

with the procedures specified in paragraph (o)(2) of this AD.

#### (l) No Reporting Requirement

Although SB A320–52–1171 R01 specifies to submit certain information to the manufacturer, and specifies that action as “RC” (Required for Compliance), this AD does not include that requirement.

#### (m) Credit for Previous Actions

This paragraph provides credit for the actions required by paragraphs (h) and (i) of this AD, if those actions were performed before the effective date of this AD using Airbus Service Bulletin A320–52–1171, dated October 29, 2015, provided that it can be conclusively determined that any part number D52371000018 was also inspected as specified in paragraph (h) of this AD.

#### (n) Parts Installation Limitation

As of the effective date of this AD, no person may install, on any airplane, an affected door specified in paragraph (g) of this AD, unless it has been inspected in accordance with the requirements of paragraph (h) of this AD and all applicable corrective actions have been done in accordance with paragraph (i) of this AD.

#### (o) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Branch, ANM–116, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Branch, send it to the attention of the person identified in paragraph (p)(2) of this AD. Information may be emailed to: [9-ANM-116-AMOC-REQUESTS@faa.gov](mailto:9-ANM-116-AMOC-REQUESTS@faa.gov). Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA; or EASA; or Airbus’s EASA DOA. If approved by the DOA, the approval must include the DOA-authorized signature.

(3) *Required for Compliance (RC)*: Except as specified in paragraphs (k) and (l) of this AD: If any service information contains procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator’s maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an

airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

#### (p) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA AD 2016–0187, dated September 19, 2016, for related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2017–0478.

(2) For more information about this AD, contact Sanjay Ralhan, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone: 425–227–1405; fax: 425–227–1149.

(3) For service information identified in this AD, contact Airbus, Airworthiness—EIAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email [account.airworth-eas@airbus.com](mailto:account.airworth-eas@airbus.com); Internet <http://www.airbus.com>. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

Issued in Renton, Washington, on May 12, 2017.

**Michael Kaszycki,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2017–10138 Filed 5–19–17; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA–2017–0475; Directorate Identifier 2016–NM–142–AD]

**RIN 2120–AA64**

#### Airworthiness Directives; Dassault Aviation Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for all Dassault Aviation Model FALCON 2000 and FALCON 2000EX airplanes. This proposed AD was prompted by reports of ice accretion on the airplane wing due to the failure of certain anti-ice piccolo tubes in the wing outboard slats. This proposed AD would require repetitive inspections of each anti-ice piccolo tube and corrective action if necessary. This proposed AD also provides an optional terminating action for the repetitive inspections. We are

proposing this AD to address the unsafe condition on these products.

**DATES:** We must receive comments on this proposed AD by July 6, 2017.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202–493–2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact Dassault Falcon Jet Corporation, Teterboro Airport, P.O. Box 2000, South Hackensack, NJ 07606; telephone 201–440–6700; Internet <http://www.dassaultfalcon.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2017–0475; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone 800–647–5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Tom Rodriguez, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone 425–227–1137; fax 425–227–1149.

#### SUPPLEMENTARY INFORMATION:

##### Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA–2017–0475; Directorate Identifier 2016–NM–142–AD” at the beginning of

your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA Airworthiness Directive 2016–0149, dated July 25, 2016 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all Dassault Aviation Model FALCON 2000 and FALCON 2000EX airplanes. The MCAI states:

Occurrences were reported of ice accretion on the wing, due to failure of the affected anti-ice piccolo tubes Part Number (P/N) FGFB725102. Investigation results indicated that some wing piccolo tubes P/N FGFB725102 could have manufacturing defects in their welded parts, which may

have caused the rupture of the tubes, due to fatigue.

This condition, if not detected and corrected, could lead to undetected significant ice accretion on the wing, possibly resulting in loss of control of the aeroplane.

