

further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by removing amendment 39–11914 (65 FR 57944, September 27, 2000), and by adding a new airworthiness directive (AD), amendment 39–11928, to read as follows:

**2000–20–03 R1 Bombardier, Inc. (Formerly Canadair):** Amendment 39–11928. Docket 2000–NM–312–AD. Revises AD 2000–20–03, Amendment 39–11914.

**Applicability:** Model CL–600–2B19 series airplanes, serial numbers 7003 through 7323 inclusive, certificated in any category.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent reduced controllability of the airplane due to an accumulation of water, ice, and slush on the aileron quadrants and control cable pulleys in the wheel bay of the

main landing gear (MLG); accomplish the following:

#### Installation

(a) Within 45 days after October 2, 2000 (the effective date of AD 2000–20–03), install splash shields in the wheel bay of the MLG in accordance with Bombardier Service Bulletin 601R–27–104, dated October 15, 1999.

#### Alternative Methods of Compliance

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, New York Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, New York ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the New York ACO.

#### Special Flight Permits

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

#### Incorporation by Reference

(d) The installation shall be done in accordance with Bombardier Service Bulletin 601R–27–104, dated October 15, 1999. This incorporation by reference was approved previously by the Director of the **Federal Register** as of October 2, 2000 (65 FR 57944, September 27, 2000). Copies may be obtained from Bombardier Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, New York Aircraft Certification Office, 10 Fifth Street, Third Floor, Valley Stream, New York 11581; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Note 3:** The subject of this AD is addressed in Canadian airworthiness directive CF–2000–28, dated August 28, 2000.

#### Effective Date

(e) This amendment becomes effective on October 2, 2000.

Issued in Renton, Washington, on October 4, 2000.

**Donald L. Riggins,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 00–26092 Filed 10–13–00; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 2000–SW–25–AD; Amendment 39–11931; AD 2000–20–19]

**RIN 2120–AA64**

#### Airworthiness Directives; Eurocopter France Model AS–350B, BA, B1, B2, B3, C, D, and D1, and AS–355E, F, F1, F2 and N Helicopters

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment supersedes an existing airworthiness directive (AD) for Eurocopter France Model AS–350B, BA, B1, B2, C, D, and D1, and AS–355E, F, F1, and F2 helicopters. That AD currently requires inspections of the main rotor head components, the main gearbox (MGB) suspension bars, and the ground resonance prevention system components. This amendment requires those same inspections, but would also apply to Model AS–350B3 and AS–355N helicopters. This amendment is prompted by the inadvertent omission of those model helicopters from the previous AD. The actions specified by this AD are intended to prevent ground resonance due to reduced structural stiffness, which could lead to failure of a main rotor head or MGB suspension component and subsequent loss of control of the helicopter.

**DATES:** Effective November 20, 2000.

The incorporation by reference of certain publications listed in the regulations was approved previously by the Director of the Federal Register as of May 23, 2000 (65 FR 20721, April 18, 2000).

**ADDRESSES:** The service information referenced in this AD may be obtained from American Eurocopter Corporation, 2701 Forum Drive, Grand Prairie, Texas 75053–4005, telephone (972) 641–3460, fax (972) 641–3527. This information may be examined at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Jim Grigg, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Rotorcraft Standards Staff, 2601 Meacham Blvd., Fort Worth, Texas 76137, telephone (817) 222–5490, fax (817) 222–5961.

**SUPPLEMENTARY INFORMATION:** A proposal to amend part 39 of the Federal

Aviation Regulations (14 CFR part 39) by superseding AD 86-15-10 R2, Amendment 39-11681 (65 FR 20721), which applies to Eurocopter France Model AS-350B, BA, B1, B2, B3, C, D, and D1, and AS-355E, F, F1, F2 and N helicopters was published in the **Federal Register** on July 20, 2000 (65 FR 44995). That action proposed to require an initial inspection within 10 hours time-in-service (TIS) and thereafter, repetitive inspections of the main rotor head components, the MGB suspension bars, and the ground resonance prevention system components at intervals not to exceed 500 hours TIS.

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were received on the proposal or the FAA's determination of the cost to the public. The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

The FAA estimates that 586 helicopters of U.S. registry will be affected by this AD, that it will take approximately 8 work hours per helicopter to accomplish the required actions, and that the average labor rate is \$60 per work hour. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$480 per helicopter, or \$281,280 for the entire fleet.

