

Issued in Renton, Washington, on April 18, 2008.

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## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2007-0046; Directorate  
Identifier 2007-NM-173-AD; Amendment  
39-15496; AD 2008-09-15]

RIN 2120-AA64

#### Airworthiness Directives; Boeing Model 737-100, -200, -200C, -300, -400, and -500 Series Airplanes

AGENCY: Federal Aviation  
Administration (FAA), DOT.

ACTION: Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for all Boeing Model 737-100, -200, -200C, -300, -400, and -500 series airplanes. This AD requires repetitive inspections for any cracking of or damage to the left side and right side flight deck No. 2, No. 4, and No. 5 windows and corrective actions if necessary. This AD results from reports of in-flight departure and separation of the flight deck windows. We are issuing this AD to detect and correct cracking in the vinyl interlayer or damage to the structural inner glass panes of the flight deck No. 2, No. 4, and No. 5 windows, which could result in loss of a window and rapid loss of cabin pressure. Loss of cabin pressure could cause crew communication difficulties or crew incapacitation.

**DATES:** This AD is effective June 6, 2008.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of June 6, 2008.

**ADDRESSES:** For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527)

is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Wayne Lockett, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6447; fax (425) 917-6590.

#### SUPPLEMENTARY INFORMATION:

##### Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to all Boeing Model 737-100, -200, -200C, -300, -400, and -500 series airplanes. That NPRM was published in the **Federal Register** on October 17, 2007 (72 FR 58766). That NPRM proposed to require repetitive inspections for any cracking of or damage to the left side and right side flight deck No. 2, No. 4, and No. 5 windows and corrective actions if necessary.

##### Changes Made to This AD

We have deleted paragraph (h)(4) of the NPRM and added a new paragraph (h) to this AD specifying that installation of metallic window blanks at cockpit eyebrow windows No. 4 and No. 5 in accordance with Supplemental Type Certificate (STC) ST01630SE terminates the initial and repetitive inspections for the flight deck No. 4 and No. 5 windows required by paragraph (f) of this AD. Incorporation of STC ST01630SE is considered a terminating action, not an alternative method of compliance (AMOC), since an AMOC can only be issued after an AD has been issued. We have also reidentified the AMOC paragraph of the NPRM as paragraph (i) in this AD.

##### Comments

We gave the public the opportunity to participate in developing this AD. We considered the comments received from the three commenters.

##### Support for the NPRM

Boeing and Continental Airlines support the NPRM.

##### Request to Expand Applicability

North Star Aerospace states that the affected window panels are also installed on Boeing Model 707 and 727 airplanes and Model 737-600, -700, -800, and -900 series airplanes, and that it has witnessed failure of the windows on these airplanes. North Star

Aerospace believes the inspections should be extended to include all airplanes equipped with window panels having part numbers (P/Ns) 5-89355-( ), 5-89357-( ), and 5-89358-( ).

We infer the commenter requests that we revise the applicability to add Model 707 and 727 airplanes and Model 737-600, -700, -800, and -900 series airplanes. Since the affected windows are interchangeable, we agree that the windows might be installed on all Model 707, 727, and 737 airplanes. However, we do not agree to expand the applicability of this AD, since we have issued separate rulemaking actions to address the unsafe condition on Model 707 and 727 airplanes and Model 737-600, -700, -800, and -900 series airplanes. Please refer to Docket Nos. FAA-2007-0264, FAA-2007-0265, and FAA-2007-0263, respectively, at <http://www.regulations.gov>. No change to this AD is necessary in this regard.

##### Request To Revise the Compliance Time

Continental Airlines requests that we revise the compliance time for the initial inspection of the flight deck No. 2 window to within 36 months or 7,500 flight hours, whichever occurs first, after the window installation. Continental Airlines states that the NPRM, which proposes to require the initial inspection within 24 months after the effective date of this AD regardless of the age or flight time of the window, unnecessarily penalizes operators who proactively inspect and replace the No. 2 window before the AD is issued. Continental Airlines also states that, according to the wording in the NPRM, a window replaced one day before the effective date of the AD would need to be re-inspected within 24 months, but a window inspected and replaced one day after the effective date of the AD would not need to be re-inspected until 36 months or 7,500 window flight hours, whichever is first.