To address this potential unsafe condition, [Dassault Aviation] DA issued Service Bulletin (SB) F2000–431 Revision 1 and SB F2000EX–391 Revision 1 (hereafter referred to collectively as ‘the applicable SB’ in this [EASA] AD) to provide instructions for endoscopic inspection of the tubes.

For the reasons described above, this [EASA] AD requires repetitive inspections of each wing outboard slat piccolo tube [for discrepancies, *i.e.*, manufacturing defects, cracking, and loss of material in the welded parts] and, depending on findings, replacement of the piccolo tube(s) [and the outboard slat] with a [new or] serviceable part.

You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2017–0475.

**Related Service Information Under 1 CFR Part 51**

Dassault Aviation issued Service Bulletin F2000–431, Revision 1, dated June 6, 2016; and Service Bulletin F2000EX–391, Revision 1, dated June 6, 2016. The service information describes procedures for endoscopic inspections

of the anti-ice piccolo tube on each wing outboard slat, and replacement or re-identification of affected anti-ice piccolo tubes and outboard slats. These documents are distinct since they apply to different airplane models. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

**FAA’s Determination and Requirements of This Proposed AD**

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

**Costs of Compliance**

We estimate that this proposed AD affects 348 airplanes of U.S. registry.

We estimate the following costs to comply with this proposed AD:

ESTIMATED COSTS

| Action           | Labor cost   | Parts cost | Cost per product            | Cost on U.S. operators          |
|------------------|--|------------|-----------------------------|---------------------------------|
| Inspection ..... | 6 work-hours × \$85 per hour = \$510 per inspection cycle. | \$0        | \$510 per inspection cycle. | \$177,480 per inspection cycle. |

We have received no definitive data that will enable us to provide cost estimates for the on-condition actions specified in this proposed AD.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority

because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and

4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**The Proposed Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**Dassault Aviation:** Docket No. FAA–2017–0475; Directorate Identifier 2016–NM–142–AD.

**(a) Comments Due Date**

We must receive comments by July 6, 2017.

**(b) Affected ADs**

None.

**(c) Applicability**

This AD applies to Dassault Aviation Model FALCON 2000 and FALCON 2000EX

airplanes, certificated in any category, all serial numbers.

**(d) Subject**

Air Transport Association (ATA) of America Code 30, Ice and Rain Protection.

**(e) Reason**

This AD was prompted by reports of ice accretion on the airplane wing due to the failure of certain anti-ice piccolo tubes in the wing outboard slats. We are issuing this AD to detect and correct manufacturing defects in the anti-ice piccolo tubes in the wing outboard slats. This condition could lead to undetected significant ice accretion on a wing, resulting in loss of control of the airplane.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Affected Anti-Ice Piccolo Tubes**

(1) For the purpose of this AD, an affected anti-ice piccolo tube meets at least one of the conditions specified in paragraphs (g)(1)(i) and (g)(1)(ii) of this AD.

(i) Has part number (P/N) FGFB725102 (left-hand side (LH)) or P/N FGFB726102 (right-hand side (RH)).

(ii) Is installed on a wing outboard slat having a part number identified in table 1 to paragraph (g)(1)(ii) of this AD.