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by removing Amendment 39-11681 (65 FR 20721, April 18, 2000), Amendment 39-6515 (55 FR 5833, February 20, 1990) and Amendment 39-5517 (52 FR 13233, April 22, 1987) and by adding a new airworthiness directive (AD), Amendment 39-11931, to read as follows:

#### 2000-20-19 Eurocopter France:

Amendment 39-11931. Docket No. 2000-SW-25-AD. Supersedes AD 86-15-10 R2, Amendment 39-11681, Docket No. 98-SW-82-AD; AD 86-15-10R1, Amendment 39-6515, Docket No. 86-ASW-22; and AD 86-15-10, Amendment 39-5517, Docket No. 86-ASW-22.

**Applicability:** Model AS-350B, BA, B1, B2, B3, C, D, and D1, and AS-355E, F, F1, F2 and N helicopters, certificated in any category.

**Note 1:** This AD applies to each helicopter identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For helicopters that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (f) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent ground resonance due to reduced structural stiffness, which could lead to failure of a main rotor head or main gearbox (MGB) suspension component and subsequent loss of control of the helicopter, accomplish the following:

(a) Within 10 hours time-in-service (TIS):

(1) For Model AS-350B, BA, B1, B2, B3, C, D, and D1 helicopters, inspect the main rotor head components, the MGB suspension bars (struts), and the landing gear ground resonance prevention components (aft spring blades and hydraulic shock absorbers) in accordance with paragraph CC.3 of Aerospatiale Service Bulletin (SB) No. 01.17a (not dated).

(2) For Model AS-355E, F, F1, F2, and N helicopters, inspect the main rotor head components, the MGB suspension bars (struts), and the landing gear ground resonance prevention components (aft spring blades and hydraulic shock absorbers) in accordance with paragraph CC.3 of SB No. 01.14a (not dated).

(b) Rework or replace damaged components in accordance with SB No. 01.17a or SB No. 01.14a, as applicable.

(c) Repeat the inspections and rework required by paragraphs (a) and (b) of this AD at intervals not to exceed 500 hours TIS.

(d) If the helicopter is subjected to a hard landing or to high surface winds when parked without effective tiedown straps installed, repeat the inspections required by paragraph (a) of this AD for the main rotor head star arms and the MGB suspension bars (struts) before further flight.

(e) After a landing with abnormal self-sustained dynamic vibrations (ground resonance type vibrations), repeat all the inspections required by paragraph (a) of this AD.

(f) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Regulations Group, Rotorcraft Directorate, FAA. Operators shall submit their requests through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, Regulations Group.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Regulations Group.

(g) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the helicopter to a location where the requirements of this AD can be accomplished.

(h) The inspections shall be done in accordance with paragraph CC.3 of Aerospatiale Service Bulletin No. 01.17a (not dated) or No. 01.14a (not dated), as applicable. The rework or replacement, if necessary, shall be done in accordance with Aerospatiale Service Bulletin No. 01.17a or No. 01.14a (neither is dated), as applicable. The incorporation by reference of those documents was previously approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51 as of May 23, 2000 (65 FR 20721, April 18, 2000). Copies may be obtained from American Eurocopter Corporation, 2701 Forum Drive, Grand Prairie, Texas 75053-4005, telephone (972) 641-3460, fax (972) 641-3527. Copies may be inspected at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(i) This amendment becomes effective on November 20, 2000.

Issued in Fort Worth, Texas, on September 29, 2000.

**Henry A. Armstrong,**  
*Manager, Rotorcraft Directorate, Aircraft Certification Service.*

[FR Doc. 00-26234 Filed 10-13-00; 8:45 am]

BILLING CODE 4910-13-U

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 71

[Airspace Docket No. 00-ACE-22]

#### Amendment to Class E Airspace; Elkhart, KS

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Direct final rule; confirmation of effective date.

**SUMMARY:** This document confirms the effective date of a direct final rule which revises Class E airspace at Elkhart, KS.

**DATES:** The direct final rule published at 65 FR 45707 is effective on 0901 UTC, November 30, 2000.

**FOR FURTHER INFORMATION CONTACT:** Kathy Randolph, Air Traffic Division, Airspace Branch, ACE-520C, DOT Regional Headquarters Building, Federal Aviation Administration, 901 Locust, Kansas City, MO 64106; telephone: (816) 329-2525.

