in the service bulletin described previously.

### **Cost Impact**

The FAA estimates that 61 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 3 work hours per airplane to accomplish the proposed actions, and that the average labor rate is \$60 per work hour. Required parts would be supplied by the manufacturer at no cost to the operators. Based on these figures, the cost impact of the actions proposed by this AD on U.S. operators is estimated to be \$10,980, or \$180 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

### **Regulatory Impact**

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1) Is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

### **List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Safety.

# **The Proposed Amendment**

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Airbus Industrie: Docket 98-NM-301-AD.

Applicability: Model A300–600 series airplanes on which Airbus Modification 4801 was accomplished during production and on which Airbus Modification 10778 (reference Airbus Service Bulletin A300–31–6051, dated June 28, 1996) has been accomplished; except those airplanes on which Airbus Modification 11683 (reference Airbus Service Bulletin A300–28–6055, dated January 28, 1997, and Revision 01, dated July 24, 1998) has been accomplished; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent continuous aft transfer of fuel due to the fuel level sensing amplifier (FLSA) not supplying electrical power to the trim tank overflow sensor, which could result in potential loss of fuel during flight, accomplish the following:

(a) Except as provided by paragraph (b) of this AD, within 2 months after the effective date of this AD, remove the FLSA of the trim tank system, modify the polarization pin code in the electronics bay, and install a new, improved FLSA, in accordance with Airbus Service Bulletin A300–28–6055, Revision 01, dated July 24, 1998.

**Note 2:** Accomplishment of the actions specified in paragraph (a) of this AD, prior to the effective date of this AD, in accordance with Airbus Service Bulletin A300–28–6055 dated January 28, 1997, is considered acceptable for compliance with the applicable actions specified in this AD.

(b) For airplanes on which Airbus Service Bulletin A300–31–6051, dated June 28, 1996, is accomplished after the effective date of this AD: Concurrent with the accomplishment of Airbus Service Bulletin A300–31–6051, accomplish the actions required by paragraph (a) of this AD, in accordance with Airbus Service Bulletin A300–28–6055, Revision 01, dated July 24, 1998.

- (c) As of the effective date of this AD, no person shall install a FLSA having part number 722–295–2, on any airplane.
- (d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM–116.

**Note 3:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM–116.

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

**Note 4:** The subject of this AD is addressed in French airworthiness directive 98–249–252(B), dated July 1, 1998.

Issued in Renton, Washington, on December 14, 1998.

### Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 98–33539 Filed 12–17–98; 8:45 am] BILLING CODE 4910–13–P

### **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

### 14 CFR Part 39

[Docket No. 97-NM-244-AD]

RIN 2120-AA64

Airworthiness Directives; McDonnell Douglas Model DC-9 Series Airplanes, and Model MD-88 and MD-90-30 Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Supplemental notice of

proposed rulemaking.

**SUMMARY:** This document revises an earlier proposed airworthiness directive (AD), applicable to certain McDonnell Douglas Model DC-9-80 series airplanes, and Model MD-88 and MD-90–30 airplanes, that would have required replacement of the lanyard assembly pins of the evacuation slides with solid stainless steel pins. That proposal was prompted by a report that, due to stress corrosion on the lanyard pins, the arms of the lanyard assembly of the evacuation slide were found to be frozen. This new action revises the proposed rule by expanding the applicability of the proposed rule to

include additional airplanes, and revising the type of replacement pins. The actions specified by this new proposed AD are intended to prevent the improper deployment of the evacuation slide due to such stress corrosion, which could delay or impede evacuation of passengers during an emergency.

**DATES:** Comments must be received by January 12, 1999.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 97-NM-244-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from The Boeing Company, Douglas Products Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Technical Publications Business Administration, Dept. C1–L51 (2–60). This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Transport Airplane Directorate, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California.

FOR FURTHER INFORMATION CONTACT: Alan Sinclair, Aerospace Engineer, Systems and Equipment Branch, ANM–130L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712; telephone (562) 627–5338; fax (562) 627–5210.

# SUPPLEMENTARY INFORMATION:

### **Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 97–NM–244–AD." The postcard will be date stamped and returned to the commenter.

# Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 97-NM-244-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

### **Discussion**

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to add an airworthiness directive (AD), applicable to certain McDonnell Douglas Model DC-9-80 series airplanes, and Model MD-88 and MD-90-30 airplanes, was published as a notice of proposed rulemaking (NPRM) in the Federal Register on April 2, 1998 (63 FR 16172). That NPRM would have required replacement of the lanyard assembly pins of the evacuation slides with solid stainless steel pins. That NPRM was prompted by a report that, due to stress corrosion on the lanyard pins, the arms of the lanyard assembly of the evacuation slide were found to be frozen. That condition, if not corrected, could result in the improper deployment of the evacuation slide, which could delay or impede evacuation of passengers during an emergency.

# Comments Received to Previous Proposal

Due consideration has been given to the comments received in response to the NPRM.

# **Requests To Reference Latest Service Information**

Several commenters request that the applicability and paragraph (a) of the proposed AD be revised to reference Revision 01 of McDonnell Douglas Alert Service Bulletin DC9–25A357. Two of these commenters state that the effectivity listing of this alert service bulletin has been revised to include additional airplanes.

