

TABLE 1.—INCORPORATION BY REFERENCE

Service bulletin	Revision	Date
Dornier Alert Service Bulletin ASB-328-71-006	1	February 16, 1995.
Dornier Service Bulletin SB-328-30-020	Original	March 17, 1994.
Dornier Service Bulletin SB-328-30-432	Original	April 26, 2002.
Dornier Service Bulletin SB-328-71-122	1	May 10, 1999.
Dornier Service Bulletin SB-328-71-125	3	May 10, 1999.

(1) The incorporation by reference of the service bulletins listed in Table 2 of this AD is approved by the Director of the Federal

Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51:

TABLE 2.—NEW SERVICE BULLETINS FOR INCORPORATION BY REFERENCE

Service bulletin	Revision	Date
Dornier Service Bulletin SB-328-30-432	Original	April 26, 2002.
Dornier Service Bulletin SB-328-71-122	1	May 10, 1999.
Dornier Service Bulletin SB-328-71-125	3	May 10, 1999.

(2) The incorporation by reference of the service bulletins listed in Table 3 of this AD was approved previously by the Director of

the Federal Register as of April 6, 1995 (60 FR 15037, March 22, 1995):

TABLE 3.—SERVICE BULLETINS PREVIOUSLY INCORPORATED BY REFERENCE

Service bulletin	Revision	Date
Dornier Alert Service Bulletin ASB-328-71-006	1	February 16, 1995.
Dornier Service Bulletin SB-328-30-020	Original	March 17, 1994.

(3) Copies may be obtained from AvCraft Aerospace GmbH, P.O. Box 1103, D-82230 Wessling, Germany. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Note 4: The subject of this AD is addressed in German airworthiness directives 1995-156/3, dated July 1, 1999; and 2002-256, dated September 5, 2002.

Effective Date

(m) This amendment becomes effective on November 17, 2004.

Issued in Renton, Washington, on September 30, 2004.

Kevin M. Mullin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-22562 Filed 10-12-04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-286-AD; Amendment 39-13821; AD 2004-20-16]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 747-200B, -200C, -200F, -300, -400, -400D, and -400F Series Airplanes; and Model 747SP Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Boeing Model 747-200B, -200C, -200F, -300, -400, -400D, and -400F series airplanes; and Model 747SP series airplanes, that requires repetitive functional tests of the auxiliary power unit (APU) and engine fire shutoff switches and repetitive replacements of the APU and engine fire shutoff switches. The AD also provides an optional terminating action for the repetitive functional tests and replacements. This action is necessary to prevent mineral build-up on the APU

and engine fire shutoff switches, which could lead to failure of the switches to discharge fire suppressant in the affected area and could result in an uncontrolled fire that could spread to the strut, wing, or aft body of the airplane. This action is intended to address the identified unsafe condition.

DATES: Effective November 17, 2004.

The incorporation by reference of a certain publication listed in the regulations is approved by the Director of the Federal Register as of November 17, 2004.

ADDRESSES: The service information referenced in this AD may be obtained from Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

FOR FURTHER INFORMATION CONTACT: Sulmo Mariano, Aerospace Engineer,

Propulsion Branch, ANM-140S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 917-6501; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Boeing Model 747-200B, -200C, -200F, -300, -400, -400D, and -400F series airplanes; and Model 747SP series airplanes, was published in the **Federal Register** on June 23, 2004 (69 FR 34971). That action proposed to require repetitive functional tests of the auxiliary power unit (APU) and engine

fire shutoff switches and repetitive replacements of the APU and engine fire shutoff switches. That action also proposed to provide an optional terminating action for the repetitive functional tests and replacements.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

TABLE 1.—ESTIMATED COSTS

Action	Work hours	Cost per airplane	Total cost
Inspection and Functional Test (per test cycle)	10-14 (depending on airplane model)	\$650-910	\$32,500-45,500

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy

of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. Section 39.13 is amended by adding the following new airworthiness directive:

2004-20-16 Boeing: Amendment 39-13821. Docket 2002-NM-286-AD.

Applicability: Model 747-200B, -200C, -200F, -300, -400, -400D, and -400F series airplanes; and Model 747SP series airplanes; as listed in Boeing Alert Service Bulletin 747-26A2274, Revision 1, dated January 9, 2003; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent mineral build-up on the auxiliary power unit (APU) and engine fire shutoff switches, which could lead to failure of the switches to discharge fire suppressant in the affected area and could result in an uncontrolled fire that could spread to the

Clarification of Summary Language

Since the proposed AD was published we noticed that in the Summary of the proposed AD we referred to "inspections" instead of "functional tests." We have corrected the Summary of this AD.

Cost Impact

There are approximately 316 airplanes of the affected design in the worldwide fleet. We estimate that 50 airplanes of U.S. registry will be affected by this AD, and that the average labor rate is \$65 per work hour. Table 1 provides the estimated costs for U.S. operators to comply with this AD.

strut, wing, or aft body of the airplane, accomplish the following:

Service Bulletin Reference

(a) The term "service bulletin," as used in this AD, means the Accomplishment Instructions of Boeing Alert Service Bulletin 747-26A2274, Revision 1, dated January 9, 2003.

Initial and Repetitive Functional Test

(b) At the later of the compliance times specified in paragraphs (b)(1) and (b)(2) of this AD, perform a functional test of the APU and engine fire shutoff switches, in accordance with the service bulletin. Repeat the functional test thereafter at intervals not to exceed 18 months.

