

Appendix 3—AD Compliance Inspection Report (Sample Format), Bell Model 47 Main Rotor Blade Grip

Provide the following information and mail or fax it to: Manager, Rotorcraft Certification Office, Federal Aviation Administration, Fort Worth, Texas, 76193-0170, USA, Fax: 817-222-5783.

Aircraft Registration No: _____

Helicopter Model: _____

Helicopter Serial Number: _____

Owner and Operator of the Helicopter: _____

	Grip #1	Grip #2
Part Number: Serial Number: Hours TIS on the part at Inspection: Crack Found (Y/N) (If yes, describe below.)		

Description of Findings

Who performed the inspections? _____

If a crack was found, describe the crack size, location, and orientation (provide a sketch or pictures with the grip part and serial number). _____

Provide any other comments. _____

Issued in Fort Worth, Texas, on March 23, 2001.

Mark R. Schilling,

Acting Manager, Rotorcraft Directorate,
Aircraft Certification Service.

[FR Doc. 01-7741 Filed 3-28-01; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-236-AD]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 767-300 Series Airplanes Modified by Supplemental Type Certificate ST00118SE

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to all Boeing Model 767-300 series airplanes modified by supplemental type certificate ST00118SE. This proposal would require modification of the in-flight entertainment (IFE) system and revision of the Airplane Flight Manual. This action is necessary to ensure that the flight crew is able to remove electrical power from the IFE system when necessary and is advised of

appropriate procedures for such action. Inability to remove power from the IFE system during a non-normal or emergency situation could result in inability to control smoke or fumes in the airplane flight deck or cabin. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by May 14, 2001.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2000-NM-236-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 2000-NM-236-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

The service information referenced in the proposed rule may be obtained from Matsushita Avionics System Corporation, 22333 29th Drive SE, Bothell, Washington 98021. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT:

Stephen S. Oshiro, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2793; fax (425) 227-1181.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.

- For each issue, state what specific change to the proposed AD is being requested.

- Include justification (e.g., reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2000-NM-236-AD." The postcard will be date-stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2000-NM-236-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The Federal Aviation Administration (FAA) recently completed a review of in-flight entertainment (IFE) systems certified by supplemental type certificate (STC) and installed on transport category airplanes. The review focused on the interface between the IFE system and airplane electrical system, with the objective of determining if any unsafe conditions exist with regard to the interface. STC's issued between 1992 and 2000 were considered for the review.

The type of IFE systems considered for review were those that contain video monitors (cathode ray tubes or liquid crystal displays; either hanging above the aisle or mounted on individual seat backs or seat trays), or complex circuitry (i.e., power supplies, electronic distribution boxes, extensive wire routing, relatively high power consumption, multiple layers of circuit protection, etc.). In addition, in-seat power supply systems that provide power to more than 20 percent of the total passenger seats were also considered for the review. The types of

IFE systems not considered for review include systems that provide only audio signals to each passenger seat, ordinary in-flight telephone systems (e.g., one telephone handset per group of seats or bulkhead-mounted telephones), systems that only have a video monitor on the forward bulkhead(s) (or a projection system) to provide passengers with basic airplane and flight information, and in-seat power supply systems that provide power to less than 20 percent of the total passenger seats.

Items considered during the review include the following:

- Can the electrical bus(es) supplying power to the IFE system be deenergized when necessary without removing power from systems that may be required for continued safe flight and landing?
- Can IFE system power be removed when required without pulling IFE system circuit breakers? [i.e., is there a switch (dedicated to the IFE system or a combination of loads) located in the flight deck or cabin that can be used to remove IFE power?]
- If the IFE system requires changes to flight crew procedures, has the airplane flight manual (AFM) been properly amended?
- If the IFE system requires changes to cabin crew procedures, have they been properly amended?
- Does the IFE system require periodic or special maintenance?

