Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

98-13-10 Cessna Aircraft Company: Amendment 39-10598; Docket No. 98-CE-59-AD.

Applicability: Model 182S airplanes, all serial numbers, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated in the body of this AD, unless already accomplished.

To detect and correct damage to the engine exhaust mufflers caused by high stresses imposed on the attachment of the exhaust at the area of the firewall and cracking, which could result in exhaust gases entering the airplane cabin with consequent crew and passenger injury, accomplish the following:

(a) Within the next 5 days after the effective date of this AD, accomplish the following:

(1) Fabricate a placard that specifies immediately inspecting all engine exhaust muffler end plates when the engine backfires upon start-up, and install this placard on the instrument panel within the pilot's clear view. The placard should utilize letters of at least 0.10-inch in height and contain the following words:

"If the engine backfires upon start-up, prior to further flight, inspect and replace (as necessary) all engine exhaust muffler end plates in accordance with AD 98–13–10"

(2) Insert a copy of this AD into the Limitations Section of the airplane flight manual (AFM).

(b) Within the next 25 hours time-inservice (TIS) after the effective date of this AD and thereafter at intervals not to exceed 25 hours TIS after the previous inspection (including any inspection accomplished after an engine backfire), inspect all engine exhaust muffler end plates (four total) for cracks on the forward (upstream) or aft (downstream) end of each muffler can. Prior to further flight, replace any engine exhaust muffler where an end plate is found cracked. The replacement does not eliminate the repetitive inspection requirement of this AD.

Note 2: Cessna Service Bulletin SB98–78–02, Issued: June 6, 1998, depicts the area to be inspected. The actions of this service bulletin are different from those required by this AD. This AD takes precedence over the actions specified in the service bulletin, and accomplishment of the service bulletin is not considered an alternative method of compliance to the actions of this AD. Copies of this service bulletin may be obtained from the Cessna Aircraft Company, Product Support, P.O. Box 7706, Wichita, Kansas 67277.

(c) Fabricating and installing the placard and inserting this AD into the Limitations Section of the AFM, as required by paragraph (a) of this AD, may be performed by the owner/operator holding at least a private pilot certificate as authorized by section 43.7 of the Federal Aviation Regulations (14 CFR 43.7), and must be entered into the aircraft records showing compliance with this AD in accordance with section 43.9 of the Federal Aviation Regulations (14 CFR 43.9).

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(e) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Wichita Aircraft Certification Office (ACO), 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Wichita ACO.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Wichita ACO.

- (f) Information related to this AD may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri.
- (g) This amendment becomes effective on July 8, 1998.

Issued in Kansas City, Missouri, on June 10, 1998.

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98–16015 Filed 6–16–98; 8:45 am] BILLING CODE 4910–13–U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-CE-08-AD; Amendment 39-10596; AD 98-13-08]

RIN 2120-AA64

Airworthiness Directives; Pilatus Aircraft Ltd. Model PC-12 Airplanes

AGENCY: Federal Aviation Administration, DOT.
ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain Pilatus Aircraft Ltd. (Pilatus) Model PC-12 airplanes. This AD requires replacing and re-routing the power return cables on the starter generator and generator 2, inserting a temporary revision to the pilot operating handbook (POH), and installing a placard near the standby magnetic compass. This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Switzerland. The actions specified by this AD are intended to prevent directional deviation on the standby magnetic compass caused by an overload of electrical current in the airplane structure, which could result in flightpath deviation during critical phases of flight in icing conditions and instrument meteorologic conditions (IMC).

DATES: Effective July 31, 1998.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of July 31, 1998.

ADDRESSES: Service information that applies to this AD may be obtained from Pilatus Aircraft Ltd., Marketing Support Department, CH–6370 Stans, Switzerland; telephone: +41 41–6196 233; facsimile: +41 41–6103 351. This information may also be examined at the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97–CE–08–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Mr. Roman T. Gabrys, Aerospace Engineer, Small Airplane Directorate, Airplane Certification Service, FAA, 1201 Walnut, suite 900, Kansas City, Missouri 64106; telephone: (816) 426–6934; facsimile: (816) 426–2169.

SUPPLEMENTARY INFORMATION:

Events Leading to the Issuance of This AD

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Pilatus Model PC–12 airplanes was published in the **Federal Register** as a notice of proposed rulemaking (NPRM) on April 1, 1998 (63 FR 15795). The NPRM proposed to require replacing and re-routing the power return cables on the starter generator and generator 2; inserting a temporary revision to the POH; and installing a placard near the standby magnetic compass, using at least 1/8-inch letters, with the following words:

"STANDBY COMPASS FOR CORRECT READING CHECK: WINDSHIELD DE-ICE LH & RH HEAVY & COOLING SYSTEM OFF."

