

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**2013–22–15 Sikorsky Aircraft Corporation:** Amendment 39–17647; FAA–2013–0514; Directorate Identifier 2012–SW–068–AD.

#### (a) Applicability

This AD applies to Model S–76A, S–76B, and S–76C helicopters with a serial number up to and including 760822 and with a spindle cuff assembly, part number (P/N) 76102–08001–043, –045 or –046, or a blade fold cuff assembly, P/N 76150–09601–041, installed, certificated in any category.

#### (b) Unsafe Condition

This AD defines the unsafe condition as a crack in a spindle cuff assembly or blade fold cuff assembly. This condition could result in failure of a spindle cuff assembly or blade fold cuff assembly, loss of a rotor blade, and subsequent loss of control of the helicopter.

#### (c) Effective Date

This AD becomes effective December 13, 2013.

#### (d) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

#### (e) Required Actions

Within 150 hours time-in-service (TIS):

(1) For each spindle cuff assembly or blade cuff assembly with 1,900 or more hours TIS, conduct a nondestructive inspection (NDI) by following the Accomplishment Instructions, paragraph 3.B., of Sikorsky S–76 Alert Service Bulletin ASB 76–65–67A, Revision A, dated July 18, 2012 (ASB), except this AD does not require you to contact Sikorsky Aircraft Corporation. This inspection must be done by a level 2 or higher technician with National Aerospace Standard 410 or equivalent certification.

(2) For each spindle cuff assembly or blade cuff assembly with less than 1,900 hours TIS, visually inspect the area indicated in Figure 4 of the ASB as “white paint application area” for a crack by using a 5x or higher power magnifying glass.

(3) If there is a crack, before further flight, replace the cracked part.

(4) If there is no crack, apply white paint by following the Accomplishment Instructions, paragraph 3.D., of the ASB.

(5) Do not install an affected spindle cuff assembly or blade fold cuff assembly on any helicopter unless it has been inspected in accordance with paragraphs (e)(1) through (e)(4) of this AD.

#### (f) Special Flight Permit

Special flight permits will not be issued.

#### (g) Alternative Methods of Compliance (AMOC)

(1) The Manager, Boston Aircraft Certification Office, FAA, may approve AMOCs for this AD. Send your proposal to:

Nicholas Faust, Aviation Safety Engineer, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, Massachusetts 01803; telephone (781) 238–7763; email [nicholas.faust@faa.gov](mailto:nicholas.faust@faa.gov).

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office before operating any aircraft complying with this AD through an AMOC.

#### (h) Subject

Joint Aircraft Service Component (JASC) Code: 6220 Main Rotor Head.

#### (i) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Sikorsky S–76 Alert Service Bulletin ASB 76–65–67A, Revision A, dated July 18, 2012.

(ii) Reserved.

(3) For service information identified in this AD, contact Sikorsky Aircraft Corporation, Attn: Manager, Commercial Technical Support, mailstop s581a, 6900 Main Street, Stratford, CT 06614; telephone (800) 562–4409; email [tsslibrary@sikorsky.com](mailto:tsslibrary@sikorsky.com); or at <http://www.sikorsky.com>.

(4) You may view this service information at FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741–6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Fort Worth, Texas, on October 24, 2013.

**Kim Smith,**

*Directorate Manager, Rotorcraft Directorate, Aircraft Certification Service.*

[FR Doc. 2013–26043 Filed 11–7–13; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA–2013–0928; Directorate Identifier 2013–CE–036–AD; Amendment 39–17645; AD 2013–22–13]

RIN 2120–AA64

#### Airworthiness Directives; PILATUS Aircraft Ltd. Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule; request for comments.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for certain PILATUS Aircraft Ltd. Model PC–7 airplanes. This AD results from mandatory continuing airworthiness information (MCAI) issued by the aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as chafing on the wiring harness attached to the engine mounting frame on the right-hand side of the engine compartment, which could cause a short circuit and could result in a fire in the engine compartment. We are issuing this AD to require actions to address the unsafe condition on these products.

**DATES:** This AD is effective November 29, 2013.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of November 29, 2013.

We must receive comments on this AD by December 23, 2013.

**ADDRESSES:** You may send comments by any of the following methods:

- **Federal eRulemaking Portal:** Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- **Fax:** (202) 493–2251.

- **Mail:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- **Hand Delivery:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this AD, contact PILATUS AIRCRAFT LTD., Customer Technical Support (MCC), P.O. Box 992, CH–6371 STANS, Switzerland; telephone: +41 (0)41 619 67 74; fax: +41 (0)41 619 67 73; Internet:

<http://www.pilatus-aircraft.com> or email: [Techsupport@pilatus-aircraft.com](mailto:Techsupport@pilatus-aircraft.com). You may view this referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

#### FOR FURTHER INFORMATION CONTACT:

Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4059; fax: (816) 329-4090; email: [doug.rudolph@faa.gov](mailto:doug.rudolph@faa.gov).

#### SUPPLEMENTARY INFORMATION:

##### Discussion

The Federal Office of Civil Aviation (FOCA), which is the aviation authority for Switzerland, has issued AD HB-2013-009, dated October 7, 2013 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

This Airworthiness Directive (AD) is prompted due to a report of chafing of the wiring harness attached to the engine mounting frame on the RH side of the engine compartment. Due to the limited space available in this area the wiring harness can chafe against the RH flexible duct for the condenser.

Such a condition, if left uncorrected, could lead to a short circuit which could cause a fire in the engine compartment.

