# **Proposed Rules**

### **Federal Register**

Vol. 63, No. 199

Thursday, October 15, 1998

This section of the FEDERAL REGISTER contains notices to the public of the proposed issuance of rules and regulations. The purpose of these notices is to give interested persons an opportunity to participate in the rule making prior to the adoption of the final rules.

#### **DEPARTMENT OF TRANSPORTATION**

# **Federal Aviation Administration**

14 CFR Part 39

[Docket No. 98-NM-258-AD]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 737–600, –700, and –800 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Boeing Model 737-600, -700, and -800 series airplanes. This proposal would require repetitive inspections to detect damage of the aft strut insulation blanket. This proposal also would require eventual replacement of the insulation blankets with new, improved blankets, which would constitute terminating action for the requirements of this AD. This proposal is prompted by reports of damaged aft strut insulation blankets. The actions specified by the proposed AD are intended to prevent such damage, which could result in exposure of the lower surface of the strut to extreme high temperatures, consequent creation

**DATES:** Comments must be received by November 30, 1998.

increased risk of a fuel tank explosion

of a source of fuel ignition, and

and fire.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 98-NM-258-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. FOR FURTHER INFORMATION CONTACT: Bernie Gonzalez, Aerospace Engineer, Propulsion Branch, ANM-140S, FAA, Transport Airplane Directorate, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2682; fax (425) 227-1181.

# SUPPLEMENTARY INFORMATION:

#### **Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 98–NM–258–AD." The postcard will be date stamped and returned to the commenter.

### Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 98-NM-258-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

#### Discussion

The FAA has received reports indicating that during an evaluation of the aft strut insulation blankets that were installed on three Boeing Model 737 series flight test airplanes, cracks were found in the face sheets of the insulation blankets. Inspections on five in-service airplanes revealed two additional airplanes with such cracked insulation blankets. At the time these damaged insulation blankets were detected, the two in-service airplanes had accumulated 730 flight hours and 946 flight hours, respectively. Damage of these insulation blankets, which are located between the engine exhaust nozzle and the underside of the aft compartment of the engine strut, could cause the temperature on the bottom of that compartment to exceed normal limits during engine operation. That compartment is located immediately below the wing fuel tank and contains hydraulic lines and components where fuel leaks may occur. Such damage, if not corrected, could result in exposure of the lower surface of the strut to extreme high temperatures, consequent creation of a source of fuel ignition, and increased risk of a fuel tank explosion and fire.

# **Explanation of Relevant Service Information**

The FAA has reviewed and approved Boeing Alert Service Bulletin 737-54A1038, dated May 7, 1998, as revised by Notice of Status Change 737-54A1038 NSC 01, dated June 18, 1998, which describes procedures for repetitive visual and borescope inspections to detect cracks in and/or separation of the face sheet of the aft strut insulation blanket. The alert service bulletin also describes procedures for replacement of the aft strut insulation blanket with a new, improved blanket, which would eliminate the need for the repetitive inspections. Accomplishment of the actions specified in the alert service bulletin is intended to adequately address the identified unsafe condition.

# **Explanation of Requirements of Proposed Rule**

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would require accomplishment of the actions

specified in the alert service bulletin described previously, except as discussed below.

### Differences Between Proposed Rule and Alert Service Bulletin

Operators should note that this AD proposes to mandate, within 18 months, the replacement of the aft strut insulation blankets that is described in Boeing Alert Service Bulletin 737–54A1038, as revised by Notice of Status Change 737–54A1038 NSC 01, as terminating action for the repetitive inspections.

The FAA has determined that longterm continued operational safety will be better assured by design changes to remove the source of the problem, rather than by repetitive inspections. Longterm inspections may not be providing the degree of safety assurance necessary for the transport airplane fleet. This, coupled with a better understanding of the human factors associated with numerous continual inspections, has led the FAA to consider placing less emphasis on inspections and more emphasis on design improvements. The proposed replacement requirement is in consonance with these conditions.

### **Cost Impact**

There are approximately 33 airplanes of the affected design in the worldwide fleet. The FAA estimates that 26 airplanes of U.S. registry would be affected by this proposed AD.

It would take approximately 1 work hour per airplane to accomplish the proposed inspection, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the inspection proposed by this AD on U.S. operators is estimated to be \$1,560, or \$60 per airplane, per inspection cycle.

It would take approximately 1 work hour per airplane to accomplish the proposed replacement, at an average labor rate of \$60 per work hour. Required parts would be provided by the manufacturer at no cost to the operators. Based on these figures, the cost impact of the replacement proposed by this AD on U.S. operators is estimated to be \$1,560, or \$60 per airplane.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

# **Regulatory Impact**

The regulations proposed herein would not have substantial direct effects on the States, on the relationship

between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

# List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

# The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows: **Authority:** 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Boeing: Docket 98-NM-258-AD.

Applicability: Model 737–600, –700, and –800 series airplanes, line numbers 1 through 64 inclusive, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not

been eliminated, the request should include specific proposed actions to address it.

*Compliance:* Required as indicated, unless accomplished previously.

