

2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);

3. Will not affect intrastate aviation in Alaska; and

4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Bombardier, Inc.: Docket No. FAA–2018–0546; Product Identifier 2017–NM–171–AD.

(a) Comments Due Date

We must receive comments by August 6, 2018.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Bombardier, Inc., Model BD–700–1A10 and BD–700–1A11 airplanes, certificated in any category, serial numbers 9002 through 9770 inclusive, 9772 through 9781 inclusive, and 9998.

(d) Subject

Air Transport Association (ATA) of America Code 53, Fuselage.

(e) Reason

This AD was prompted by reports of multiple in-flight departures of the aft belly fairing access panels. We are issuing this AD to prevent in-flight departures of the aft belly fairing access panels, which could result in runway hazards or hazards to people on the ground.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Access Panel Modification

Within 15 months after the effective date of this AD, modify the aft belly fairing access panels by replacing the attachments, in

accordance with the Accomplishment Instructions of the applicable service information identified in paragraphs (g)(1) and (g)(2) of this AD.

(1) For Model BD–700–1A10 airplanes: Bombardier Service Bulletin 700–53–050, or 700–53–6008, both Revision 01, both dated December 16, 2016.

(2) For Model BD–700–1A11 airplanes: Bombardier Service Bulletin 700–1A11–53–025, or 700–53–5009, both Revision 01, both dated December 16, 2016.

(h) Credit for Previous Actions

This paragraph provides credit for actions required by paragraph (g) of this AD, if those actions were performed before the effective date of this AD using the applicable service information identified in paragraphs (h)(1) through (h)(4) of this AD.

(1) Bombardier Service Bulletin 700–1A11–53–025, dated July 14, 2016.

(2) Bombardier Service Bulletin 700–53–050, dated July 14, 2016.

(3) Bombardier Service Bulletin 700–53–5009, dated July 14, 2016.

(4) Bombardier Service Bulletin 700–53–6008, dated July 14, 2016.

(i) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, New York ACO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to ATTN: Program Manager, Continuing Operational Safety, FAA, New York ACO Branch, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516–228–7300; fax 516–794–5531. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, New York ACO Branch, FAA; or Transport Canada Civil Aviation (TCCA); or Bombardier Inc.’s TCCA Design Approval Organization (DAO). If approved by the DAO, the approval must include the DAO-authorized signature.

(j) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) Canadian Airworthiness Directive CF–2017–31, dated September 22, 2017, for related information. This MCAI may be found in the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2018–0546.

(2) For more information about this AD, contact Andrea Jimenez, Aerospace Engineer, Airframe and Mechanical Systems Section, FAA, New York ACO Branch, 1600 Stewart

Avenue, Suite 410, Westbury, NY 11590; telephone 516–228–7330; fax 516–794–5531.

(3) For service information identified in this AD, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514–855–5000; fax 514–855–7401; email thd.crj@aero.bombardier.com; internet <http://www.bombardier.com>. You may view this service information at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

Issued in Des Moines, Washington, on June 8, 2018.

Michael Kaszycki,

Acting Director, System Oversight Division, Aircraft Certification Service.

[FR Doc. 2018–13126 Filed 6–19–18; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2018–0556; Product Identifier 2018–NM–015–AD]

RIN 2120–AA64

Airworthiness Directives; Airbus Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for all Airbus Model A318 series; Model A319 series; Model A320 series; and Model A321–111, –112, –131, –211, –212, –213, –231, and –232 airplanes. This proposed AD was prompted by reports of multiple angle of attack (AoA) probe blockages. This proposed AD would require all elevator aileron computer (ELAC) units to be upgraded with new software, or replaced with upgraded units. We are proposing this AD to address the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by August 6, 2018.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202–493–2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

• *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact Airbus, Airworthiness Office—EIAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; internet <http://www.airbus.com>. You may view this service information at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

Examining the AD Docket

You may examine the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2018–0556; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone 800–647–5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Sanjay Ralhan, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone 206–231–3223; fax 206–231–3398.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA–2018–0556; Product Identifier 2018–NM–015–AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this NPRM. We will consider all comments received by the closing

date and may amend this NPRM based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this NPRM.

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2018–0007R1, dated January 19, 2018 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all Airbus Model A318 series; Model A319 series; Model A320 series; and Model A321–111, –112, –131, –211, –212, –213, –231, and –232 airplanes. The MCAI states:

Occurrences were reported on multiple Angle of Attack (AoA) probes blockages. Investigation results indicated the need for improved AoA monitoring in order to detect cases of AoA probe blockage.

