Proposed Rules

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This section of the FEDERAL REGISTER contains notices to the public of the proposed issuance of rules and regulations. The purpose of these notices is to give interested persons an opportunity to participate in the rule making prior to the adoption of the final rules.

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2016-9074; Directorate Identifier 2016-NM-097-AD]

RIN 2120-AA64

Airworthiness Directives; Airbus Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking

(NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for all Airbus Model A318-111 and -112 airplanes, Model A319-111, -112, -113, -114, and -115 airplanes, Model A320-211, -212, and -214 airplanes, and Model A321–111, –112, –211, –212, and -213 airplanes. This proposed AD was prompted by reports of engine fan cowl door (FCD) losses on airplanes equipped with CFM56 engines due to operator failure to close the FCD during ground operations. This proposed AD would require modification and reidentification of certain FCDs or replacement of certain FCDs. This proposed AD would also require installation of a placard. We are proposing this AD to prevent in-flight loss of an engine FCD and possible consequent damage to the airplane. **DATES:** We must receive comments on this proposed AD by November 10,

this proposed AD by November 10, 2016.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.
 - Fax: 202-493-2251.
- Mail: U.S. Department of Transportation, Docket Operations, M— 30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.

• Hand Delivery: Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact Airbus, Airworthiness Office—EIAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet http://www.airbus.com. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425 227–1221.

Examining the AD Docket

You may examine the AD docket on the Internet at http:// www.regulations.gov by searching for and locating Docket No. FAA-2016-9074; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Sanjay Ralhan, Aerospace Engineer,

Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone 425–227–1405; fax 425–227–1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the ADDRESSES section. Include "Docket No. FAA-2016-9074; Directorate Identifier 2016-NM-097-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to http://

www.regulations.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA Airworthiness Directive, 2016–0069, dated April 11, 2016 (referred to after this as the Mandatory Continuing Airworthiness Information, or "the MCAI"), to correct an unsafe condition for all Airbus Model A318–111 and –112 airplanes, Model A319–111, –112, –113, –114, and –115 airplanes, Model A320–211, –212, and –214 airplanes, and Model A321–111, –112, –211, –212, and –213 airplanes. The MCAI states:

Fan Cowl Door (FCD) losses were reported on aeroplanes equipped with CFM56 engines.

Investigations confirmed that in all cases the fan cowls were opened prior to the flight and were not correctly re-secured. During the pre-flight inspection, it was then not detected that the FCD were not properly latched.

This condition, if not detected and corrected, could lead to in-flight loss of a FCD, possibly resulting in damage to the aeroplane and/or injury to persons on the ground.

Prompted by these events, new FCD front latch and keeper assembly were developed, having a specific key necessary to un-latch the FCD. This key cannot be removed unless the FCD front latch is safely closed. The key, after removal, must be stowed in the flight deck at a specific location, as instructed in the applicable Aircraft Maintenance Manual. Applicable Flight Crew Operating Manual has been amended accordingly. After modification, the FCD is identified with a different Part Number (P/N).

For the reasons described above, this [EASA] AD requires modification and reidentification of FCD [or replacement of the FCD].

You may examine the MCAI in the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–9074.

Related Service Information Under 1 CFR Part 51

We reviewed Airbus Service Bulletin A320–71–1068, Revision 01, dated April 28, 2016. This service information describes procedures for modifying the left-hand and right-hand FCDs on engines 1 and 2; installing a placard;

and re-identifying both the left-hand and right-hand FCDs with the new part number. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

FAA's Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of these same type designs.

Differences Between This Proposed AD and the MCAI or Service Information

The parts installation prohibition specified in paragraph (5) of the MCAI

depends on the configuration of the airplane. However, paragraph (k) of this proposed AD prohibits installing certain parts for all airplanes as of the effective date of this proposed AD.

Costs of Compliance

We estimate that this proposed AD affects 400 airplanes of U.S. registry.

We estimate the following costs to comply with this proposed AD:

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Modification and re-identification of (or re- placement of) FCD, and Installation of Placard.		\$4,865	\$5,460	\$2,184,000

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

- 1. Is not a "significant regulatory action" under Executive Order 12866;
- 2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);

- 3. Will not affect intrastate aviation in Alaska; and
- 4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(h), 40113, 44701.

§39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Airbus: Docket No. FAA-2016-9074; Directorate Identifier 2016-NM-097-AD.

(a) Comments Due Date

We must receive comments by November 10, 2016.

(b) Affected ADs

None.

(c) Applicability

This AD applies to the Airbus airplanes, certificated in any category, identified in paragraphs (c)(1) through (c)(4) of this AD, all manufacturer serial numbers.

- (1) Airbus Model A318-111 and -112 airplanes.
- (2) Airbus Model A319–111, –112, –113, –114, and –115 airplanes.
- (3) Airbus Model A320–211, –212, and –214 airplanes.
- (4) Airbus Model A321–111, –112, –211, –212, and –213 airplanes.

