin F50-256 (F50-20-5), Revision 1, dated December 22, 1996.

(1) Perform a one-time inspection of the clearances between the wiring harnesses and the adjacent equipment, support brackets, and structural elements. If any clearance is outside the limits specified in the service bulletin, prior to further flight, readjust the clearances in accordance with the service bulletin.

(2) Install Teflon protective strips on the wiring harnesses in the vicinity of the equipment supports.

(3) Install rubber protective strips to the rear edges of the equipment supports.

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR

21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(d) The actions shall be done in accordance with Dassault Service Bulletin F50-256 (F50-20-5), Revision 1, dated December 22, 1996. This incorporation by reference was approved by the Director of the **Federal Register** in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Dassault Falcon Jet Corporation, Teterboro Airport, P.O. Box 2000, South Hackensack, New Jersey 07606. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the **Federal Register**, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in French airworthiness directive 96-094-017(B)R1, dated December 18, 1996.

(e) This amendment becomes effective on April 15, 1998.

Issued in Renton, Washington, on March 3, 1998.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98-6022 Filed 3-10-98; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. 97-NM-277-AD; Amendment 39-10380; AD 98-06-02]

RIN 2120-AA64

**Airworthiness Directives;
Construcciones Aeronauticas, S.A.
(CASA) Model C-212 Series Airplanes**

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain CASA Model C-212 series airplanes, that requires a one-time inspection to detect discrepancies of the spherical bearing of the aileron control rod, and corrective action, if necessary; and installation of an improved retainer washer in the movable joint of the aileron control rod. This amendment is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by this AD are intended to prevent loss of the movable joint of the aileron control rod, caused by deterioration of the hinges, which could result in reduced controllability of the airplane.

DATES: Effective April 15, 1998.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of April 15, 1998.

ADDRESSES: The service information referenced in this AD may be obtained from Construcciones Aeronauticas, S.A., Getafe, Madrid, Spain. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Norman B. Martenson, Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain CASA Model C-212 series airplanes was published in the **Federal Register** on December 31, 1997 (62 FR 68237). That action proposed to require a one-time inspection to detect discrepancies of the spherical bearing of the aileron control rod, and corrective action, if necessary; and installation of an improved retainer washer in the movable joint of the aileron control rod.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 38 airplanes of U.S. registry will be affected by this AD, that it will take approximately 5 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will cost approximately \$56 per airplane. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$13,528, or \$356 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and

that no operator would accomplish those actions in the future if this AD were not adopted.

Regulatory Impact

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

98-06-02 Construcciones Aeronauticas, S.A. (CASA): Amendment 39-10380. Docket 97-NM-277-AD.

Applicability: Model C-212 airplanes, as listed in CASA Service Bulletin SB-212-27-48, dated February 28, 1996; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been

modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent loss of the movable joint of the aileron control rod, caused by deterioration of the hinges, which could result in reduced controllability of the airplane, accomplish the following:

(a) Within 100 flight hours after the effective date of this AD, accomplish the requirements of paragraphs (a)(1) and (a)(2) of this AD in accordance with CASA Service Bulletin SB-212-27-48, dated February 28, 1996.

(1) Perform an inspection of the spherical bearings of the aileron control rod to detect discrepancies. If any discrepancy is found, prior to further flight, replace the whole terminal. And

(2) Install an improved retainer washer in the movable joint of the aileron control rod.

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(d) The actions shall be done in accordance with CASA Service Bulletin SB-212-27-48, dated February 28, 1996. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Construcciones Aeronauticas, S.A., Getafe, Madrid, Spain. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in Spanish airworthiness directive 05/96, dated May 13, 1996.

(e) This amendment becomes effective on April 15, 1998.

Issued in Renton, Washington, on March 3, 1998.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98-6021 Filed 3-10-98; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-CE-81-AD; Amendment 39-10381; AD 98-06-03]

RIN 2120-AA64

Airworthiness Directives; EXTRA Flugzeugbau GmbH Model EA-300 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain EXTRA Flugzeugbau GmbH (EXTRA) Model EA-300 airplanes. This AD requires removing the elevator mass balance assemblies and replacing them with reinforced elevator mass balance assemblies of improved design. This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Germany. The actions specified by this AD are intended to prevent damage and possible jamming of the airplane's control system, which, if not corrected, could cause loss of control of the airplane.

DATES: Effective April 24, 1998.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of April 24, 1998.

ADDRESSES: Service information that applies to this AD may be obtained from Extra Flugzeugbau, GmbH, Schwarze Heide 21, 46569 Hünxe, Germany, telephone: 49-2358-9137-0; facsimile: 49-2858-9137-30. This information may also be examined at the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97-CE-81-AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Karl M. Schletzbaum, Aerospace Engineer, Small Airplane Directorate, Aircraft