This condition, if not corrected, could lead to undue activation of the AoA protection, reverting to manual control of the aeroplane, which, under specific circumstances, could result in reduced control of the aeroplane.

To address this potential unsafe condition, Airbus developed several Elevator Aileron Computer (ELAC) standards, *i.e.* ELAC units loaded with a specific software Part Number (P/N), and EASA issued AD 2017–0008, retaining part of the requirements of EASA AD 2015–0088R1 [which corresponds to FAA AD 2016–17–03, Amendment 39–18616 (81 FR 55358, August 19, 2016) (“AD 2016–17–03”)], which was superseded, and requiring an upgrade of all ELAC units with ELAC L99 standard, which introduces improvements in the AoA probe monitoring for Current Engine Option (CEO) aeroplanes, and also incorporates flight control aspects for New Engine Option (NEO) aeroplanes.

Since that [EASA] AD was issued, it was determined that clarification is necessary for the Parts Installation requirements, and some typographical (P/N) errors were detected. This [EASA] AD is revised accordingly.

You may examine the MCAI in the AD docket on the internet at <http://www.regulations.gov> by searching for

and locating Docket No. FAA–2018–0556.

Relationship of Proposed AD to AD 2016–17–03

This NPRM does not propose to supersede AD 2016–17–03. Rather, we have determined that a stand-alone AD would be more appropriate to address the changes in the MCAI. This proposed AD would require all ELAC units to be upgraded with new software, or replaced with upgraded units. Accomplishment of the proposed actions would then terminate all of the requirements of AD 2016–17–03.

Related Service Information Under 1 CFR Part 51

Airbus has issued Service Bulletin A320–27–1263, Revision 00, dated April 28, 2017; and Service Bulletin A320–27–1264, Revision 00, dated April 28, 2017. The service information describes the upgrade or replacement of ELAC units. These documents are distinct because they apply to different airplane configurations.

This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type designs.

Costs of Compliance

We estimate that this proposed AD affects 1,250 airplanes of U.S. registry.

We estimate the following costs to comply with this proposed AD:

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Modification	3 work-hours × \$85 per hour = \$255	Up to \$7,970	Up to \$8,225	Up to \$10,281,250

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue

rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII:

Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

This proposed AD is issued in accordance with authority delegated by the Executive Director, Aircraft Certification Service, as authorized by FAA Order 8000.51C. In accordance with that order, issuance of ADs is normally a function of the Compliance and Airworthiness Division, but during this transition period, the Executive Director has delegated the authority to issue ADs applicable to transport category airplanes to the Director of the System Oversight Division.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866;

2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);

3. Will not affect intrastate aviation in Alaska; and

4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Airbus: Docket No. FAA–2018–0556; Product Identifier 2018–NM–015–AD.

(a) Comments Due Date

We must receive comments by August 6, 2018.

(b) Affected ADs

This AD affects AD 2016–17–03, Amendment 39–18616 (81 FR 55358, August 19, 2016) (“AD 2016–17–03”).

(c) Applicability

This AD applies to the Airbus airplanes identified in paragraphs (c)(1) through (c)(4) of this AD, certificated in any category, all manufacturer serial numbers.

(1) Model A318–111, –112, –121, and –122 airplanes.

(2) Model A319–111, –112, –113, –114, –115, –131, –132, and –133 airplanes.

(3) Model A320–211, –212, –214, –216, –231, –232, –233, –251N, and –271N airplanes.

(4) Model A321–111, –112, –131, –211, –212, –213, –231, and –232 airplanes.

(d) Subject

Air Transport Association (ATA) of America Code 27, Flight Controls.

(e) Reason

This AD was prompted by reports of multiple angle of attack (AoA) probe blockages. We are issuing this AD to address the blockage of AoA probes. This condition, if not corrected, could lead to undue activation of the AoA protection, reverting to manual control of the airplane, which, under specific circumstances, could result in reduced control of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Definition of Affected Elevator Aileron Computer (ELAC)

For the purposes of this AD, ELAC units having a part number (P/N) listed in table 1 to paragraphs (g), (h), and (i) of this AD are hereafter referred to as “affected ELAC” in this AD.