We do not agree to revise the compliance time for the initial inspection of the flight deck No. 2 window. According to paragraph (e) of this AD, an operator is responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done. If the initial inspection of the No. 2 window was done before the effective date of this AD in accordance with Boeing Alert Service Bulletin 737-56A1023, dated May 24, 2007, then the initial inspection does not need to be accomplished again; only the repetitive inspections would need to be accomplished in accordance with the service bulletin at the applicable

interval specified in the service bulletin. If the initial and repetitive inspections of the No. 2 window are done before the effective date of this AD, but are not done in accordance with the service bulletin, then those inspections are not acceptable for compliance with this AD unless an AMOC is issued for those prior inspections. Under the provisions of paragraph (i) of this AD, we will consider requests for approval of an AMOC if sufficient data are submitted to substantiate that prior inspections incorporate similar criteria to what is provided for in the service bulletin. Therefore, no change to this AD is necessary in this regard.

#### **Request for an AMOC for a Parts Manufacturer Approval (PMA) Equivalent Part**

Continental Airlines states that the FAA has approved a new, improved flight deck No. 2 window designed by GKN Aerospace Transparency Systems, under PMA Holder No. PQ1250NM, Supplement 10, dated September 17, 2007. Continental Airlines states that the new, improved No. 2 window was designed to prevent the premature failure of the window, and that new, improved window addresses the unsafe condition of the NPRM. Continental Airlines, therefore, requests that we add a new AMOC paragraph to this AD, which would exempt the new, improved No. 2 window from the required inspections.

We do not agree to allow the PMA equivalent No. 2 window as an AMOC to the required inspections. Although the window has been approved as a PMA equivalent part, the commenter has not provided data showing that the PMA equivalent window is not susceptible to the same vinyl interlayer cracking. However, under the provisions of paragraph (i) of this AD, we will consider requests for approval of an AMOC if sufficient data are submitted to substantiate that the design change would provide an acceptable level of safety. No change to this AD is necessary in this regard.

#### **Conclusion**

We reviewed the relevant data, considered the comments received, and determined that air safety and the public interest require adopting the AD with the changes described previously. We also determined that these changes will not increase the economic burden on any operator or increase the scope of the AD.

#### **Costs of Compliance**

There are about 2,685 airplanes of the affected design in the worldwide fleet.

This AD affects about 799 airplanes of U.S. registry. The required inspections take about 2 work hours per airplane, at an average labor rate of \$80 per work hour. Based on these figures, the estimated cost of the AD for U.S. operators is \$127,840, or \$160 per airplane, per inspection cycle.

#### **Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### **Regulatory Findings**

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979), and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

You can find our regulatory evaluation and the estimated costs of compliance in the AD Docket.

#### **List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### **Adoption of the Amendment**

- Accordingly, under the authority delegated to me by the Administrator,

the FAA amends 14 CFR part 39 as follows:

#### **PART 39—AIRWORTHINESS DIRECTIVES**

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### **§ 39.13 [Amended]**

- 2. The FAA amends § 39.13 by adding the following new AD:

**2008–09–15 Boeing:** Amendment 39–15496.  
Docket No. FAA–2007–0046; Directorate Identifier 2007–NM–173–AD.

#### **Effective Date**

- (a) This airworthiness directive (AD) is effective June 6, 2008.

#### **Affected ADs**

- (b) None.

#### **Applicability**

- (c) This AD applies to all Boeing Model 737–100, –200, –200C, –300, –400, and –500 series airplanes, certificated in any category.

#### **Unsafe Condition**

- (d) This AD results from reports of in-flight departure and separation of flight deck windows. We are issuing this AD to detect and correct cracking in the vinyl interlayer or damage to the structural inner glass panes of the flight deck No. 2, No. 4, and No. 5 windows, which could result in loss of a window and rapid loss of cabin pressure. Loss of cabin pressure could cause crew communication difficulties or crew incapacitation.