Accomplishment of the proposed action as specified in the NPRM would be in accordance with Pilatus PC XII Service Bulletin No. 24–002, Rev. No. 1, dated September 20, 1996.

The NPRM was the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Switzerland.

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were received on the proposed rule or the FAA's determination of the cost to the public.

The FAA's Determination

After careful review of all available information related to the subject presented above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed except for minor editorial corrections. The FAA has determined that these minor corrections will not change the meaning of the AD and will not add any additional burden upon the public than was already proposed.

Cost Impact

The FAA estimates that 40 airplanes in the U.S. registry will be affected by this AD, that it will take approximately 12 workhours per airplane to accomplish the cable re-routing and replacement, and that the average labor rate is approximately \$60 an hour. Parts will be provided free from the manufacturer upon request. Incorporating the POH revisions and installing a placard may be performed by the owner/operator holding at least a private pilot certificate as authorized by section 43.7 of the Federal Aviation

Regulations (14 CFR 43.7), and must be entered into the aircraft records showing compliance with this AD in accordance with section 43.9 of the Federal Aviation Regulations (14 CFR 43.9).

Based on these figures, the total cost impact of this AD on U.S. operators is estimated to be \$28,800, or \$720 per airplane.

Regulatory Impact

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a ''significant rule'' under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the final evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

98–13–08 Pilatus Aircraft Ltd.: Amendment 39–10596; Docket No. 97-CE–08-AD.

Applicability: Model PC-12 airplanes, serial numbers 101 through 147, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (f) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within the next 100 hours time-in-service (TIS) after the effective date of this AD, unless already accomplished.

To prevent directional deviation on the standby magnetic compass caused by an overload of electrical current in the airplane structure, which could result in flight-path deviation during critical phases of flight in icing conditions and Instrument Meteorologic Conditions (IMC), accomplish the following:

- (a) Re-route and replace the starter generator cable and the generator 2 power return cables with new cables of improved design in accordance with the Accomplishment Instructions section in Pilatus PC XII Service Bulletin (SB) No. 24–002, Rev. No. 1, dated September 20, 1996.
- (b) Remove the temporary revision titled "Electrical Cables," dated March 7, 1996, from the Pilot Operating Handbook (POH) and insert a temporary revision titled "Electrical Cables" Rev. 1, dated July 12, 1996, in accordance with the Accomplishment Instructions section in Pilatus PC XII SB No. 24–002, Rev. No. 1, dated September 20, 1996.
- (c) Install a placard with the following words (using at least 1/8-inch letters) near the standby magnetic compass in accordance with the Accomplishment Instructions section in Pilatus PC XII SB No. 24–002, Rev. No. 1, dated September 20, 1996:
- "STANDBY COMPASS FOR CORRECT READING CHECK: WINDSHIELD DE-ICE LH & RH HEAVY & COOLING SYSTEM OFF."
- (d) Incorporating the POH revisions and installing a placard, as required by paragraphs (b) and (c) of this AD, may be performed by the owner/operator holding at least a private pilot certificate as authorized by section 43.7 of the Federal Aviation Regulations (14 CFR 43.7), and must be entered into the aircraft records showing compliance with this AD in accordance with section 43.9 of the Federal Aviation Regulations (14 CFR 43.9).
- (e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.
- (f) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Small Airplane Directorate, 1201 Walnut, suite 900, Kansas

City, Missouri 64106. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Small Airplane Directorate.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Small Airplane Directorate.

(g) Questions or technical information related to Pilatus PC XII SB No. 24-002, Rev. No. 1, dated September 20, 1996, should be directed to Pilatus Aircraft Ltd., Customer Liaison Manager, CH-6370 Stans, Switzerland; telephone: +41 41 6196 233; facsimile: +41 41 6103 351. This service information may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

(h) The modification, replacement, insertion, and installation required by this AD shall be done in accordance with Pilatus PC XII Service Bulletin No. 24-002, Rev. No. 1, dated September 20, 1996. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Pilatus Aircraft Ltd., Customer Liaison Manager, CH-6370 Stans, Switzerland. Copies may be inspected at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington,

Note 3: The subject of this AD is addressed in Swiss AD No. HB-96-140, dated March 18, 1996, and Swiss AD No. HB 97-001, dated January 1, 1997.