In order to correct and control the situation, this AD requires a one-time inspection of the wiring harness and flexible duct of the condenser for chafing. If major damage is found, the damaged parts must be replaced. If minor damage is found, this AD requires the installation of a protective sleeve on the wiring harness.

You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA-2013-0928.

#### Relevant Service Information

PILATUS Aircraft Ltd. has issued PILATUS PC-7 Service Bulletin No. 24-009, dated September 6, 2013. The

actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

#### FAA's Determination and Requirements of This AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with this State of Design Authority, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are issuing this AD because we evaluated all information provided by the State of Design Authority and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

#### FAA's Determination of the Effective Date

An unsafe condition exists that requires the immediate adoption of this AD. The FAA has found that the risk to the flying public justifies waiving notice and comment prior to adoption of this rule because chafing of the wiring harness in the engine compartment could cause a short circuit and could result in a fire in the engine compartment. Therefore, we determined that notice and opportunity for public comment before issuing this AD are impracticable and that good cause exists for making this amendment effective in fewer than 30 days.

#### Comments Invited

This AD is a final rule that involves requirements affecting flight safety, and we did not precede it by notice and opportunity for public comment. We invite you to send any written relevant data, views, or arguments about this AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2013-0928; Directorate Identifier 2013-CE-036-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this AD. We will consider all comments received by the closing date and may amend this AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this AD.

#### Costs of Compliance

We estimate that this AD will affect 10 products of U.S. registry. We also estimate that it will take about 1 work-hour per product to comply with the basic requirements of this AD. The average labor rate is \$85 per work-hour.

Based on these figures, we estimate the cost of this AD on U.S. operators to be \$850, or \$85 per product.

In addition, we estimate that necessary follow-on actions will take about 5 work-hours to install a protective sleeve on the wiring harness, if minor damage is found, and require parts costing approximately \$100, for a cost of \$525 per product. We also estimate that it will take about 12 work-hours to replace damaged parts, if major damage is found, and require parts costing approximately \$500, for a cost of \$1,520 per product. We have no way of determining the number of products that may need these actions.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

**2013–22–13 PILATUS Aircraft Ltd.:**  
Amendment 39–17645; Docket No.  
FAA–2013–0928; Directorate Identifier  
2013–CE–036–AD.

#### (a) Effective Date

This airworthiness directive (AD) becomes effective November 29, 2013.

#### (b) Affected ADs

None.

#### (c) Applicability

This AD applies to PILATUS Aircraft Ltd. Model PC–7 airplanes, manufacturers' serial numbers (MSN) 101 through 618, certificated in any category.

#### (d) Subject

Air Transport Association of America (ATA) Code 24: Electrical Power.

#### (e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as chafing on the wiring harness attached to the engine mounting frame on the right-hand side of the engine compartment. We are issuing this AD to prevent a short circuit, which could result in fire in the engine compartment.

#### (f) Actions and Compliance

Unless already done, do the actions specified in paragraphs (f)(1) and (f)(2) of this AD.

(1) Within the next 90 days after November 29, 2013 (the effective date of this AD), visually inspect the wiring harness and the flexile duct in the engine compartment for signs of chafing following the Accomplishment Instructions in PILATUS

Aircraft Ltd. PC–7 Service Bulletin No. 24–009, dated September 6, 2013.

(2) If, during the inspection required in paragraph (f)(1) of this AD, any signs of chafing are found, before further flight, take all necessary corrective actions following the Accomplishment Instructions in PILATUS Aircraft Ltd. PC–7 Service Bulletin No. 24–009, dated September 6, 2013.

#### (g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4059; fax: (816) 329–4090; email: [doug.rudolph@faa.gov](mailto:doug.rudolph@faa.gov). Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, a federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to a penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB Control Number. The OMB Control Number for this information collection is 2120–0056. Public reporting for this collection of information is estimated to be approximately 5 minutes per response, including the time for reviewing instructions, completing and reviewing the collection of information. All responses to this collection of information are mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at: 800 Independence Ave. SW., Washington, DC 20591, Attn: Information Collection Clearance Officer, AES–200.

#### (h) Related Information

Refer to MCAI Federal Office of Civil Aviation (FOCA) AD HB–2013–009, dated October 7, 2013, for related information. You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA–2013–0928.

#### (i) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) PILATUS Aircraft Ltd. PC–7 Service Bulletin No. 24–009, dated September 6, 2013.

(ii) Reserved.

(3) For PILATUS Aircraft Ltd. service information identified in this AD, contact PILATUS AIRCRAFT LTD., Customer Technical Support (MCC), P.O. Box 992, CH–6371 STANS, Switzerland; telephone: +41 (0)41 619 67 74; fax: +41 (0)41 619 67 73; Internet: <http://www.pilatus-aircraft.com> or email: [Techsupport@pilatus-aircraft.com](mailto:Techsupport@pilatus-aircraft.com).

(4) You may view this service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Kansas City, Missouri on October 24, 2013.

**Earl Lawrence,**

*Manager, Small Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2013–25953 Filed 11–7–13; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA–2013–0927; Directorate Identifier 2013–CE–030–AD; Amendment 39–17644; AD 2013–22–12]

**RIN 2120–AA64**

#### Airworthiness Directives; DG Flugzeugbau GmbH Gliders

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule; request for comments.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for all DG Flugzeugbau GmbH Models DG–800A, DG–800B, DG–500MB gliders. This AD results from mandatory continuing airworthiness information (MCAI) issued by the aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as a defective starter motor control unit, which could activate the starter motor without pressing the starter button. We are issuing this AD to require actions to address the unsafe condition on these products.