Applicability: Model 757 series airplanes, line numbers 1 through 1018 inclusive, equipped with Rolls Royce engines; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent flammable fluids from leaking into the interior compartment of the nacelle struts where ignition sources exist, which could result in the ignition of flammable fluids and an uncontained fire, accomplish the following:

Inspection

(a) Within 3,000 flight hours after the effective date of this AD: Do a detailed inspection of the support brackets and associated fasteners for the hydraulic lines located in the nacelle struts for loose and/or damaged parts, by accomplishing all of the actions specified in Part 1 of the Accomplishment Instructions of Boeing Alert Service Bulletin 757-54A0045 (for Model 757-200 series airplanes), dated May 22, 2003; or Boeing Alert Service Bulletin 757-54A0046 (for Model 757-300 series airplanes), dated May 29, 2003; as applicable. Do the actions per the applicable service bulletin. Repeat the inspection thereafter at intervals not to exceed 3,000 flight hours.

Note 1: For the purposes of this AD, a detailed inspection is defined as: "An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required."

Related Investigative and Corrective Actions

(b) Except as required by paragraph (d) of this AD: If any loose or damaged parts are found during any inspection required by paragraph (a) of this AD, before further flight, do all of the related and investigative corrective actions specified in Part 2 of the Accomplishment Instructions of Boeing Alert Service Bulletin 757-54A0045 (for Model 757-200 series airplanes), dated May 22, 2003; or Boeing Alert Service Bulletin 757-54A0046 (for Model 757-300 series airplanes), dated May 29, 2003; as applicable. Do the actions per the applicable service bulletin. Accomplishment of these actions constitutes terminating action for the repetitive inspections required by paragraph (a) of this AD.

Optional Terminating Action

(c) Accomplishment of all of the actions specified in Part 2 of the Accomplishment Instructions of Boeing Alert Service Bulletin 757–54A0045 (for Model 757–200 series airplanes), dated May 22, 2003; or Boeing Alert Service Bulletin 757–54A0046 (for Model 757–300 series airplanes), dated May 29, 2003; as applicable; constitutes terminating action for the repetitive inspections required by paragraph (a) of this AD.

Repair Information

(d) If any damage is found during any inspection required by this AD, and the service bulletin specifies contacting Boeing for appropriate action: Before further flight, repair per a method approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA. For a repair method to be approved by the Manager, Seattle ACO, as required by this paragraph, the Manager's approval letter must specifically refer to this AD.

Alternative Methods of Compliance

(e) In accordance with 14 CFR 39.19, the Manager, Seattle ACO, is authorized to approve alternative methods of compliance (AMOCs) for this AD.

Issued in Renton, Washington, on May 5, 2004.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 04–11041 Filed 5–14–04; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-344-AD]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A310 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Airbus Model A310 series airplanes. This proposal would require modification of certain wires in the right-hand wing. This action is necessary to ensure that fuel quantity indication wires are properly separated from wires carrying 115-volt alternating current (AC). Improper separation of such wires, in the event of wire damage, could lead to a short circuit and a possible ignition source, which could result in a fire in the airplane. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by June 16, 2004.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002–NM-344–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056.

Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227–1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 2002–NM–344–AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 or 2000 or ASCII text.

The service information referenced in the proposed rule may be obtained from Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Tim Backman, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2797; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (e.g., reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2002–NM–344–AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002–NM-344–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056.

Discussion

The FAA has examined the underlying safety issues involved in recent fuel tank explosions on several large transport airplanes, including the adequacy of existing regulations, the service history of airplanes subject to those regulations, and existing maintenance practices for fuel tank systems. As a result of those findings, we issued a regulation titled "Transport Airplane Fuel Tank System Design Review, Flammability Reduction and Maintenance and Inspection Requirements' (67 FR 23086, May 7, 2001). In addition to new airworthiness standards for transport airplanes and new maintenance requirements, this rule included Special Federal Aviation Regulation No. 88 ("SFAR 88," Amendment 21-78, and subsequent Amendments 21-82 and 21-83).

Among other actions, SFAR 88 requires certain type design (i.e., type certificate (TC) and supplemental type certificate (STC)) holders to substantiate that their fuel tank systems can prevent ignition sources in the fuel tanks. This requirement applies to type design holders for large turbine-powered transport airplanes and for subsequent modifications to those airplanes. It requires them to perform design reviews and to develop design changes and maintenance procedures if their designs do not meet the new fuel tank safety standards. As explained in the preamble to the rule, we intended to adopt airworthiness directives to mandate any changes found necessary to address unsafe conditions identified as a result of these reviews.

In evaluating these design reviews, we have established four criteria intended to define the unsafe conditions associated with fuel tank systems that require corrective actions. The percentage of operating time during

which fuel tanks are exposed to flammable conditions is one of these criteria. The other three criteria address the failure types under evaluation: Single failures, single failures in combination with another latent condition(s), and in-service failure experience. For all four criteria, the evaluations included consideration of previous actions taken that may mitigate the need for further action.

Based on this process, we have determined that the actions identified in this proposed AD are necessary to reduce the potential of ignition sources inside fuel tanks, which, in combination with flammable fuel vapors, could result in fuel tank explosions and consequent loss of the airplane.

As a result of the design reviews of the fuel tank system, the Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, notified the FAA that an unsafe condition may exist on certain Airbus Model A310 series airplanes. The DGAC advises that analysis of wire routing has revealed that Route 2S of the fuel electrical circuit, located in the righthand wing, does not provide adequate separation of fuel quantity indication wires from wires carrying 115-volt alternating current (AC). Improper separation of such wires, in the event of wire damage, could lead to a short circuit and a possible ignition source, which could result in a fire in the airplane.