(1) Within 18 months since the date of issuance of the original Airworthiness Certificate or the original Export Certificate of Airworthiness.

(2) Within 90 days after the effective date of this AD.

Fire Shutoff Switch Failure

(c) If any fire shutoff switch fails during any functional test required by paragraph (b) or (f) of this AD, before further flight, replace the switch with a new or serviceable switch, in accordance with the service bulletin. Repeat the switch replacement thereafter at intervals not to exceed 36 months.

Replacement

(d) Within 36 months after the effective date of this AD, replace all APU and engine fire shutoff switches that have not been previously replaced per paragraph (c) of this AD with new or serviceable switches, in accordance with the service bulletin. Repeat the switch replacement thereafter at intervals not to exceed 36 months.

Deactivation of Lucas Humidifier

(e) Operators may terminate the repetitive requirements of paragraphs (b), (c), and (d) of this AD by accomplishing the actions in

paragraphs (e)(1) and (e)(2) of this AD, except as provided by paragraph (f) of this AD.

(1) Deactivate the Lucas humidifier, part number (P/N) M01AA0101, M01AB0101, M01AB0102, or M01AB0103, in accordance with the service bulletin.

(2) Before further flight following the deactivation specified in paragraph (e)(1) of this AD, replace all APU and engine fire shutoff switches with new or serviceable switches in accordance with the service bulletin.

Reactivation of Lucas Humidifier

(f) For any airplanes on which Lucas humidifier, P/N M01AA0101, M01AB0101, M01AB0102, or M01AB0103 is reactivated after the effective date of this AD: Do the requirements of paragraphs (f)(1) and (f)(2) of this AD at the times specified in those paragraphs.

(1) Within 18 months after reactivating the humidifier, and thereafter at intervals not to exceed 18 months, do the functional test required by paragraph (b) of this AD.

(2) Within 36 months after reactivating the humidifier, and thereafter at intervals not to exceed 36 months, replace all APU and engine fire shutoff switches that have not been previously replaced per paragraph (c) of this AD. Do the replacements per paragraph (d) of this AD.

Actions Accomplished Per Previous Issue of Service Bulletin

(g) Unless otherwise specified in this AD, actions accomplished before the effective date of this AD per Boeing Alert Service Bulletin 747-26A2274, dated August 29, 2002, are considered acceptable for compliance with the corresponding action specified in this AD.

Alternative Methods of Compliance

(h) In accordance with 14 CFR 39.19, the Manager, Seattle Aircraft Certification Office, FAA, is authorized to approve alternative methods of compliance for this AD.

Incorporation by Reference

(i) The actions shall be done in accordance with Boeing Alert Service Bulletin 747-26A2274, Revision 1, dated January 9, 2003. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Effective Date

(j) This amendment becomes effective on November 17, 2004.

Issued in Renton, Washington, on September 30, 2004.

Kevin M. Mullin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-22563 Filed 10-12-04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-211-AD; Amendment 39-13819; AD 2004-20-14]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A300 B4 Series Airplanes and Model A300 B4-600, A300 B4-600R, and A300 F4-600R (Collectively Called A300-600) Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD), applicable to all Airbus Model A300 B4 series airplanes and all Airbus Model A300-600 series airplanes. That AD currently requires a one-time high frequency eddy current inspection to detect cracking of the splice fitting at fuselage frame (FR) 47 between stringers 24 and 25; and corrective actions if necessary. This amendment requires new repetitive inspections of an expanded area and adds airplanes to the applicability in the existing AD. The actions specified by this AD are intended to detect and correct cracking of the splice fitting at fuselage FR 47, which could result in reduced structural integrity of the airplane. This action is intended to address the identified unsafe condition.

DATES: Effective November 17, 2004.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of November 17, 2004.

ADDRESSES: The service information referenced in this AD may be obtained from Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

FOR FURTHER INFORMATION CONTACT: Tim Backman, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2797; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) by superseding AD 2001-03-14, amendment 39-12118 (66 FR 10957, February 21, 2001), was published as a supplemental notice of proposed rulemaking (NPRM) in the **Federal Register** on August 4, 2004 (69 FR 47035). The proposal is applicable to all Airbus Model A300 series airplanes and all Airbus Model A300-600 series airplanes. The action proposed to require new repetitive high frequency eddy current inspections to detect cracking of an expanded area, and corrective actions, if necessary; and to add airplanes to the applicability in the existing AD.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Clarification of Service Information Requirements for Paragraph (b) of This AD

In our response to comments in the Preamble of the supplemental NPRM, we stated our intent to revise paragraphs (a), (b), and (c) of the supplemental NPRM to refer to Revision 02 of the referenced Airbus service bulletins as the appropriate sources of service information for accomplishment of the required actions. (Revision 01 of those service bulletins was referenced in the original NPRM for accomplishment of the required actions.) However, while we revised paragraphs (a) and (c) of the supplemental NPRM, we inadvertently omitted the revision to paragraph (b). Therefore, we have revised paragraph (b) of this final rule to reference Revision 02 of Airbus Service Bulletin A300-53-6123 as the appropriate source of service information for the required actions in that paragraph.

Conclusion

After careful review of the available data, the FAA has determined that air safety and the public interest require the adoption of the rule with the change