(i) This amendment becomes effective on July 31, 1998.

Issued in Kansas City, Missouri, on June 9,

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98-16023 Filed 6-16-98; 8:45 am] BILLING CODE 4910-13-U

FEDERAL TRADE COMMISSION

16 CFR Parts 2 and 4

Delegation of Authority to Respond To Requests for Information

AGENCY: Federal Trade Commission (FTC).

ACTION: Final rules.

SUMMARY: The Commission is revising its rules to authorize the General Counsel's designee to: determine whether information is confidential or should be placed on the public record; respond to requests for nonpublic information by Federal and State agencies; determine which portions of closed meeting transcripts or minutes to

make public; determine which portions of compliance reports, prior approval requests and related supplemental materials, will be treated as confidential when confidential treatment is requested at the time of submission; and respond to requests to use nonpublic memoranda as writing samples or for purposes of teaching, lecturing or writing. The General Counsel will designate the Deputy General Counsel or an Assistant General Counsel (or a senior manager in an equivalent level) to make these determinations. The Commission is adopting these changes in order to improve and expedite the process for responding to such requests. The changes will affect internal procedures only and are not intended to influence the outcomes of requests made under the Rules.

The Commission is inserting crossreferences to certain confidentiality rules to clarify and make consistent its procedures and is removing language that is thereby made repetitive or is otherwise unnecessary.

EFFECTIVE DATE: These amendments are effective June 17, 1998.

FOR FURTHER INFORMATION CONTACT:

Laura Berger, Attorney, 202–326–2471, Office of the General Counsel, FTC, Sixth Street & Pennsylvania Avenue, NW, Washington, D.C. 20580.

SUPPLEMENTARY INFORMATION: The rule amendments relate solely to agency practice and thus are not subject to the notice and comment requirements of the Administrative Procedure Act, 5 U.S.C. 553(a)(2), or to the requirements of the Regulatory Flexibility Act, 5 U.S.C. 601(2). The Paperwork Reduction Act, 44 U.S.C. 3501–3520, does not apply to these amendments because they do not involve a request for any person to report, keep records, or disclose information, and because the amendment is purely administrative and does not affect persons as defined by the Act. See 5 CFR 1320.3(c), 5 CFR 1320.3(c)(4).

List of Subjects

16 CFR Part 2

Administrative practice and procedure.

16 CFR Part 4

Administrative practice and procedure, Sunshine Act.

For the reasons set forth in the preamble, the Federal Trade Commission amends Title 16, chapter 1, subchapter A, of the Code of Federal Regulations as follows:

PART 2—NONADJUDICATIVE PROCEDURES

1. The authority citation for part 2, continues to read as follows:

Authority: Sec. 6, 38 Stat. 721; 15 U.S.C.

2. Section 2.33 is revised to read as follows:

§ 2.33 Compliance procedure.

The Commission may in its discretion require that a proposed agreement containing an order to cease and desist be accompanied by an initial report signed by the respondent setting forth in precise detail the manner in which the respondent will comply with the order when and if entered. Such report will not become part of the public record unless and until the accompanying agreement and order are accepted by the Commission. At the time any such report is submitted a respondent may request confidentiality for any portion thereof with a precise showing of justification therefor as set out in § 4.9(c) and the General Counsel or the General Counsel's designee will dispose of such requests in accordance with that section.

3. Section 2.41(f)(5) is revised to read as follow:

§ 2.41 Reports of compliance.

(f) * * *

(5) Persons submitting information that is subject to public record disclosure under this section may request confidential treatment for that information or portions thereof in accordance with § 4.9(c) and the General Counsel or the General Counsel's designee will dispose of such requests in accordance with that section. Nothing in this section requires that confidentiality requests be resolved prior to, or contemporaneously with, the disposition of the application.

PART 4—MISCELLANEOUS RULES

4. The authority citation for part 4 continues to read as follows:

Authority: Sec. 6, 38 Stat. 721; 15 U.S.C. 46.

5. Section 4.9 is amended by revising paragraphs (b)(7)(i), (c)(1) and (c)(3) to read as follows:

§ 4.9 The public record. *

(b) * * *

*

(7) Compliance/Enforcement (16 CFR 2.33, 2.41). (i) Reports of compliance filed pursuant to the rules in this chapter or pursuant to a provision in a Commission order and supplemental