(2) The airplane may be operated in accordance with the provisions and limitations specified in an operator's FAA-approved Minimum Equipment List (MEL), provided that no more than one thrust reverser on the airplane is inoperative.

#### **New Requirements of This AD**

- (d) Within 1,000 hours time-in-service after the most recent test of the CDU cone brake performed in accordance with paragraph (a) of this AD, or within 650 hours time-inservice after the effective date of this AD, whichever occurs first: Perform a functional test to detect discrepancies of the CDU cone brake on each thrust reverser, in accordance with Boeing Service Bulletin 747-78A2166, Revision 1, dated October 9, 1997, or paragraph III.B. of the Accomplishment Instructions of Boeing Alert Service Bulletin 747-78A2130, dated May 26, 1994. Repeat the functional test thereafter at the interval specified in paragraph (d)(1) or (d)(2) of this AD, as applicable. Accomplishment of such functional test constitutes terminating action for the repetitive test of the CDU cone brake required by paragraph (a) of this AD.
- (1) For airplanes equipped with thrust reversers NOT modified in accordance with Boeing Service Bulletin 747–78–2144, Revision 1, dated April 11, 1996: Repeat the functional test at intervals not to exceed 650 hours time-in-service.
- (2) For airplanes equipped with thrust reversers modified in accordance with Boeing Service Bulletin 747–78–2144, Revision 1, dated April 11, 1996: Repeat the functional test at intervals not to exceed 1,000 hours time-in-service.
- (e) If any functional test required by paragraph (d) of this AD cannot be successfully performed, or if any discrepancy is found during any functional test required by paragraph (d) of this AD, accomplish either paragraph (e)(1) or (e)(2) of this AD.
- (1) Prior to further flight, correct the discrepancy found, in accordance with Boeing Service Bulletin 747–78A2166, Revision 1, dated October 9, 1997, or paragraph III.B. of the Accomplishment Instructions of Boeing Alert Service Bulletin 747–78A2130, dated May 26, 1994. Or
- (2) The airplane may be operated in accordance with the provisions and limitations specified in the operator's FAA-approved MEL, provided that no more than one thrust reverser on the airplane is inoperative.

# **Alternative Methods of Compliance**

(f) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle Aircraft Certification Office, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

#### **Special Flight Permits**

(g) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

#### **Incorporation by Reference**

- (h) Except as provided by paragraphs (c)(2) and (e)(2) of this AD, the actions shall be done in accordance with Boeing Alert Service Bulletin 747–78A2130, dated May 26, 1994, or Boeing Service Bulletin 747–78A2166, Revision 1, dated October 9, 1997, as applicable.
- (1) The incorporation by reference of Boeing Service Bulletin 747–78A2166, Revision 1, dated October 9, 1997, is approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51.
- (2) The incorporation by reference of Boeing Alert Service Bulletin 747–78A2130, dated May 26, 1994, was approved previously by the Director of the Federal Register as of April 13, 1995 (60 FR 13623, March 14, 1995).
- (3) Copies may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124–2207. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.
- (i) This amendment becomes effective on August 25, 1999.

Issued in Renton, Washington, on July 12, 1999.

#### D. L. Riggin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 99–18198 Filed 7–20–99; 8:45 am] BILLING CODE 4910–13–P

#### **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

### 14 CFR Part 39

[Docket No. 99-NM-113-AD; Amendment 39-11230; AD 99-15-10]

RIN 2120-AA64

# Airworthiness Directives; Boeing Model 777 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule; request for

comments.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD) that is applicable to certain Boeing Model 777 series airplanes. This action requires replacement of a certain engine-driven pump (EDP) supply shutoff valve, which is located in the aft strut fairing, with a new shutoff valve. This

amendment is prompted by reports of failure of the shutoff valve due to corrosion in the direct current motor in the shutoff valve. The actions specified in this AD are intended to prevent failure of an EDP supply shutoff valve. Such failure, in the event of an engine fire, could result in an uncontrolled fire in the engine compartment.

**DATES:** Effective August 5, 1999.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of August 5, 1999.

Comments for inclusion in the Rules Docket must be received on or before September 20, 1999.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 99-NM-113-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

The service information referenced in this AD may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124–2207. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

# FOR FURTHER INFORMATION CONTACT:

Kenneth W. Frey, Aerospace Engineer, Systems and Equipment Branch, ANM– 130S; FAA, Transport Airplane Directorate, Seattle Aircraft Certification Office (ACO), 1601 Lind Avenue, SW., Renton, Washington; telephone (425) 227–2673; fax (425) 227–1181.