(d) Subject

Air Transport Association (ATA) of America Code 71, Powerplant.

(e) Reason

This AD was prompted by reports of engine fan cowl door (FCD) losses on airplanes equipped with CFM56 engines due to operator failure to close the FCD during ground operations. We are issuing this AD to prevent in-flight loss of an engine FCD and possible consequent damage to the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Modification of Affected Fan Cowl Doors and Placard Installation

Within 35 months after the effective date of this AD, accomplish concurrently the actions in paragraphs (g)(1), (g)(2) and (g)(3) of this AD, in accordance with the Accomplishment Instructions of Airbus Service Bulletin A320–71–1068, Revision 01, dated April 28, 2016.

- (1) Modify the left-hand and right-hand FCDs on engines 1 and 2.
- (2) Install a placard on the box located at the bottom of the 120 volt unit (120 VU) panel, or at the bottom of the coat stowage, as applicable to airplane configuration.
- (3) Re-identify both the left-hand and right-hand FCDs with the new part number, as applicable, as specified in table 1 to paragraphs (g), (h), (i), and (k) of this AD.

TABLE 1 TO PARAGRAPHS (g), (h), (i), AND (k) OF THIS AD—FAN COWL DOOR PART NUMBER (P/N) CHANGE

Door position	Old P/N	New P/N
Left-hand Side—CFM56-5A		238M0301-501
	238-0301-503	238M0301-503
	238-0301-505	238M0301-505
	238-0301-507	238M0301-507
	238-0301-511	238M0301-511
	238-0301-513	238M0301-513
	238-0301-515	238M0301-515
	238-0301-517	238M0301-517
	238-0301-519	238M0301-519
	238-0301-521	238M0301–521
	238-0301-523	238M0301–523
	238-0301-525	238M0301–525
	238-0301-527	238M0301–527
	238-0301-529	238-0301-533
	238-0301-531	238-0301-535
Right-hand Side—CFM56-5A		238M0302–501
Tight hand dide of wide of the	238-0302-503	238M0302–503
	238-0302-505	238M0302-505
	238-0302-509	238M0302-509
	238-0302-511	238M0302-511
	238-0302-513	238M0302-513
	238-0302-515	238M0302-515
	238-0302-517	238M0302-517
	238-0302-519	238M0302-519
	238-0302-521	238M0302-521
	238-0302-523	238M0302-523
	238-0302-525	238M0302-525
	238-0302-527	238M0302-527
	238-0302-529	238M0302–529
	238-0302-531	238M0302–531
	238-0302-533	238M0302–533
	238-0302-535	238M0302–535
	238-0302-537	238M0302–537
	238-0302-539	238-0302-547
	238-0302-541	238-0302-549
	238-0302-543	238-0302-551
Left-hand Side—CFM56–5B	238-0302-545	238-0302-553
Left-hand Side—CFM56-5B		642M3001–503
	642–3001–505	642M3001-505
	642–3001–507	642–3001–511
Right-hand Side—CFM56-5B	642–3001–509	642–3001–513
Right-hand Side—CFM56-5B		642M3002–503
	642–3002–505	642M3002-505
	642–3002–507	642M3002-507
	642–3002–509	642M3002-509
	642–3002–511	642–3002–519
	642–3002–513	642–3002–521
	642–3002–515	642-3002-523
	642–3002–517	642-3002-525

(h) Replacement of Affected Fan Cowl Door With New Door Design

Replacing the FCDs, having a part number listed as "Old P/N" in table 1 to paragraphs (g), (h), (i), and (k) of this AD, with the FCDs having the corresponding part number listed as "New P/N" in table 1 to paragraphs (g), (h), (i), and (k) of this AD, is equal to compliance with paragraphs (g)(1) and (g)(3) of this AD. The replacement must be done in accordance with instructions approved by the Manager, International Branch, ANM—116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA), or approved under Airbus's EASA Design Organization Approval (DOA).

(i) Compliance Information for Airplanes on Which Airbus Modification 157517 Is Embodied

An airplane on which Airbus modification 157517 has been embodied in production, is compliant with the requirements of paragraphs (g)(1) and (g)(3) of this AD, provided it is determined that no FCD, having a part number identified as "Old P/N" in table 1 to paragraphs (g), (h), (i), and (k) of this AD, are installed on that airplane at the effective date of the AD.

(j) Compliance Information for Airplanes on Which Airbus Modification 157519 or Modification 157521 is Embodied

An airplane on which Airbus modification 157519 or modification 157521 has been embodied in production is compliant with the requirements of paragraph (g)(2) of this AD.

(k) Parts Installation Prohibition

As of the effective date of this AD, do not install on any airplane an FCD part number identified as "Old P/N" in table 1 to paragraphs (g), (h), (i), and (k) of this AD.