#### **Compliance**

- (e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

#### **Repetitive Inspections and Replacement**

- (f) At the applicable times specified in Tables 1, 2, and 3 of paragraph 1.E. of Boeing Alert Service Bulletin 737–56A1023, dated May 24, 2007, except as provided by paragraph (g) of this AD: Do the internal and external detailed inspections for any cracking of or damage to the left side and right side flight deck No. 2, No. 4, and No. 5 windows and do the applicable corrective actions before further flight, by accomplishing all of the applicable actions specified in the Accomplishment Instructions of the service bulletin. Repeat the inspections thereafter at the applicable interval specified in paragraph 1.E. of the service bulletin.

#### **Exception to Compliance Times**

- (g) Where Tables 1, 2, and 3 of paragraph 1.E. of Boeing Alert Service Bulletin 737–56A1023, dated May 24, 2007, specify counting the compliance time from “\* \* \* the date on this service bulletin,” this AD requires counting the compliance time from the effective date of this AD.

### Optional Terminating Action

(h) Installation of metallic window blanks at cockpit eyebrow windows No. 4 and No. 5 in accordance with Supplemental Type Certificate ST01630SE terminates the initial and repetitive inspections for the flight deck No. 4 and No. 5 windows required by paragraph (f) of this AD. All other applicable actions required by paragraph (f) of this AD must be fully complied with.

### Alternative Methods of Compliance (AMOCs)

(i)(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD, if it is approved by an Authorized Representative for the Boeing Commercial Airplanes Delegation Option Authorization Organization who has been authorized by the Manager, Seattle ACO, to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

### Material Incorporated by Reference

(j) You must use Boeing Alert Service Bulletin 737-56A1023, dated May 24, 2007, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207.

(3) You may review copies of the service information incorporated by reference at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Renton, Washington, on April 18, 2008.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate,  
Aircraft Certification Service.*

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## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2007-28664; Directorate Identifier 2007-NM-007-AD; Amendment 39-15492; AD 2008-09-11]

RIN 2120-AA64

### Airworthiness Directives; Boeing Model 777-200, -300, and -300ER Series Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for all Boeing Model 777-200, -300, and -300ER series airplanes. This AD requires a one-time inspection to determine the material of the forward and aft gray water drain masts. For airplanes having composite gray water drain masts, this AD also requires installation of a bonding jumper between a ground and the clamp on the tube of the forward and aft gray water composite drain masts. This AD results from a report of charred insulation blankets and burned wires around the forward gray water composite drain mast found during an inspection of the forward cargo compartment on a Model 767-300F airplane. We are issuing this AD to prevent a fire near a composite drain mast and possible disruption of the electrical power system due to a lightning strike on a composite drain mast, which could result in the loss of several functions essential for safe flight.

**DATES:** This AD is effective June 6, 2008.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of June 6, 2008.

**ADDRESSES:** For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207.

### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West

Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

### FOR FURTHER INFORMATION CONTACT:

Nicholas Wilson, Aerospace Engineer, Cabin Safety and Environmental Systems Branch, ANM-150S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6476; fax (425) 917-6590.

### SUPPLEMENTARY INFORMATION:

#### Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to all Boeing Model 777-200, -200LR, -300, and -300ER series airplanes. That NPRM was published in the **Federal Register** on July 10, 2007 (72 FR 37475). That NPRM proposed to require a one-time inspection to determine the material of the forward and aft gray water drain masts. For airplanes having composite gray water drain masts, that NPRM also proposed to require installation of a bonding jumper between a ground and the clamp on the tube of the forward and aft gray water composite drain masts.

#### Comments

We gave the public the opportunity to participate in developing this AD. We have considered the single comment received.

### Request To Remove Airplanes From the Applicability Statement of the Proposed AD

Boeing requests that we revise the applicability statement of the NPRM to remove certain airplanes. Boeing states that Model 777 airplanes beginning with line number 525 have a ground bracket and copper bonding jumper installed in production for the aft composite gray water drain mast, and an aluminum drain mast installed in the forward drain mast position. Additionally, Boeing points out that all Model 777-200LR series airplanes produced prior to line number 525 have a bonding jumper installed on the aft composite gray water drain mast and an aluminum forward gray water drain mast. Therefore, Boeing asserts that these airplanes should not be subject to this AD.

We partially agree. For the reasons Boeing stated, we have determined that these airplanes should not be subject to this AD. However, we do not agree to revise the applicability statement of this AD as suggested by Boeing. Instead, we have revised the applicability statement of this final rule to state, "This AD