SUPPLEMENTARY INFORMATION: The FAA has received reports of failures of the engine-driven pump (EDP) supply shutoff valves located in the aft strut fairings. Subsequently, the airplane manufacturer investigated this failure mode and reported to the FAA that failure of the supply shutoff valves was caused by corrosion in the direct current (DC) motors in the valves. Such corrosion forms between the stator and rotor in the DC motor in the supply shutoff valve assembly. Since the DC motor drives the actuator in the motoroperated supply shutoff valve to the commanded position, corrosion in the motor prevents the motor and the actuator from operating. In the event of an engine fire, failure of an EDP supply shutoff valve, if not corrected, could result in an uncontrolled fire in the engine compartment.

# **Explanation of Relevant Service Information**

The FAA has reviewed and approved Boeing Alert Service Bulletin 777–29A0022, Revision 1, dated May 21, 1999, which describes procedures for replacement of a certain EDP supply shutoff valve with a new EDP supply shutoff valve. The Boeing alert service bulletin references Parker-Hannifin Service Bulletin 2910044–29–48, "Introduction of Motor Operated Shutoff Valve, Part Number (P/N) 2960034–101," as an additional source of service information to accomplish the replacement.

# **Explanation of the Requirements of the Rule**

Since an unsafe condition has been identified that is likely to exist or develop on other Boeing Model 777 series airplanes of the same type design, this AD is being issued to prevent failure of an EDP supply shutoff valve due to corrosion in the DC motor in the shutoff valve. In the event of an engine fire, failure of an EDP supply shutoff valve could result in an uncontrolled fire in the engine compartment. This AD requires replacement of a certain EDP supply shutoff valve, which is located in the aft strut fairing, with a new shutoff valve.

# Differences Between the AD and the Alert Service Bulletin

Boeing Alert Service Bulletin 777-29A0022, Revision 1, specifies that the replacement actions required by this AD may be accomplished in accordance with the procedures specified in the alert service bulletin, or in accordance with an "operator's equivalent procedure. However, this AD requires that the actions be accomplished in accordance with the procedures specified in the Boeing alert service bulletin. An "operator's equivalent procedure" may be used only if approved as an alternative method of compliance in accordance with paragraph (c) of this AD.

#### **Determination of Rule's Effective Date**

Since a situation exists that requires the immediate adoption of this regulation, it is found that notice and opportunity for prior public comment hereon are impracticable, and that good cause exists for making this amendment effective in less than 30 days.

# **Comments Invited**

Although this action is in the form of a final rule that involves requirements affecting flight safety and, thus, was not preceded by notice and an opportunity for public comment, comments are

invited on this rule. Interested persons are invited to comment on this rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified under the caption ADDRESSES. All communications received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this rule must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 99–NM–113–AD." The postcard will be date stamped and returned to the commenter.

### **Regulatory Impact**

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

The FAA has determined that this regulation is an emergency regulation that must be issued immediately to correct an unsafe condition in aircraft, and that it is not a "significant regulatory action" under Executive Order 12866. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final

regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

# List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### **Adoption of the Amendment**

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

**99–15–10 BOEING:** Amendment 39–11230. Docket 99–NM–113–AD.

Applicability: Model 777 series airplanes, equipped with an engine-driven pump supply shutoff valve having Boeing part number S271W741–21; certificated in any category.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent failure of an engine-driven pump (EDP) supply shutoff valve, which, in the event of an engine fire, could result in an uncontrolled fire in the engine compartment, accomplish the following:

### Replacement

(a) Except as provided by paragraph (b) of this AD, within 90 days after the effective date of this AD, replace any EDP supply shutoff valve, Boeing part number (P/N) S271W741–21, that is located in each aft strut fairing, with a new EDP supply shutoff valve, Boeing P/N S271W741–22, in accordance with Boeing Alert Service Bulletin 777–29A0022, Revision 1, dated May 21, 1999.

(b) Where Boeing Alert Service Bulletin 777–29A0022, Revision 1, dated May 21, 1999, specifies that replacements may be accomplished in accordance with an operator's "equivalent procedure," those actions must be accomplished in accordance with the applicable chapter of the Boeing 777 Airplane Maintenance Manual (AMM) specified in the alert service bulletin.