**SUPPLEMENTARY INFORMATION:** The FAA published this direct final rule with a request for comments in the **Federal Register** on July 25, 2000 (65 FR 45707). The FAA uses the direct final rulemaking procedure for a non-controversial rule where the FAA believes that there will be no adverse public comment. This direct final rule advised the public that no adverse comments were anticipated, and that unless a written adverse comment, or a written notice of intent to submit such an adverse comment, were received within the comment period, the regulation would become effective on November 30, 2000. No adverse comments were received, and thus this document confirms that this direct final rule will become effective on that date.

Issued in Kansas City, MO on September 28, 2000.

**Herman J. Lyons, Jr.,**  
*Manager, Air Traffic Division, Central Region.*  
[FR Doc. 00-26525 Filed 10-13-00; 8:45 am]

BILLING CODE 4910-13-M

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 71

[Airspace Docket No. 00-ACE-21]

#### Amendment to Class E Airspace; Columbia, MO

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Direct final rule; confirmation of effective date.

**SUMMARY:** This document confirms the effective date of a direct final rule which revises Class E airspace at Columbia, MO.

**DATES:** The direct final rule published at 65 FR 43686 is effective on 0901 UTC, November 30, 2000.

**FOR FURTHER INFORMATION CONTACT:** Brenda Mumper, Air Traffic Division, Airspace Branch, ACE-520A, DOT Regional Headquarters Building, Federal Aviation Administration, 901 Locust, Kansas City, MO 64106; telephone: (816) 329-2524.

**SUPPLEMENTARY INFORMATION:** The FAA published this direct final rule with a request for comments in the **Federal Register** on July 14, 2000 (65 FR 43686). The FAA uses the direct final rulemaking procedure for a non-controversial rule where the FAA believes that there will be no adverse public comment. This direct final rule advised the public that no adverse comments were anticipated, and that unless a written adverse comment, or a written notice of intent to submit such an adverse comment, were received within the comment period, the regulation would become effective on November 30, 2000. No adverse comments were received, and thus this document confirms that this direct final rule will become effective on that date.

Issued in Kansas City, MO on September 28, 2000.

**Herman J. Lyons, Jr.,**  
*Manager, Air Traffic Division, Central Region.*  
[FR Doc. 00-26526 Filed 10-13-00; 8:45 am]

BILLING CODE 4910-13-M

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 71

[Airspace Docket No. 00-ASW-6]

#### Amendment of Federal Airways in the Vicinity of Dallas/Fort Worth; TX

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** This action amends thirteen Federal airways in the vicinity of Dallas/Fort Worth, TX. In the NPRM for this action, the FAA proposed to modify 14 airways. However, after considering aircraft routing through the Shreveport, LA, Approach Control terminal airspace and retention of transition fixes for the Shreveport Regional Airport and Barksdale Air Force Base (AFB), the FAA has decided not to make the proposed modification of V-566. The FAA is taking this action to simplify the airway structure, thereby, enhancing the management of aircraft operations in the area.

**EFFECTIVE DATE:** 0901 UTC, November 30, 2000.

**FOR FURTHER INFORMATION CONTACT:** Bil Nelson, Airspace and Rules Division, ATA-400, Office of Air Traffic Airspace Management, Federal Aviation Administration, 800 Independence Avenue, SW., Washington, DC 20591; telephone: (202) 267-8783.

#### SUPPLEMENTARY INFORMATION:

##### Background

As a result of a recent airspace review, the FAA has determined that airways in the vicinity of Dallas/Fort Worth, TX, should be amended to improve management of the aircraft operations. On June 12, 2000, the FAA published in the **Federal Register** (65 FR 36805) a proposal to amend fourteen Federal airways. However, after considering aircraft routing through the Shreveport, LA, Approach Control terminal airspace and retention of transition fixes for the Shreveport Regional Airport and Barksdale AFB, the FAA decided not to make the proposed modification of V-566.

Interested parties were invited to participate in this proposed rulemaking by submitting such written data, views, or arguments as they may desire. No comments were received. Except for editorial changes, and the decision not to amend V-566, this amendment is the same as that proposed in the NPRM.