

FIGURE 1 TO PARAGRAPH (g)—SPEY 506–14A HIGH-PRESSURE COMPRESSOR (HPC) STAGE 12 ROTOR DISK MAXIMUM APPROVED LIFE

	Flight cycles
HPC stage 12 rotor disk, P/N EU25917, EU56963, and JR10242 .....	14,700

FIGURE 2 TO PARAGRAPH (g)—SPEY 555–15, SPEY 555–15H, SPEY 555–15N, OR SPEY 555–15P HPC STAGE 12 ROTOR DISK, P/N EU25917, EU56963, JR10242, AND JR18449, MAXIMUM APPROVED LIFE

Take-off monitoring procedure	Maximum approved lives (flight cycles)
(A) With no high-pressure (HP) revolutions per minute (RPM) monitoring .....	11,500
HP RPM monitoring; stated RPM not exceeded on more than 15% of occasions:	
(B) 100% N2 .....	13,600
(C) 99% N2 .....	17,100
(D) 98% N2 .....	19,300
(E) 97% N2 .....	20,500
(F) No HP RPM monitoring required Datum (Average N2 at 99.5%) .....	16,800

FIGURE 3 TO PARAGRAPH (g)—SPEY 555–15, SPEY 555–15H, SPEY 555–15N, OR SPEY 555–15P HPC STAGE 12 ROTOR DISK, P/N EU25917, EU56963, JR10242, AND JR18449, IN-SERVICE REPLACEMENT LIMITS

Take-off monitoring procedure	In-service replacement limits (flight cycles)
(A) With no HP RPM monitoring .....	13,800
HP RPM monitoring; stated RPM not exceeded on more than 15% of occasions:	
(B) 100% N2 .....	15,600
(C) 99% N2 .....	17,600
(D) 98% N2 .....	19,700
(E) 97% N2 .....	22,100
(F) No HP RPM monitoring required Datum (Average N2 at 99.5%) .....	17,300

**(h) Installation Prohibition**

After the effective date of this AD, installation of a serviceable spare engine or release to service of an engine after any shop visit is allowed, provided the affected part has not exceeded the Maximum Approved Lives in Figures 1 or 2 to paragraph (g) of this AD.

**(i) Definition**

For the purpose of this AD, a shop visit is the induction of an engine into the shop for maintenance or overhaul. The separation of engine flanges solely for the purpose of transporting the engine without subsequent engine maintenance does not constitute an engine shop visit.

**(j) Alternative Methods of Compliance (AMOCs)**

The Manager, Engine Certification Office, FAA, may approve AMOCs for this AD. Use the procedures found in 14 CFR 39.19 to make your request. You may email your request to: [ANE-AD-AMOC@faa.gov](mailto:ANE-AD-AMOC@faa.gov).

**(k) Related Information**

(1) For more information about this AD, contact Robert Green, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 1200 District Avenue, Burlington, MA 01803; phone: 781–238–7754; fax: 781–238–7199; email: [robert.green@faa.gov](mailto:robert.green@faa.gov).

(2) Refer to MCAI European Aviation Safety Agency (EASA), AD 2017–0014, dated January 30, 2017, for more information. You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA–2017–0186.

(3) RRD Alert Non-Modification Service Bulletin Sp72–A1071, Revision 1, dated January 27, 2017, which is not incorporated by reference in this AD, can be obtained from RRD, using the contact information in paragraph (k)(4) of this AD.

(4) For RRD service information identified in this AD, contact Rolls-Royce Deutschland Ltd & Co KG, Eschenweg 11–15827 Dahlewitz, Blankenfelde-Mahlow, Germany; phone: +49 0 33–7086–1944; fax: +49 0 33–7086–3276.

(5) You may view this service information at FAA, Engine & Propeller Directorate, 1200 District Avenue, Burlington, MA 01803. For information on the availability of this material at the FAA, call 781–238–7125.

Issued in Burlington, Massachusetts, on May 9, 2017.

