safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 37 Fokker Model F28 Mark 1000 through 4000 series airplanes of U.S. registry will be affected by this AD, that it will take approximately 4 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will cost approximately \$3,554 per airplane. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$140,378, or \$3,794 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

Regulatory Impact

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a 'significant rule'' under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

98–06–33 Fokker: Amendment 39–10412. Docket 96–NM–176–AD.

Applicability: Fokker Model F28 Mark 1000 through 4000 series airplanes, equipped with flexible hydraulic hoses, part number (P/N) A71462–401; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD: and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent pressurization of the downlock-actuator during extreme inward sideload conditions (such as touching down at a large crab angle) and consequent lifting of the toggle-links of the main landing gear (MLG), which could result in the collapse of the MLG and reduced controllability of the airplane during landing, accomplish the following:

(a) Within 12 months after the effective date of this AD, replace the flexible hydraulic hoses, P/N A71462–401, that connect to the UP-port of the actuator of the MLG with new flexible hoses, P/N 97867–1, that have built-in restrictor check-valves, in accordance with Fokker Service Bulletin F28/32–123, Revision 1, dated June 30, 1994.

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM–116.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM–116.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to

a location where the requirements of this AD can be accomplished.

(d) The actions shall be done in accordance with Fokker Service Bulletin F28/32–123, Revision 1, dated June 30, 1994. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Fokker Services B.V., Technical Support Department, P.O. Box 75047, 1117 ZN Schiphol Airport, the Netherlands. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in Dutch airworthiness directive BLA 94–095 (A), dated July 15, 1994.

(e) This amendment becomes effective on April 27, 1998.

Issued in Renton, Washington, on March 12, 1998.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 98–7093 Filed 3–20–98; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 94-NM-212-AD; Amendment 39-10419; AD 98-07-01]

RIN 2120-AA64

Airworthiness Directives; British Aerospace BAe Model ATP Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain British Aerospace BAe Model ATP airplanes, that requires inspections and tests for damage of the engine power cables, and replacement of any damaged cable with a new cable. This amendment also provides for optional modification of the engine power control cable pulley assembly. This amendment is prompted by a report of failure of an engine power cable, which could cause loss of function of the power control levers on the console. The actions specified by this AD are intended to prevent loss of function of the power control levers on the console, and subsequent loss of normal control of engine power.

DATES: Effective April 27, 1998.

The incorporation by reference of certain publications listed in the regulations is approved by the Director

of the Federal Register as of April 27, 1998.

ADDRESSES: The service information referenced in this AD may be obtained from AI(R) American Support, Inc., 13850 Mclearen Road, Herndon, Virginia 20171. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC. FOR FURTHER INFORMATION CONTACT:

FOR FURTHER INFORMATION CONTACT: Norman B. Martenson, Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055–4056; telephone (425) 227–2110; fax (425) 227–1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain British Aerospace BAe Model ATP airplanes was published in the **Federal Register** on September 13, 1995 (60 FR 47501). That action proposed to require inspections and tests for damage of the engine power cables, and replacement of any damaged cable with a new cable.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the single comment received.

The commenter supports the proposed rule.

Actions Since Issuance of Proposal

Since the issuance of the proposal, the manufacturer has issued British Aerospace Service Bulletin ATP-76-18, dated June 21, 1995, which describes procedures for the modification of the engine power control cable pulley assembly. The modification involves increasing the diameter of the pulley of the engine power control quadrant's lower pulley group between stations 398FS to 408FS from 1.5 inches to 2.36 inches, and repositioning of the lower pulley group slightly forward and upward. The service bulletin specifies that, if accomplished, this modification would extend the fatigue life of the engine power control cables, and would allow the repetitive inspection interval to be increased from 1,000 landings to 5,000 landings. The Civil Aviation Authority (CAA), which is the airworthiness authority for the United Kingdom, classified this service bulletin as optional.

The FAA has revised this final rule to add accomplishment of this modification as an option to permit extension of the repetitive inspection interval specified in this AD. Additionally, the cost impact information, below, has been revised to specify the number of work hours that would be required to accomplish the optional modification.

Conclusion

After careful review of the available data, including the comment noted above, the FAA has determined that air safety and the public interest require the adoption of the rule with the change previously described. The FAA has determined that this change will neither increase the economic burden on any operator nor increase the scope of the AD.

Cost Impact

The FAA estimates that 10 airplanes of U.S. registry will be affected by this AD, that it will take approximately 2 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$1,200, or \$120 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

Should an operator elect to accomplish the optional modification provided by this AD, it would take approximately 80 work hours per airplane to accomplish it, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of this optional modification is estimated to be \$4,800 per airplane.

Regulatory Impact

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44

FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

98-07-01 British Aerospace Regional Aircraft [Formerly Jetstream Aircraft Limited, British Aerospace (Commercial Aircraft) Limited]: Amendment 39– 10419. Docket 94-NM-212-AD.

Applicability: BAe Model ATP airplanes, constructor's numbers 2002 through 2063 inclusive; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent loss of function of the power control levers on the console, and subsequent loss of normal control of engine power due to failure of the engine power cables, accomplish the following:

(a) Perform a detailed visual inspection and tests for damage of the engine power cables, in accordance with Jetstream Service Bulletin ATP-76-16, dated October 14, 1994, at the earlier of the times specified in paragraphs (a)(1) and (a)(2) of this AD. Thereafter repeat this inspection and tests at intervals not to exceed 1,000 landings.