TABLE 1 TO PARAGRAPH (g)(1)(ii) OF THIS AD—AFFECTED OUTBOARD SLATS PART NUMBERS

| LH  | RH                                  |
|---|-------------------------------------|
| FGFB134 .....                                 | FGFB144.                            |
| FGFB134A1 to FGFB134A9 inclusive .....        | FGFB144A1 to FGFB144A9 inclusive.   |
| FGFB134B1 .....                               | FGFB144B1.                          |
| FFGFB134C1 to FGFB134C4 inclusive .....       | FGFB144C1 to FGFB144C4 inclusive.   |
| From FGFB134D1 to FGFB134D4 inclusive .....   | FGFB144D1 to FGFB144D4 inclusive.   |
| FGFB135 and FGFB135M .....                    | FGFB145 and FGFB145M.               |
| FGFB135A1 to FGFB135A4 inclusive .....        | FGFB145A1 to FGFB145A4 inclusive.   |
| From FGFB135A1M to FGFB135A4M inclusive ..... | FGFB145A1M to FGFB145A4M inclusive. |
| From FGFB135B1 to FGFB135B3 inclusive .....   | FGFB145B1 to FGFB145B3 inclusive.   |
| FGFB135B1M to FGFB135B3M inclusive .....      | FGFB145B1M to FGFB145B3M inclusive. |
| F2MB135 .....                                 | F2MB145.                            |
| F2MB135A1 .....                               | F2MB145A1.                          |
| F2MB135L1 to F2MB135L5 inclusive .....        | F2MB145L1 to F2MB145L5 inclusive.   |

(2) If the outboard slat part number is identified in table 2 to paragraph (g)(2) of this AD, the anti-ice piccolo tube is not affected

because the outboard slat has already been retrofitted with a new stiffened anti-ice

piccolo tube, and no action is required by this AD for that piccolo tube.

TABLE 2 TO PARAGRAPH (g)(2) OF THIS AD—SERVICEABLE OUTBOARD SLATS PART NUMBERS

| LH  | RH                                       |
|---|--|
| FGFB134P .....                                  | FGFB144P.                                |
| FGFB134A1P through FGFB134A9P inclusive .....   | FGFB144A1P through FGFB144A9P inclusive. |
| FGFB134B1P .....                                | FGFB144B1P.                              |
| FFGFB134C1P to FGFB134C4P inclusive .....       | FGFB144C1P to FGFB144C4P inclusive.      |
| From FGFB134D1P to FGFB134D4P inclusive .....   | FGFB144D1P to FGFB144D4P inclusive.      |
| FGFB135P and FGFB135MP .....                    | FGFB145P and FGFB145MP.                  |
| FGFB135A1P to FGFB135A4P inclusive .....        | FGFB145A1P to FGFB145A4P inclusive.      |
| From FGFB135A1MP to FGFB135A4MP inclusive ..... | FGFB145A1MP to FGFB145A4MP inclusive.    |
| From FGFB135B1P to FGFB135B3P inclusive .....   | FGFB145B1P to FGFB145B3P inclusive.      |
| FGFB135B1MP to FGFB135B3MP inclusive .....      | FGFB145B1MP to FGFB145B3MP inclusive.    |
| F2MB135P .....                                  | F2MB145P.                                |
| F2MB135A1P .....                                | F2MB145A1P.                              |
| F2MB135L1P to F2MB135L5P inclusive .....        | F2MB145L1P to F2MB145L5P inclusive.      |
| F2MB135L6 to F2MB135L7 inclusive .....          | F2MB145L6 to F2MB145L7 inclusive.        |

**(h) Inspections**

If an anti-ice piccolo tube has been determined to be affected, as specified in paragraph (g) of this AD: At the applicable time specified in table 3 to paragraph (h) of this AD, do an endoscopic inspection for

discrepancies, *i.e.*, manufacturing defects, cracking, and loss of material in the welded parts of each affected anti-ice piccolo tube, in accordance with the Accomplishment Instructions of Dassault Service Bulletin F2000–431, Revision 1, dated June 6, 2016; or Service Bulletin F2000EX–391, Revision 1,

dated June 6, 2016; as applicable. Repeat the endoscopic inspection thereafter at intervals not to exceed those specified in table 3 to paragraph (h) of this AD, until the modification specified in paragraph (j) of this AD is done.