**Robert J. Ganley,**

*Acting Manager, Engine & Propeller Directorate, Aircraft Certification Service.*  
[FR Doc. 2017–10437 Filed 5–25–17; 8:45 am]

**BILLING CODE 4910–13–P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2017–0451; Directorate Identifier 2017–CE–015–AD; Amendment 39–18885; AD 2017–10–11]

**RIN 2120–AA64**

**Airworthiness Directives; Stemme AG Gliders**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule; request for comments.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for Stemme AG Model S10–VT gliders (type certificate previously held by Stemme GmbH & Co. KG). This AD results from mandatory continuing airworthiness information (MCAI) issued by the aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as certain propeller front transmission gear wheels having insufficient material strength because of improper heat treatment

during manufacturing. We are issuing this AD to require actions to address the unsafe condition on these products.

**DATES:** This AD is effective June 15, 2017.

We must receive comments on this AD by July 10, 2017.

**ADDRESSES:** You may send comments by any of the following methods:

- **Federal eRulemaking Portal:** Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- **Fax:** (202) 493-2251.
- **Mail:** U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.
- **Hand Delivery:** U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0451; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4165; fax: (816) 329-4090; email: [jim.rutherford@faa.gov](mailto:jim.rutherford@faa.gov).

#### SUPPLEMENTARY INFORMATION:

##### Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued AD No. 2017-0072-E, dated April 26, 2017 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

It has been determined that, following improper heat treatment during manufacture, the material strength of an identified batch of propeller front transmission gear wheels is insufficient.

This condition, if not corrected, can lead to failure of a gear wheel, with possible loss of the power transmission between the

engine and the propeller and, eventually, reduced control of the sailplane.

Prompted by these determinations, Stemme issued Service Bulletin (SB) P062-980010, providing instructions not to operate sailplanes equipped with affected parts.

For the reasons described above, this [EASA] AD requires replacement of affected gearboxes.

You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0451.

#### FAA’s Determination and Requirements of the AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with this State of Design Authority, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are issuing this AD because we evaluated all information provided by the State of Design Authority and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

#### FAA’s Determination of the Effective Date

An unsafe condition exists that requires the immediate adoption of this AD. The FAA has found that the risk to the flying public justifies waiving notice and comment prior to adoption of this rule because improper heat treatment of the propeller front transmission gear wheels could cause the gear wheel to fail. This failure could cause loss of power between the engine and the propeller, which could result in reduced control. Therefore, we determined that notice and opportunity for public comment before issuing this AD are impracticable and that good cause exists for making this amendment effective in fewer than 30 days.

#### Comments Invited

This AD is a final rule that involves requirements affecting flight safety, and we did not precede it by notice and opportunity for public comment. We invite you to send any written relevant data, views, or arguments about this AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2017-0451; Directorate Identifier 2017-CE-015-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this AD. We will consider all comments received by the closing date and may

amend this AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this AD.

#### Costs of Compliance

There are approximately 50 Stemme AG Model S10-VT gliders of U.S. registry but only 14 serial numbers of the part number (P/N) 11AG front gearbox. Therefore, the most gliders of U.S. registry affected would be 14. According to Stemme AG, there are only 2 of the affected gearboxes installed on gliders of U.S. registry.

At the issue date of this AD, no design solution is available to restore the airworthiness of the respective type design to a level corresponding to its approved type design specifications. Therefore, the FAA cannot determine the cost of returning the affected gliders to service.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

(1) Is not a “significant regulatory action” under Executive Order 12866,

(2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

##### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

**2017-10-11 Stemme AG:** Amendment 39-18885; Docket No. FAA-2017-0451; Directorate Identifier 2017-CE-015-AD.

##### (a) Effective Date

This airworthiness directive (AD) becomes effective June 15, 2017.

##### (b) Affected ADs

None.

##### (c) Applicability

This AD applies to Stemme AG Model S10-VT gliders (type certificate previously held by Stemme GmbH & Co. KG), all serial numbers, that are:

(1) Equipped with a front gearbox, part number (P/N) 11AG, with a serial number listed in table 1 to paragraph (c) of this AD; and

(2) are certificated in any category.