(l) Installation of Approved Parts

Installation on an airplane of right-hand and left-hand FCD, having a part number approved after the effective date of this AD, is equal to compliance with the requirements of paragraphs (g)(1) and (g)(3) of this AD for that airplane only, provided the conditions specified in paragraphs (l)(1) and (l)(2) of this AD are met.

(1) The part number must be approved by the Manager, International Branch, ANM—

116, Transport Airplane Directorate, FAA; or EASA, or approved under Airbus's EASA

(2) The FCD installation must be accomplished in accordance with airplane modification instructions approved by the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA; or EASA; or approved under Airbus's EASA DOA.

(m) Credit for Previous Actions

This paragraph provides credit for actions required by paragraph (g) of this AD, if those actions were performed before the effective date of this AD using Airbus Service Bulletin A320–71–1068, dated December 18, 2015, which is not incorporated by reference in this AD.

(n) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Sanjay Ralhan, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone 425-227-1405; fax 425-227-1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) Contacting the Manufacturer: For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM—116, Transport Airplane Directorate, FAA; or the EASA; or Airbus's EASA DOA. If approved by the DOA, the approval must include the DOA-authorized signature.

(o) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA Airworthiness Directive 2016–0069, dated April 11, 2016, for related information. This MCAI may be found in the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA– 2016–9074.

(2) For service information identified in this AD, contact Airbus, Airworthiness Office—EIAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet http://www.airbus.com. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For

information on the availability of this material at the FAA, call 425–227–1221.

Issued in Renton, Washington, on August 31, 2016.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 2016–21703 Filed 9–23–16; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF THE TREASURY

Internal Revenue Service

26 CFR Parts 1 and 301

[REG-131418-14]

RIN 1545-BN27

Reporting for Qualified Tuition and Related Expenses; Education Tax Credits; Correction

AGENCY: Internal Revenue Service (IRS), Treasury.

ACTION: Correction to a notice of proposed rulemaking and notice of public hearing.

SUMMARY: This document contains corrections to a notice of proposed rulemaking and notice of public hearing (REG-131418-14) that was published in the **Federal Register** on Tuesday, August 2, 2016 (81 FR 50657). The proposed regulations that revise the rules for reporting qualified tuition and related expenses under section 6050S on a Form 1098-T, "Tuition Statement," and conforms the regulations to the changes made to section 6050S by the Protecting Americans from Tax Hikes Act of 2015.

DATES: Written or electronic comments and request for a public hearing for the notice of proposed rulemaking at 81 FR 50657, August 2, 2016, are still being accepted and must be received by October 31, 2016.

FOR FURTHER INFORMATION CONTACT:

Concerning the proposed regulations, Gerald Semasek of the Office of Associate Chief Counsel (Procedure and Administration) for the proposed regulations under sections 6050S and 6724, (202) 317–6845, and Sheldon Iskow of the Office of Associate Chief Counsel (Income Tax and Accounting) for the proposed regulations under section 25A, (202) 317–4718; concerning the submission of comments and requests for a public hearing, Regina Johnson, (202) 317–6901 (not toll-free calls).

SUPPLEMENTARY INFORMATION:

Background

The notice of proposed rulemaking that is subject of this document is under section 6050S of the Internal Revenue Code

Need for Correction

As published, the notice of proposed rulemaking and notice of public hearing (REG-131418-14) contains errors that may prove to be misleading and are in need of clarification.

Correction of Publication

Accordingly, the notice of proposed rulemaking and notice of public hearing (REG-131418-14) that are subject to FR Doc. 2016-18032 are corrected as follows:

■ 1. On page 50662, in the preamble, second column, the third line from the bottom of the first full paragraph, the language "requiring eligible educational institution" is corrected to read "requiring eligible educational institutions".

§1.25A-0 [Corrected]

- 2. On page 50664, second column, amendatory instruction 2, the language "9. Revising the entry for § 1.25A—2(f)(6)." is corrected to read "9. Revising the entry for § 1.25A—5(f)(6).".
- 3. On page 50664, third column, entry for (e)(3), the language "Effective/applicability dates." is corrected to read "Applicability dates.".
- 4. On page 50664, third column, entry for (f)(4), the language "Effective/applicability date." is corrected to read "Applicability date.".
- 5. On page 50664, third column, entry for (e), the language "Effective/applicability date." is corrected to read "Applicability date.".
- 6. On page 50664, third column, entry for (g), the language "Effective/ applicability date." is corrected to read "Applicability date.".

§1.6050S-0 [Corrected]

■ 7. On page 50667, second column, entry for (c)(1)(iii)(E), the language "consequences of refunds, reimbursements." is corrected to read "consequences of refunds, reimbursements,".

§1.6050S-1 [Corrected]

■ 8. On page 50669, third column, in the second line of paragraph (b)(2)(vii), Example 5. (i), the language, "2016 fall semester" is corrected to read "Z as a full-time student for the 2016 fall semester"; and in the tenth the language, "\$11,000 for the 2017 spring semesters." is corrected to read "\$11,000 for the 2017 spring semester.".