TABLE 3 TO PARAGRAPH (h) OF THIS AD—COMPLIANCE TIMES FOR INSPECTIONS

| Airplane model                | Initial inspection   | Repetitive inspection intervals |
|-------------------------------|--|---------------------------------|
| FALCON 2000 airplanes .....   | Prior to exceeding 2,000 flight cycles since the airplane's first flight, or within 1,000 flight cycles after the effective date of this AD, whichever occurs later. | 2,000 flight cycles.            |
| FALCON 2000EX airplanes ..... | Prior to exceeding 1,000 flight cycles since the airplane's first flight, or within 500 flight cycles after the effective date of this AD, whichever occurs later.   | 1,000 flight cycles.            |

**(i) Corrective Action**

If any discrepancy is found during any inspection required by paragraph (h) of this AD: Before further flight, replace the affected anti-ice piccolo tube with a new or serviceable part, and replace or re-identify the affected wing outboard slat as applicable, in accordance with the Accomplishment Instructions of Dassault Service Bulletin F2000-431, Revision 1, dated June 6, 2016; or Service Bulletin F2000EX-391, Revision 1, dated June 6, 2016; as applicable.

**(j) Optional Terminating Action**

Modification of an airplane by installing a new or serviceable anti-ice piccolo tube, and replacing or re-identifying the affected wing outboard slat, terminates the repetitive inspections required by paragraph (h) of this AD, if done in accordance with the Accomplishment Instructions of Dassault Service Bulletin F2000-431, Revision 1, dated June 6, 2016; or Service Bulletin F2000EX-391, Revision 1, dated June 6, 2016; as applicable.

**(k) Parts Installation Prohibition**

As of the time specified in paragraph (k)(1) or (k)(2) of this AD, as applicable, no person may install on any airplane an affected anti-ice piccolo tube or an affected outboard slat.

(1) For an airplane that, on the effective date of this AD, has an affected anti-ice piccolo tube or an affected outboard slat installed: Before further flight after modification of that airplane as required by paragraph (i) of this AD.

(2) For an airplane that, on the effective date of this AD, does not have an affected anti-ice piccolo tube or an affected outboard slat installed: As of the effective date of this AD.

**(l) Later-Approved Parts**

Installation on an airplane of an anti-ice piccolo tube having a part number approved after the effective date of this AD is acceptable for compliance with the requirements of paragraph (i) or paragraph (j) of this AD, as applicable, provided the conditions in paragraphs (l)(1) and (l)(2) of this AD are met.

(1) The anti-ice piccolo tube part number must be approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA); or Dassault Aviation's EASA Design Organization Approval (DOA).

(2) The installation of the anti-ice piccolo tube must be accomplished in accordance with a method approved by the Manager, International Branch, ANM-116, Transport

Airplane Directorate, FAA; or the EASA; or Dassault Aviation's EASA DOA.

**(m) Other FAA AD Provisions**

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to the attention of the person identified in paragraph (n)(2) of this AD. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA; or the EASA; or Dassault Aviation's EASA DOA. If approved by the DOA, the approval must include the DOA-authorized signature.

**(n) Related Information**

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA AD 2016-0149, dated July 25, 2016, for related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0475.

(2) For more information about this AD, contact Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone 425-227-1137; fax 425-227-1149.

(3) For service information identified in this AD, contact Dassault Falcon Jet Corporation, Teterboro Airport, P.O. Box 2000, South Hackensack, NJ 07606; telephone 201-440-6700; Internet <http://www.dassaultfalcon.com>. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Issued in Renton, Washington, on May 11, 2017.

**Michael Kaszycki,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2017-10135 Filed 5-19-17; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

**[Docket No. FAA-2017-0480; Directorate Identifier 2016-NM-204-AD]**

**RIN 2120-AA64**

**Airworthiness Directives; Airbus Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for certain Airbus Model A300 B4-600, B4-600R, and F4-600R series airplanes, and Model A300 C4-605R Variant F airplanes (collectively called Model A300-600 series airplanes); and Model A310 series airplanes. This proposed AD was prompted by cracking in the door sill area of the aft cargo door. This proposed AD would require repetitive inspections of the aft cargo door lower torsion box area, and corrective actions if necessary. We are proposing this AD to address the unsafe condition on these products.

**DATES:** We must receive comments on this proposed AD by July 6, 2017.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202-493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room