- (i) Replace the DPV assembly with a serviceable assembly, and then perform a system test of the thrust reverser, or
- (ii) Deactivate the thrust reverser. However, the DPV assembly must be replaced with a serviceable assembly, and a system test of the thrust reverser performed within 10 days after deactivation.
- (2) Replace the DPV assembly with a serviceable assembly, and then perform a system test of the thrust reverser.
- (b) Thereafter, at intervals not to exceed 7,000 flight hours TIS since the last check or replacement of the DPV assembly, perform one of the following, in accordance with paragraphs 2.B and 2.C. of the Accomplishment Instructions of Middle River Aircraft Systems ASB No. 78A5053, dated October 30, 2001:
- (1) Perform a DPV assembly pressure check for leakage, and, if necessary, either:
- (i) Replace the DPV assembly with a serviceable assembly and then perform a system test of the thrust reverser, or
- (ii) Deactivate the thrust reverser. However, the DPV assembly must be replaced with a serviceable assembly, and a system test of the thrust reverser performed within 10 days after deactivation.
- (2) Replace the DPV assembly with a serviceable assembly and then perform a system test of the thrust reverser.

Definition of Serviceable DPV Assembly

(c) For the purpose of this AD, a serviceable DPV assembly is an assembly that has either accumulated zero TSN, or has accumulated zero TIS after having passed the tests in the Middle River Aircraft Systems Component Maintenance Manual GEK 85007 (78–31–51), Directional Pilot Solenoid Valve, Page Block 101, Testing and Troubleshooting, or has been successfully leak checked in accordance with paragraph 2.B. of the Accomplishment Instructions of Middle River Aircraft Systems ASB No. 78A5053, dated October 30, 2001.

Alternative Methods of Compliance

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Engine Certification Office (ECO). Operators must submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, ECO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the ECO.

Special Flight Permits

(e) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be done.

Incorporation by Reference

(f) The actions required by this AD must be done in accordance with Middle River Aircraft Systems CF6–80E1 Alert Service Bulletin (ASB) No. 78A5053, dated October 30, 2001. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Middle River Aircraft Systems, Mail Point 46, 103 Chesapeake Park Plaza, Baltimore, MD, 21220—4295, attn: Warranty Support, telephone: (410) 682–0094, fax: (410) 682–0100. Copies may be inspected, by appointment, at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA; or at the Office of the Federal Register, 800 North Capital Street, NW., suite 700, Washington, DC.

Effective Date

(g) This amendment becomes effective on May 1, 2002.

Issued in Burlington, Massachusetts, on March 14, 2002.

Francis A. Favara,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service. [FR Doc. 02–6912 Filed 3–26–02; 8:45 am]

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-CE-07-AD; Amendment 39-12687; AD 2002-06-10]

RIN 2120-AA64

Airworthiness Directives; Pilatus Aircraft Ltd. Models PC-12 and PC-12/ 45 Airplanes

AGENCY: Federal Aviation Administration, DOT. **ACTION:** Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain Pilatus Aircraft Ltd. (Pilatus) Models PC-12 and PC-12/45 airplanes. This AD requires you to replace the metered connector and oxygen tubing and related components in the rear seat bench. This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Switzerland. The actions specified by this AD are intended to correct the potential for insufficient oxygen quantity that is available to occupants of the rear seat bench in some emergency conditions. This condition could result in reduced occupant safety at the rear bench seat location.

DATES: This AD becomes effective on May 6, 2002.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulations as of May 6, 2002.

ADDRESSES: You may get the service information referenced in this AD from Pilatus Aircraft Ltd., Customer Liaison Manager, CH-6371 Stans, Switzerland; telephone: +41 41 619 63 19; facsimile: +41 41 619 6224; or from Pilatus Business Aircraft Ltd., Product Support Department, 11755 Airport Way, Broomfield, Colorado 80021; telephone: (303) 465–9099; facsimile: (303) 465– 6040. You may view this information at the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 2001-CE-07-AD, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4059; facsimile: (816) 329–4090.

SUPPLEMENTARY INFORMATION:

Discussion

What Events Have Caused This AD?

The Federal Office for Civil Aviation (FOCA), which is the airworthiness authority for Switzerland, recently notified FAA that an unsafe condition may exist on certain Pilatus Models PC–12 and PC–12/45 airplanes. The FOCA reports that, because of a design problem, the flow of oxygen to each occupant on the rear seat bench is insufficient. The current configuration uses two-metered connectors, which restricts the flow of oxygen.