- (1) Prior to the accumulation of 1,000 total landings on the engine power cable, or within 200 landings after the effective date of this AD, whichever occurs later.
- (2) Within 75 days after the effective date of this AD.
- (b) If any damaged engine power cable is found, prior to further flight, replace the damaged engine power cable with a new cable in accordance with Jetstream Service Bulletin ATP-76-16, dated October 14, 1994. Except as provided by paragraph (c) of this AD, repeat the inspection and tests required by paragraph (a) of this AD thereafter at intervals not to exceed 1,000 landings.
- (c) Modification of the engine power control cable pulley assembly in accordance with British Aerospace Service Bulletin ATP-76-18, dated June 21, 1995, allows the interval for accomplishment of the repetitive inspection and tests required by paragraph (a) of this AD to be increased to 5,000 landings.
- (d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM–116.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM_116

- (e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.
- (f) The inspection, tests, and replacement shall be done in accordance with Jetstream Service Bulletin ATP-76-16, dated October 14, 1994. The modification, if accomplished, shall be done in accordance with Jetstream Service Bulletin ATP-76-18, dated June 21, 1995. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from AI(R) American Support, Inc., 13850 Mclearen Road, Herndon, Virginia 20171. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington,
- (g) This amendment becomes effective on April 27, 1998.

Issued in Renton, Washington, on March 16, 1998.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 98–7365 Filed 3–20–98; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Airspace Docket No. 97-AGL-61]

Modification of Class D Airspace; Minot AFB, ND; and Class E Airspace; Minot, ND

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: This action modifies Class D airspace at Minot Air Force Base (AFB), ND, and Class E airspace at Minot, ND. A review of the Instrument Landing System (ILS) 1 or Tactical Air Navigation (TACAN) Runway 29 Standard Instrument Approach Procedure (SIAP), the Instrument Landing System/Distance Measuring Equipment (ILS/DME) 2 Runway 29 SIAP, the ILS/DME Runway 11 SIAP, and the TACAN Runway 11 SIAP for Minot AFB necessitates these modifications. Controlled airspace extending upward from the surface, controlled airspace extending upward from 700 feet above ground level (AGL), and controlled airspace extending upward from 1,200 feet AGL is needed to contain aircraft executing these approaches. This proposal would increase the radius and remove the extensions to the Class D airspace for Minot AFB, ND, and would increase the radius and add a northwest extension to that portion of the Minot, ND, Class E airspace associated with Minot AFB, ND.

EFFECTIVE DATE: 0901 UTC, June 18, 1998.

FOR FURTHER INFORMATION CONTACT: Michelle M. Behm, Air Traffic Division, Airspace Branch, AGL–520, Federal Aviation Administration, 2300 East Devon Avenue, Des Plaines, Illinois 60018, telephone (847) 294–7568.

SUPPLEMENTARY INFORMATION:

History

On Monday, December 22, 1997, the FAA proposed to amend 14 CFR part 71 to modify Class D airspace at Minot AFB, ND, and Class E airspace at Minot, ND (62 FR 66838). A recent joint FAA/Air Force review of the controlled

airspace for Minot AFB revealed a need to reinstate controlled airspace inadvertently dropped during the 1993 United States airspace reclassification. This action was completed by Final Rule on November 5, 1997 (97-AGL-59, 62 FR 59783). Further review of the current instrument approach procedures for Minot AFB, including the ILS 1 or TACAN Runway 29 SIAP, the ILS/DME 2 Runway 29 SIAP, the ILS/DME Runway 11 SIAP, and the TACAN Runway 11 SIAP, indicated the need to modify the existing controlled airspace. The proposal was to increase the radius and remove the extensions to the Class D airspace for Minot AFB, ND, and to increase the radius and add a northwest extension to that portion of the Minot, ND, Class E airspace associated with Minot AFB, ND to contain Instrument Flight Rules (IFR) operations in controlled airspace during portions of the terminal operation and while transiting between the enroute and terminal environments.

Interested parties were invited to participate in this rulemaking proceeding by submitting written comments on the proposal to the FAA. No comments objecting to the proposal were received. Class D airspace designations are published in paragraph 5000, and Class E airspace designations for airspace areas extending upward from 700 feet or more above the surface of the earth are published in paragraph 6005 FAA Order 7400.9E dated September 10, 1997, and effective September 16, 1997, which is incorporated by reference in 14 CFR 71.1. The Class D and Class E airspace designations listed in this document will be published subsequently in the Order.

The Rule

This amendment to 14 CFR part 71 modifies Class D airspace at Minot AFB, ND and Class E airspace at Minot, ND, to accommodate aircraft executing the ILS 1 or TACAN Runway 29 SIAP, the ILS/DME 2 Runway 29 SIAP, the ILS/ DME Runway 11 SIAP, and the TACAN Runway 11 SIAP, and IFR operations at Cooperstown Municipal Airport by increasing the radius and removing the extensions to the Class D airspace for Minot AFB, ND, and by increasing the radius and adding a northwest extension to that portion of the Minot, ND, Class E airspace associated with Minot AFB, ND. The areas will be depicted on appropriate aeronautical charts.

The FAA has determined that this regulation only involves an established body of technical regulations for which frequent and routine amendments are