In all, approximately 180 IFE systems approved by STC were reviewed by the FAA. The review results indicate that potential unsafe conditions exist on some IFE systems installed on various transport category airplanes. These conditions can be summarized as:

- Electrical bus(es) supplying power to the IFE system cannot be deenergized when necessary without removing power from systems that may be required for continued safe flight and landing.
- Power cannot be removed from the IFE system when required without pulling IFE system circuit breakers (i.e., there is no switch dedicated to the IFE

system or combination of systems for the purpose of removing power).

- Installation of the IFE system has affected crew (flight crew and/or cabin crew) procedures, but the procedures have not been properly revised.

FAA's Determination

As part of its review of IFE systems, the FAA has determined that an unsafe condition exists on Boeing Model 767–300 series airplanes modified by STC ST00118SE. The IFE system on these airplanes is connected to an electrical bus that cannot be deactivated without also removing power from airplane systems necessary for safe flight and landing. Additionally, the airplane manufacturer's published flight crew and cabin crew emergency procedures do not provide for the removal of power from the IFE system. This condition, if not corrected, could result in inability to remove power from the IFE system during a non-normal or emergency situation, and consequent inability to control smoke or fumes in the airplane flight deck or cabin.

Explanation of Relevant Service Information

The FAA has reviewed and approved Matsushita Avionics Systems Corporation Service Bulletin S2GFAB767–23–1, dated February 7, 2001, which describes procedures for modification of the IFE system. This modification involves installation of additional wiring and a new relay so that the “master power switch” will remove power from the IFE system. (Currently, the “master power switch” removes power from high-energy components, but does not remove power from the overhead video, passenger service, and economy-class audio systems.)

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would

require accomplishment of the actions specified in the service bulletin described previously, except as discussed in the “Differences Between Service Bulletin and This Proposed AD” section, below. The proposed AD also would require revising procedures to be followed in the event of “Electrical Smoke or Fire,” as contained in the “Emergency Procedures” section of the Airplane Flight Manual, to ensure that the flight crew is advised of instructions for removing power from the IFE system during an emergency related to smoke or fumes. Accomplishment of these actions is intended to adequately address the identified unsafe condition.

Differences Between Service Bulletin and This Proposed AD

Operators should note that the service bulletin states that the actions are to be completed “at the earliest opportunity where manpower and facilities are available.” The FAA finds that such a compliance time will not ensure that the proposed actions are accomplished in a timely manner.

In developing an appropriate compliance time for this action, the FAA considered not only the degree of urgency associated with addressing the subject unsafe condition, but the amount of time necessary to accomplish the proposed actions, and the practical aspect of accomplishing the proposed actions within an interval of time that parallels normal scheduled maintenance for the affected operators. In consideration of these factors, the FAA has determined that 18 months after the effective date of this AD represents an appropriate interval of time allowable wherein an acceptable level of safety can be maintained.

Other Relevant Proposed Rulemaking

This proposed action is one of a number of proposed AD's on airplanes modified by STC's that have been determined to be subject to similar unsafe conditions. Other currently proposed AD's include the following airplanes and STC's:

Model/Series	STC No.	Docket No.
Boeing 757–200	SA1727GL	2000–NM–228–AD
McDonnell Douglas DC–9–51 and DC–9–83	SA8026NM	2000–NM–229–AD
McDonnell Douglas DC–10–30	ST00054SE	2000–NM–231–AD
Boeing 767–300 and 767–300ER	SA5765NM	2000–NM–232–AD
	SA5978NM	
Boeing 767–300	ST00157SE	2000–NM–233–AD
Boeing 747–100 and –200	ST00196SE	2000–NM–234–AD
Boeing 767–200	SA5134NM	2000–NM–235–AD
Boeing 737–300	ST00171SE	2000–NM–237–AD
Boeing 767–200	SA4998NM	2000–NM–238–AD
Boeing 767–300	SA7019NM–D	2000–NM–239–AD
Boeing 747–100 and –200	SA8622SW	2000–NM–240–AD
McDonnell Douglas DC–10–30	SA8452SW	2000–NM–241–AD