What Is the Potential Impact if FAA Took No Action?

If not corrected, insufficient oxygen quantity that is available to occupants of the rear seat bench in some emergency conditions could occur. This condition could result in reduced occupant safety at the rear bench seat location.

Has FAA Taken Any Action to This Point?

We issued a proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Pilatus Models PC–12 and PC–12/45 airplanes. This proposal was published in the **Federal Register** as a notice of proposed rulemaking (NPRM) on January 2, 2002 (67 FR 29). The NPRM proposed to require you to replace the two-metered connector and oxygen tubing with a system that incorporates a singlemetered connector in the rear seat bench.

Was the Public Invited To Comment?

The FAA encouraged interested persons to participate in the making of this amendment. We did not receive any comments on the proposed rule or on our determination of the cost to the public.

FAA's Determination

What Is FAA's Final Determination on This Issue?

After careful review of all available information related to the subject

presented above, we have determined that air safety and the public interest require the adoption of the rule as proposed except for minor editorial corrections. We have determined that these minor corrections:

- —Provide the intent that was proposed in the NPRM for correcting the unsafe condition; and
- —Do not add any additional burden upon the public than was already proposed in the NPRM.

Cost Impact

How Many Airplanes Does This AD Impact?

We estimate that this AD affects 5 airplanes in the U.S. registry.

What Is the Cost Impact of This AD on Owners/Operators of the Affected Airplanes?

We estimate the following costs to accomplish the replacements:

Labor cost	Parts cost	Total cost per air- plane	Total cost on U.S. operators
2 workhours at \$60 per hour = \$120	Pilatus will provide free parts	\$120	\$600

Compliance Time of This AD

What Will Be the Compliance Time of This AD?

The compliance time of this AD is "within the next 30 calendar days after the effective date of this AD".

Why Is the Compliance Time Presented in Calendar Time Instead of Hours Time-in-Service (TIS)?

The oxygen flow on the rear bench seat is reduced through two metered connectors when only one reduction is necessary. Because these parts of poor design could have been installed in the field or at the factory, the problem has the same chance of occurring on an airplane with 50 hours TIS as one with 1,000 hours TIS. Therefore, we believe that 30 calendar days will:

- Ensure that the unsafe condition does not go undetected for a long period of time on the affected airplanes; and
- —Not inadvertently ground any of the affected airplanes.

Regulatory Impact

Does This AD Impact Various Entities?

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

Does This AD Involve a Significant Rule or Regulatory Action?

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the final evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. FAA amends § 39.13 by adding a new AD to read as follows:

2002–06–10 Pilatus Aircraft Ltd.: Amendment 39–12687; Docket No. 2001–CE–07–AD.

(a) What airplanes are affected by this AD? This AD affects the following airplane models and serial numbers with rear bench seats (part number 525.22.12.016) installed, that are certificated in any category:

Model	Serial Nos.
PC-12 and PC-12/45	From 101 through 365

(b) Who must comply with this AD? Anyone who wishes to operate any of the above airplanes must comply with this AD.

- (c) What problem does this AD address? The actions specified by this AD are intended to correct the potential for insufficient oxygen quantity that is available to occupants of the rear seat bench in some emergency conditions. This condition could result in reduced occupant safety at the rear bench seat location.
- (d) What actions must I accomplish to address this problem? To address this problem, you must accomplish the following:

Actions	Compliance	Procedures
(1) Remove the oxygen tubing and connector assembly, the oxygen tubing support bracket and grommet assembly, and the hose clamp (or FAA-approved equivalent parts) from the rear bench seat (part number 525.22.12.016). Replace these parts with parts of improved design as specified in the service bulletin (or FAA-approved equivalent parts).	Within the next 30 days after May 6, 2002 (the effective date of this AD), unless already accomplished.	Follow the ACCOMPLISHMENT INSTRUCTIONS section of Pilatus Aircraft Ltd. PC-12 Service Bulletin No. 35-002, dated December 19, 2000.

Actions	Compliance	Procedures	
(2) Do not install any of the components referenced in paragraph (d)(1) of this AD unless they are of improved design as specified in the service bulletin (or FAA-approved equivalent parts).		As specified in (Pilatus Aircraft Ltd., PC-12 Service Bulletin No. 35-002, dated December 19, 2000.	