Explanation of Relevant Service Information

Airbus has issued Service Bulletin A310-28-2148, Revision 01, dated October 29, 2002. That service bulletin describes procedures for modifying the routing of wires in the right-hand wing by installing cable sleeves. Accomplishment of the actions specified in the service bulletin is intended to adequately address the identified unsafe condition. The DGAC classified this service bulletin as mandatory and issued French airworthiness directive 2002-578(B), dated November 27, 2002, to ensure the continued airworthiness of these airplanes in France.

FAA's Conclusions

This airplane model is manufactured in France and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the DGAC has kept the FAA informed of the situation described above. The

FAA has examined the findings of the DGAC, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, the proposed AD would require accomplishment of the actions specified in the service bulletin described previously.

Cost Impact

The FAA estimates that 46 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 9 work hours per airplane to accomplish the proposed actions, and that the average labor rate is \$65 per work hour. Required parts would cost approximately \$1,880 per airplane. Based on these figures, the cost impact of the proposed AD on U.S. operators is estimated to be \$113,390, or \$2,465 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities

under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Airbus: Docket 2002-NM-344-AD.

Applicability: Model A310 series airplanes on which neither Airbus Modification 12427 nor 12435 has been accomplished, certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To ensure that fuel quantity indication wires are properly separated from wires carrying 115-volt alternating current (AC), accomplish the following:

Modification

(a) Within 4,000 flight hours after the effective date of this AD: Modify the routing of wires in the right-hand wing by installing cable sleeves, per the Accomplishment Instructions of Airbus Service Bulletin A310–28–2148, Revision 01, dated October 29, 2002

Actions Accomplished Previously

(b) Modification of the routing of wires accomplished before the effective date of this AD per Airbus Service Bulletin A310–28–2148, dated January 23, 2002, is acceptable for compliance with the corresponding requirements of paragraph (a) of this AD.

Alternative Methods of Compliance

(c) In accordance with 14 CFR 39.19, the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, is authorized to approve alternative methods of compliance for this AD.

Note 1: The subject of this AD is addressed in French airworthiness directive 2002–578(B), dated November 27, 2002.

Issued in Renton, Washington, on May 5, 2004.

Ali Bahrami.

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 04–11040 Filed 5–14–04; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF LABOR

Occupational Safety and Health Administration

29 CFR Part 1926

[Docket No. S-030]

RIN 1218-AC01

Safety Standards for Cranes and Derricks

AGENCY: Occupational Safety and Health Administration (OSHA), U.S. Department of Labor.

ACTION: Notice of changes in dates and times of June Negotiated Rulemaking Committee meeting.

SUMMARY: The Occupational Safety and Health Administration (OSHA) announces that the Crane and Derrick Negotiated Rulemaking Advisory Committee (C-DAC) has extended the dates and times of the June meeting in Phoenix, AZ. The meeting will be on June 1, 2, 3 and 4, 2004 and held at the Home Builders Association of Central Arizona facility located at 3200 East Camelback Road, Suite 180, Phoenix, AZ 85018. The June meeting will begin at 1 p.m. on June 1st and 8:30 a.m. on June 2, 3, and 4. The meeting is expected to last three and a half days. The Committee will review summary notes of the prior meeting and review draft regulatory text. The meeting will be open to the public. For more details, please see the original June Federal Register notice published at Volume 69 of the Federal Register, page 22748, April 27, 2004.

Signed at Washington, DC, this 11th day of May, 2004.

John L. Henshaw,

Assistant Secretary of Labor for Occupational Safety and Health.

[FR Doc. 04-11099 Filed 5-14-04; 8:45 am]

BILLING CODE 4510-26-P

DEPARTMENT OF HOMELAND SECURITY

Coast Guard

33 CFR Part 117

[CGD05-04-052]

RIN 1625-AA09

Drawbridge Operation Regulation; Spa Creek, Annapolis, MD

AGENCY: Coast Guard, DHS.

ACTION: Notice of proposed rulemaking.

SUMMARY: The Commander, Fifth Coast Guard District, is proposing to change the regulations that govern the operation of the S181 Bridge, mile 4.0, across Spa Creek, at Annapolis, Maryland. These regulations are necessary to facilitate public safety and expedite vehicular traffic from the city of Annapolis after the annual fireworks display. This proposed change to the drawbridge operation schedule will allow the S181 Bridge to remain in the closed position from 8:30 p.m. to 11 p.m. on July 4, of every year. In the event of inclement weather, the alternate date is July 5. **DATES:** Comments and related material must reach the Coast Guard on or before July 16, 2004.

ADDRESSES: You may mail comments and related material to the Commander (oan—b), Fifth Coast Guard District, Federal Building, 4th Floor, 431 Crawford Street, Portsmouth, Virginia 23703–5004. Comments and material received from the public, as well as documents indicated in this preamble as being available in the docket, will become part of this docket and will be available for inspection or copying at the above address between 8 a.m. and 4 p.m., Monday through Friday, except Federal holidays.

FOR FURTHER INFORMATION CONTACT: Gary S. Heyer, Bridge Management Specialist, Fifth Coast Guard District, at (757) 398–6227.

SUPPLEMENTARY INFORMATION:

Request for Comments

We encourage you to participate in this rulemaking by submitting comments and related material. If you do so, please include your name and address, identify the docket number for this rulemaking (CGD05–04–052), indicate the specific section of this document to which each comment applies, and give the reason for each comment. Please submit all comments and related material in an unbound format, no larger than 8½ by 11 inches, suitable for copying. If you would like to know they reached us, please enclose