TABLE 1 TO PARAGRAPH (c) OF THIS AD—AFFECTED P/N 11AG (FRONT GEARBOX) S/NS

80058/0814	80065/0616
80059/0915	80066/0716
80060/0915	80067/0916
80061/1115	80068/1016
80062/1215	80069/0117
80063/0116	80070/0217
80064/0416	80071/0217

**Note 1 to paragraph (c) of this AD:** Page 2 of Stemme AG Service Bulletin No. P062-980010, dated April 21, 2017, provides a pictorial of where the serial number of the affected gearboxes are located.

#### (d) Subject

Air Transport Association of America (ATA) Code 61: Propellers/Propulsors.

#### (e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) issued by the aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as certain propeller front transmission gear wheels having insufficient material strength because of improper heat treatment during manufacturing. We are issuing this AD to prevent failure of the propeller front transmission gear wheels. This failure could cause loss of power between the engine and the propeller, which could result in reduced control.

#### (f) Actions and Compliance

Unless already done, do the following actions.

(1) Before further flight after June 15, 2017 (the effective date of this AD), replace the front gearbox following a method approved specifically for this AD by the FAA, Small Airplane Directorate. Contact the FAA using the information in paragraph (g)(1) of this AD to obtain FAA-approved repair instructions approved specifically for compliance with this AD and incorporate those instructions.

**Note 2 to paragraph (f)(1) of this AD:** At the issue date of this AD, no design solution is available to restore the airworthiness of the respective type design to a level corresponding to its approved type design specifications.

(2) As of June 15, 2017 (the effective date of this AD), do not install a front gear box listed in paragraph (c) of this AD.

#### (g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) **Alternative Methods of Compliance (AMOCs):** The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4165; fax: (816) 329-4090; email: [jim.rutherford@faa.gov](mailto:jim.rutherford@faa.gov). Before using any approved AMOC on any glider to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) **Airworthy Product:** For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

#### (h) Related Information

(1) Refer to MCAI European Aviation Safety Agency (EASA) AD No. 2017-0072-E, dated April 26, 2017, and Stemme AG Service Bulletin No. P062-980010, dated

April 21, 2017, for related information. You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0451.

(2) For information concerning this action, contact STEMME AG service information identified in this AD, contact STEMME AG, Flugplatzstrasse F2, Nr. 6-7, D-15344 Strausberg, Germany; telephone: +49 (0) 3341-3612-0, fax: +49 (0) 3341-3612-30; Internet: [www.stemme.com](http://www.stemme.com).

Issued in Kansas City, Missouri, on May 9, 2017.

**Melvin Johnson,**

*Acting Manager, Small Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2017-10402 Filed 5-25-17; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 71

[Docket No. FAA-2016-9511; Airspace Docket No. 16-ASW-20]

#### Amendment of Class D Airspace; Kingsville, TX

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** This action amends the legal description of Class D airspace extending up to 2,500 feet above the surface at Kingsville Naval Air Station (NAS), Kingsville, TX. This action is necessary due to the decommissioning of the Kingsville radio beacon (RBN), and cancellation of the RBN approach, and enhances the safety and management of instrument flight rules (IFR) operations at the airport. This action also updates the geographic coordinates of Kingsville Naval Air Station, Kingsville, TX, to coincide with the FAA's aeronautical database.

**DATES:** Effective 0901 UTC, September 14, 2017. The Director of the Federal Register approves this incorporation by reference action under Title 1, Code of Federal Regulations, part 51, subject to the annual revision of FAA Order 7400.11 and publication of conforming amendments.

**ADDRESSES:** FAA Order 7400.11A, Airspace Designations and Reporting Points, and subsequent amendments can be viewed online at [http://www.faa.gov/air\\_traffic/publications/](http://www.faa.gov/air_traffic/publications/). For further information, you can contact the Airspace Policy Group, Federal Aviation Administration, 800 Independence Avenue SW., Washington, DC 20591; telephone: (202) 267-8783. The Order is also available for inspection at the