(e) Can I comply with this AD in any other way? You may use an alternative method of compliance or adjust the compliance time if:

(1) Your alternative method of compliance provides an equivalent level of safety; and

(2) The Manager, Standards Office, Small Airplane Directorate, approves your alternative. Submit your request through an FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Standards Office, Small Airplane Directorate.

Note 1: This AD applies to each airplane identified in paragraph (a) of this AD, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if you have not eliminated the unsafe condition, specific actions you propose to address it.

- (f) Where can I get information about any already-approved alternative methods of compliance? Contact Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4059; facsimile: (816) 329–4090.
- (g) What if I need to fly the airplane to another location to comply with this AD? The FAA can issue a special flight permit under sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate your airplane to a location where you can accomplish the requirements of this AD.
- (h) Are any service bulletins incorporated into this AD by reference? Actions required by this AD must be done in accordance with Pilatus Aircraft Ltd. PC–12 Service Bulletin No. 35-002, dated December 19, 2000. The Director of the Federal Register approved this incorporation by reference under 5 U.S.C. 552(a) and 1 CFR part 51. You can get copies from Pilatus Aircraft Ltd., Customer Liaison Manager, CH-6371 Stans, Switzerland, or from Pilatus Business Aircraft Ltd., Product Support Department, 11755 Airport Way, Broomfield, Colorado 80021. You can look at copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 2: The subject of this AD is addressed in Swiss AD HB 2001–001, dated December 28, 2000.

(i) When does this amendment become effective? This amendment becomes effective on May 6, 2002.

Issued in Kansas City, Missouri, on March 14, 2002.

Dorenda D. Baker,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 02–6911 Filed 3–26–02; 8:45 am]

SECURITIES AND EXCHANGE COMMISSION

17 CFR Part 200

[Release Nos. 33–8071; 34–45601 35–27505; 39–2396; IC–25471; and IA–2021]

Approved Information Collections

AGENCY: Securities and Exchange Commission.

ACTION: Final rule.

SUMMARY: The Commission is amending its rules concerning information collection requirements under the Paperwork Reduction Act to update the display of OMB control numbers associated with the Commission's collection of information. The display does not alter any Commission collection of information.

EFFECTIVE DATE: March 27, 2002.

FOR FURTHER INFORMATION CONTACT:

Michael E. Bartell, Associate Executive Director at U.S. Securities and Exchange Commission, 450 5th Street, NW., Washington, DC 20549, 202–942–8800.

SUPPLEMENTARY INFORMATION: The Securities and Exchange Commission ("Commission") is updating the information displayed at 17 CFR 200.800. This amendment sets forth control numbers assigned by the Office of Management and Budget to information collection requirements of the Commission as of the date of this release. The Commission finds that this amendment pertains to agency organization, procedure, or practice and

is therefore not subject to the provisions of the Administrative Procedure Act, 5 U.S.C. 553, requiring advance notice and opportunity for public comment prior to publication. In addition, the Commission finds good cause for immediate effectiveness upon publication in the Federal Register. Delayed effectiveness is unnecessary because this is a technical amendment and will not affect any rights or obligations of members of the public.

List of Subjects in 17 CFR Part 200

Administrative practice and procedure, Freedom of information, Privacy.

Text of Amendment

For the reasons set forth in the Preamble, Title 17, Chapter II of the Code of Federal Regulations is amended as follows:

PART 200—ORGANIZATION; CONDUCT AND ETHICS; AND INFORMATION AND REQUESTS

1. The title to Subpart N and Subpart N are revised to read as follows:

Subpart N—Commission Information Collection Requirements Under the Paperwork Reduction Act: OMB Control Numbers

Authority: 44 U.S.C. 3506; 44 U.S.C. 3507.

§ 200.800 OMB control numbers assigned pursuant to the Paperwork Reduction Act.

(a) Purpose: This subpart collects and displays the control numbers assigned to information collection requirements of the Commission by the Office of Management and Budget pursuant to the Paperwork Reduction Act of 1980, 44 U.S.C. 3500 et seq. This subpart displays current OMB control numbers for those information collection requirements of the Commission that are rules and regulations and codified in 17 CFR either in full text or incorporated by reference with the approval of the Director of the Office of the Federal Register.

(b) Display.

Information collection requirement	17 CFR part or section where identified and described	Current OMB Con- trol No.
Regulation S–X	PART 210	3235-0009