Model/Series	STC No.	Docket No.
Boeing 737-700	ST09100AC-D ST09104AC-D ST09105AC-D ST09106AC-D	2000-NM-242-AD
Boeing 767-200	ST09022AC-D	2000-NM-243-AD
Boeing 747SP	ST09097AC-D	2000-NM-244-AD
Boeing 747-400	SA8843SW	2000-NM-245-AD
Airbus A340-211	ST0902AC-D	2000-NM-246-AD

Cost Impact

There are approximately 20 airplanes of the affected design in the worldwide fleet. The FAA estimates that no airplanes of U.S. registry would be affected by this proposed AD. All airplanes included in the applicability of this proposed AD currently are operated by non-U.S. operators under foreign registry; therefore, they are not directly affected by this AD action. However, the FAA considers that this AD is necessary to ensure that the unsafe condition is addressed in the event that any of these subject airplanes are imported and placed on the U.S. Register in the future.

Should an affected airplane be imported and placed on the U.S. Register in the future, it would take approximately 50 work hours per airplane to accomplish the proposed modification, at an average labor rate of \$60 per work hour. Required parts would be provided at no charge to the operator. Based on these figures, the cost impact of the proposed modification would be \$3,000 per airplane.

Should an affected airplane be imported and placed on the U.S. Register in the future, it would take approximately 1 work hour per airplane to accomplish the proposed AFM revision, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the proposed AFM revision would be \$60 per airplane.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this proposed AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Boeing: Docket 2000-NM-236-AD.

Applicability: Model 767-300 series airplanes modified by supplemental type

certificate (STC) ST00118SE; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To ensure that the flight crew is able to remove electrical power from the in-flight entertainment (IFE) system when necessary and is advised of appropriate procedures for such action; accomplish the following:

Modification and Airplane Flight Manual Revision

(a) Within 18 months after the effective date of this AD, accomplish paragraphs (a)(1) and (a)(2) of this AD.

(1) Modify the IFE system installed on the airplane in accordance with Matsushita Avionics Systems Corporation Service Bulletin S2GFAB767-23-1, dated February 7, 2001.

(2) Revise the procedures under "Electrical Smoke or Fire" in the "Emergency Procedures" section of the FAA-approved Airplane Flight Manual (AFM) to include the following information. This may be accomplished by inserting a copy of this AD into the AFM.

"If Smoke Source Cannot Be Located

Utility bus switches—OFF.

Establish communications with cabin crew.

Instruct cabin crew to place in-flight entertainment (IFE) system Master Power Switch in "OFF" position.

Obtain confirmation from cabin crew that electrical power to the IFE system has been removed."

Spares

(b) As of the effective date of this AD, no person shall install an IFE system in accordance with STC ST00118SE on any airplane, unless it is modified and the AFM is revised in accordance with this AD.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

Special Flight Permits

(d) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on March 23, 2001.

Vi L. Lipski,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 01-7740 Filed 3-28-01; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-234-AD]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 747-100 and -200 Series Airplanes Modified by Supplemental Type Certificate ST00196SE

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to all Boeing Model 747-100 and -200 series airplanes modified by supplemental type certificate ST00196SE. This proposal would require modification of the in-flight entertainment (IFE) system and revision of the Airplane Flight Manual. This action is necessary to ensure that the flight crew is able to remove electrical power from the IFE system when necessary and is advised of appropriate procedures for such action. Inability to remove power from the IFE system during a non-normal or emergency situation could result in inability to control smoke or fumes in the airplane flight deck or cabin. This action is intended to address the identified unsafe condition.

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The service information referenced in the proposed rule may be obtained from JAMCO America, Inc., 1018-80th Street SW, Everett, Washington 98023. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT:

Stephen S. Oshiro, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2793; fax (425) 227-1181.

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