#### **Alternative Methods of Compliance**

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Seattle Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

#### **Special Flight Permits**

(d) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

#### **Incorporation by Reference**

(e) The replacement shall be done in accordance with Boeing Alert Service Bulletin 777–29A0022, Revision 1, dated May 21, 1999. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124–2207. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(f) This amendment becomes effective on August 5, 1999.

Issued in Renton, Washington, on July 13,

### D. L. Riggin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 99–18365 Filed 7–20–99; 8:45 am] BILLING CODE 4910–13–P

#### **DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration** 

14 CFR Part 71

[Airspace Docket No. 99-ACE-36]

Amendment to Class E Airspace; Parsons, KS

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Direct final rule; request for comments.

**SUMMARY:** This action amends Class E airspace area at Tri-City Airport, Parsons, KS. A review of the Class E airspace area for Tri-City Airport indicates it does not comply with the criteria for 700 feet Above Ground Level (AGL) airspace required for diverse departures as specified in FAA Order 7400.2D. The Class E airspace has been enlarged to conform to the criteria of FAA Order 7400.2D.

In addition, a minor revision to the Airport Reference Point (ARP) coordinates is included in this document.

The intended effect of this rule is to provide additional controlled Class E airspace for aircraft operating under Instrument Flight Rules (IFR), revise the ARP, and comply with the criteria of FAA Order 7400.2D.

**DATES:** Effective date: 0901 UTC, November 4, 1999.

Comments for inclusion in the Rules Docket must be received on or before August 15, 1999.

ADDRESSES: Send comments regarding the rule in triplicate to: Manager, Airspace Branch, Air Traffic Division, ACE-520, Federal Aviation Administration, Docket Number 99– ACE-36, 601 East 12th Street, Kansas City, MO 64106.

The official docket may be examined in the Office of the Regional Counsel for the Central Region at the same address between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

An informal docket may also be examined during normal business hours in the Air Traffic Division at the same address listed above.

### FOR FURTHER INFORMATION CONTACT:

Kathy Randolph, Air Traffic Division, Airspace Branch, ACE–520C, Federal Aviation Administration, 601 East 12th Street, Kansas City, MO 64106; telephone: (816) 426–3408.

**SUPPLEMENTARY INFORMATION: This** amendment to 14 CFR 71 revises the Class E airspace at Parsons, KS. A review of the Class E airspace for Tri-City Airport, KS, indicates it does not meet the criteria for 700 feet AGL airspace required for diverse departures as specified in FAA Order 7400.2D. The criteria in FAA Order 7400.2D for an aircraft to reach 1200 feet AGL is based on a standard climb gradient of 200 feet per mile plus the distance from the ARP to the end of the outermost runway. Any fractional part of a mile is coverted to the next higher tenth of a mile. The amendment at Parsons, Tri-City Airport,

KS, will provide additional controlled airspace for aircraft operating under IFR, revise the ARP, and comply with the criteria of FAA Order 7400.2D. The area will be depicted on appropriate aeronautical charts. Class E airspace areas extending upward from 700 feet or more above the surface of the earth are published in paragraph 6005 of FAA Order 7400.9F, dated September 10, 1998, and effective September 16, 1998, which is incorporated by reference in 14 CFR 71.1. The Class E airspace designation listed in this document will be published subsequently in the Order.

#### The Direct Final Rule Procedure

The FAA anticipates that this regulation will not result in adverse or negative comment and, therefore, is issuing it as a direct final rule. Previous actions of this nature have not been controversial and have not resulted in adverse comments or objections. The amendment will enhance safety for all flight operations by designating an area where VFR pilots may anticipate the presence of IFR aircraft at lower altitudes, especially during inclement weather conditions. A greater degree of safety is achieved by depicting the area on aeronautical charts. Unless a written adverse or negative comment, or a written notice of intent to submit an adverse or negative comment is received within the comment period, the regulation will become effective on the date specified above. After the close of the comment period, the FAA will publish a document in the **Federal Register** indicating that no adverse or negative comments were received and confirming the date on which the final rule will become effective. If the FAA does receive, within the comment period, an adverse or negative comment, or written notice of intent to submit such a comment, a document withdrawing the direct final rule will be published in the **Federal Register**, and a notice of proposed rulemaking may be published with a new comment period.

#### **Comments Invited**

Although this action is in the form of a final rule and was not preceded by a notice of proposed rulemaking, comments are invited on this rule. Interested persons are invited to comment on this rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified under the caption ADDRESS. All communications received on or before the closing date for comments will be considered, and this rule may be amended or withdrawn