The FAA concurs with the commenters' requests to reference Revision 01 of the alert service bulletin.

Since issuance of the NPRM, the FAA has reviewed and approved Revision 01 of McDonnell Douglas Alert Service Bulletin DC9-25A357, dated March 16, 1998. The replacement procedures described in this revised alert service bulletin are essentially identical to those described in the original version (which was referenced in the proposed AD as the appropriate source of service information for accomplishment of the replacement). However, the effectivity listing of the alert service bulletin, among other items (including affected spares), has been revised to include additional Model DC-9 series airplanes and MD-88 airplanes that are subject to the identified unsafe condition. Therefore, the FAA has revised the supplemental NPRM to reference Revision 01 of the alert service bulletin as the appropriate source of service information (for certain airplanes) for determining the applicability of the supplemental NPRM, and as an additional source of service information for accomplishing the required replacement. The FAA also has revised the cost impact information and paragraph (b) of the supplemental NPRM according to the revised information specified in Revision 01 of the alert service bulletin.

# **Request To Reference Correct Type of Pin**

One commenter points out that the pin specified in the referenced alert service bulletin is not stainless steel, but rather a corrosion-resistant steel pin. The commenter states that a solid pin in lieu of the current roll pin would not be of any benefit in preventing corrosion since both the existing pin [part number (P/N) MS39086-140] and the proposed solid pin (P/N MS16555-628) are made of the same material (410 cress steel). The FAA acknowledges that the pin specified in the referenced alert service bulletin is not stainless steel. The FAA has consulted with Boeing and determined that the alert service bulletin incorrectly describes the subject pin as "solid stainless steel." Therefore, the FAA has revised paragraph (a) of the supplemental NPRM to read "solid corrosion-resistant pins" instead of "solid stainless steel pins."

# Conclusion

Since these changes expand the scope of the originally proposed rule, the FAA has determined that it is necessary to reopen the comment period to provide additional opportunity for public comment.

### **Cost Impact**

There are approximately 2,167 McDonnell Douglas Model DC-9 series airplanes, and Model MD-88 and MD-90-30 airplanes of the affected design in the worldwide fleet. The FAA estimates that 1,200 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 2 work hours per airplane to accomplish the proposed actions, and that the average labor rate is \$60 per work hour. Required parts would cost approximately \$2 per airplane. Based on these figures, the cost impact of the proposed AD on U.S. operators is estimated to be \$146,400, or \$122 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

# **Regulatory Impact**

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

## The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

McDonnell Douglas: Docket 97–NM–244–

Applicability: Model DC-9 series airplanes and Model MD-88 airplanes, as listed in McDonnell Douglas Alert Service Bulletin DC9-25A357, Revision 01, dated March 16, 1998; and Model MD-90-30 airplanes, as listed in McDonnell Douglas Alert Service Bulletin MD90-25A019, dated February 11, 1997; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent the improper deployment of the evacuation slide, which could delay or impede evacuation of passengers during an emergency, accomplish the following:

(a) Within 180 days after the effective date of this AD, replace the lanyard assembly pins of the evacuation slides with solid corrosion-resistant pins, in accordance with McDonnell Douglas Alert Service Bulletin MD80–25A357, dated February 11, 1997, or McDonnell Douglas Alert Service Bulletin DC9–25A357, Revision 01, dated March 16, 1998 (for Model DC–9 series airplanes and Model MD–88 airplanes); or McDonnell Douglas Alert Service Bulletin MD90–25A–019, dated February 11, 1997 (for Model MD–90–30 airplanes); as applicable.

(b) As of the effective date of this AD, no lanyard assembly, part number (P/N) 3961899–1 or P/N 3956939–501, shall be installed on any airplane unless that assembly has been modified in accordance with the requirements of paragraph (a) of this AD

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on December 14, 1998.

### Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 98–33537 Filed 12–17–98; 8:45 am] BILLING CODE 4910–13–P

### **DEPARTMENT OF THE TREASURY**

**Internal Revenue Service** 

26 CFR Parts 1 and 35

[REG-118662-98]

RIN 1545-AW78

### **New Technologies in Retirement Plans**

**AGENCY:** Internal Revenue Service (IRS), Treasury.

**ACTION:** Notice of proposed rulemaking and notice of public hearing.

**SUMMARY:** This document contains proposed amendments to the regulations governing certain notices and consent required in connection with distributions from retirement plans. Specifically, these proposed regulations set forth applicable standards for the transmission of those notices and consent through electronic media and modify the timing requirements for providing certain distribution-related notices. The proposed regulations provide guidance to plan sponsors and administrators by interpreting the notice and consent requirements in the context of the electronic administration of retirement plans. The proposed regulations affect retirement plan sponsors, administrators, and participants. This document also provides notice of a public hearing on these proposed regulations.

**DATES:** Written comments must be received by March 18, 1999. Outlines of topics to be discussed at the public hearing scheduled for April 15, 1999, at 10 a.m. must be received by March 25, 1999.

ADDRESSES: Send submissions to: CC:DOM:CORP:R (REG-118662-98), Room 5226, Internal Revenue Service, POB 7604, Ben Franklin Station, Washington, DC. 20044. Submissions