To prevent damage of the aft strut insulation blankets, which could result in exposure of the lower surface of the strut to extreme high temperatures, consequent creation of a source of fuel ignition, and increased risk of a fuel tank explosion and fire, accomplish the following:

(a) Within 500 flight hours since date of manufacture of the airplane, or within 30 days after the effective date of this AD, whichever occurs later, perform a visual or borescope inspection to detect damage (cracks greater than 2.00 inches and/or separation of the face sheet) of the aft strut insulation blanket, part number (P/N) S315A213-42, in accordance with Boeing Alert Service Bulletin 737-54A1038, dated May 7, 1998, as revised by Notice of Status Change 737-54A1038 NSC 01, dated June 18, 1998. Thereafter, repeat the visual or borescope inspection at intervals not to exceed 250 flight hours.

(b) If damage (cracks greater than 2.00 inches and/or separation of the face sheet) of any aft strut insulation blanket is detected during any inspection required by paragraph (a) of this AD, prior to further flight, accomplish either paragraph (b)(1) or (b)(2) of this AD.

(1) Replace any damaged insulation blanket having P/N S315A213–42 with a new insulation blanket having P/N S315A213–42, in accordance with Boeing Alert Service Bulletin 737–54A1038, dated May 7, 1998, as revised by Notice of Status Change 737–54A1038 NSC 01, dated June 18, 1998. Thereafter, repeat the visual or borescope inspection required by paragraph (a) of this AD at intervals not to exceed 250 flight hours. Or

(2) Replace any damaged insulation blanket having P/N S315A213–42 with a new, improved insulation blanket having P/N S315A213–47, in accordance with Boeing Alert Service Bulletin 737–54A1038, dated May 7, 1998, as revised by Notice of Status Change 737–54A1038 NSC 01, dated June 18, 1998. Accomplishment of this replacement constitutes terminating action for the repetitive inspection requirements of this AD.

(c) Within 18 months after the effective date of this AD, replace any aft strut insulation blanket having P/N S315A213–42 with a new, improved insulation blanket having P/N S315A213–47, in accordance with Boeing Alert Service Bulletin 737–54A1038, dated May 7, 1998, as revised by Notice of Status Change 737–54A1038 NSC 01, dated June 18, 1998. Accomplishment of this replacement constitutes terminating action for the requirements of this AD.

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on October 5, 1998.

#### Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 98–27603 Filed 10–14–98; 8:45 am] BILLING CODE 4910–13–U

### **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

14 CFR Part 39

[Docket No. 98-NM-250-AD]

RIN 2120-AA64

# Airworthiness Directives; Fokker Model F.28 Mark 0100 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Notice of proposed rulemaking

(NPRM).

**SUMMARY:** This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Fokker Model F.28 Mark 0100 series airplanes. This proposal would require modification of the aft cabin sidewall area to improve decompression venting. For certain airplanes, this proposal also would require modification of the aft wardrobe/ stowage area door and installation of decompression panels to improve decompression venting. This proposal is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by the proposed AD are intended to prevent damage to the cabin floor in the event of sudden decompression in the cargo compartment, which could result in injury to passengers, reduced structural integrity of the airplane, and the loss of airplane systems.

**DATES:** Comments must be received by November 16, 1998.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–114, Attention: Rules Docket No. 98–NM–250–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056. Comments may be inspected at this

location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Fokker Services B.V., Technical Support Department, P.O. Box 75047, 1117 ZN Schiphol Airport, the Netherlands. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Norman B. Martenson, Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

#### SUPPLEMENTARY INFORMATION:

### **Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 98–NM–250–AD." The postcard will be date stamped and returned to the commenter.

### **Availability of NPRMs**

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 98-NM-250-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

### Discussion

The Rijksluchtvaartdienst (RLD), which is the airworthiness authority for

the Netherlands, notified the FAA that an unsafe condition may exist on certain Fokker Model F.28 Mark 0100 series airplanes. The RLD advises that the decompression venting provisions in the aft cabin sidewall area and in the aft wardrobe/stowage area are inadequate in reducing the pressure differential between the passenger and cargo compartments in the event of a sudden decompression of the cargo compartment. Such inadequate reduction in the pressure differential could result in damage to the cabin floor. This condition, if not corrected, could result in injury to passengers, reduced structural integrity of the airplane, and loss of airplane systems.

# **Explanation of Relevant Service Information**

Fokker has issued Service Bulletin SBF100–25–082, Revision 1, dated May 7, 1998, which describes procedures for modification of the aft cabin sidewall area to improve decompression venting. For airplanes equipped with an aft service/emergency door, Fokker also has issued Service Bulletin SBF100–25–083, dated April 30, 1998, which describes procedures for modification of the aft wardrobe/stowage area door and installation of decompression panels to improve decompression venting.

Accomplishment of the actions specified in these service bulletins is intended to adequately address the identified unsafe condition. The RLD classified these service bulletins as mandatory and issued Dutch airworthiness directive BLA 1998–065 (A), dated May 29, 1998, in order to assure the continued airworthiness of these airplanes in the Netherlands.

### **FAA's Conclusions**

This airplane model is manufactured in the Netherlands and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the RLD has kept the FAA informed of the situation described above. The FAA has examined the findings of the RLD, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

# **Explanation of Requirements of Proposed Rule**

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same