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Table 1 to paragraphs (g), (h), and (i) of this AD – Affected ELAC Part Numbers

ELAC P/N	Designation	FIN
3945122202	ELAC A320-111 Type Def.	2 CE 1 / 2
3945122203	ELAC L50C	2 CE 1 / 2
3945122303	ELAC L50C	2 CE 1 / 2
3945122304	ELAC L60	2 CE 1 / 2
3945122305	ELAC L61B	2 CE 1 / 2
3945122306	ELAC L61F	2 CE 1 / 2
3945122307	ELAC L62C	2 CE 1 / 2
C12370AA01	ELAC L68C	2 CE 1 / 2
3945122501	ELAC L69	2 CE 1 / 2
3945122502	ELAC L69J	2 CE 1 / 2
3945122503	ELAC L77	2 CE 1 / 2
3945122504	ELAC L78	2 CE 1 / 2
3945122505	ELAC A L80	2 CE 1 / 2
3945123505	ELAC A' L80	2 CE 1 / 2
3945128101	ELAC B L80	2 CE 1 / 2
3945122506	ELAC A L81	2 CE 1 / 2
3945123506	ELAC A' L81	2 CE 1 / 2
3945128102	ELAC B L81	2 CE 1 / 2
3945122507	ELAC A L82	2 CE 1 / 2
3945123507	ELAC A' L82	2 CE 1 / 2
3945128103	ELAC B L82	2 CE 1 / 2
3945122608	ELAC A L83	2 CE 1 / 2
3945123608	ELAC A' L83	2 CE 1 / 2
3945122609	ELAC A L84	2 CE 1 / 2
3945123609	ELAC A' L84	2 CE 1 / 2
3945128204	ELAC B L90L	2 CE 1 / 2
3945128205	ELAC B L90N	2 CE 1 / 2
3945128206	ELAC B L91	2 CE 1 / 2
3945129101	ELAC B L91 data loadable	2 CE 1 / 2 SW1
3945128207	ELAC B L92	2 CE 1 / 2
3945128208	ELAC B L92L	2 CE 1 / 2
3945128209	ELAC B L93	2 CE 1 / 2
3945129103	ELAC B L93 data loadable	2 CE 1 / 2 SW1
3945128210	ELAC B L94	2 CE 1 / 2

ELAC P/N	Designation	FIN
3945129104	ELAC B L94 data loadable	2 CE 1 / 2 SW1
3945128212	ELAC B L96	2 CE 1 / 2
3945129106	ELAC B L96 data loadable	2 CE 1 / 2 SW1
3945129107	ELAC B L96 H-A data loadable	2 CE 1 / 2 SW1
3945128214	ELAC B L97	2 CE 1 / 2
3945129108	ELAC B L97 data loadable	2 CE 1 / 2 SW1
3945128215	ELAC B L97+	2 CE 1 / 2
3945129109	ELAC B L97+ data loadable	2 CE 1 / 2 SW1
3945128216	ELAC B L98	2 CE 1 / 2
3945129110	ELAC B L98 data loadable	2 CE 1 / 2 SW1

BILLING CODE 4910-13-C**(h) Required Actions**

For airplanes with ELAC part numbers listed in table 1 to paragraphs (g), (h), and (i) of this AD: Within the applicable compliance times defined in figure 1 to paragraph (h) of this AD, upgrade each ELAC by uploading L99 software part number (P/N) 3945129111 or by replacing the existing ELAC with ELAC L99 P/N 3945128217 in accordance with the Accomplishment Instructions of Airbus Service Bulletin A320-27-1263, Revision 00, dated April 28, 2017, or Airbus Service Bulletin A320-27-1264, Revision 00, dated April 28, 2017, as applicable, or in accordance with modification instructions approved by the Manager, International

Section, Transport Standards Branch, FAA; or the European Aviation Safety Agency (EASA); or Airbus's EASA Design Organization Approval (DOA); or in accordance with modification instructions approved by an EASA DOA (other than the Airbus EASA DOA), provided the conditions specified in paragraphs (h)(1) through (h)(4) of this AD are met. If approved by the DOA, the approval must include the DOA-authorized signature.

(1) Absence of electronic centralized aircraft monitoring (ECAM) warning or maintenance message related to ELAC, before the data-loadable ELAC unit is removed and software is loaded.

(2) The data-loadable ELAC unit is removed as specified in Airbus aircraft

maintenance manual (AMM) Task 27-93-34-000-001-A.

(3) The data-loadable ELAC unit is checked by two different means, either line replaceable unit (LRU) identification and label call up, or Alpha Call Up ELA 1 and ELA 2.

(4) After the software is loaded, the data-loadable ELAC unit is re-installed as specified in Airbus AMM Task 27-93-34-400-001-A.

Note 1 to paragraph (h) of this AD: Non-data-loadable ELAC L99 P/N 3945128217 units are fully interchangeable and mixable with data-loadable ELAC L99 P/N 3945129100 units with L99 software P/N 3945129111 loaded.

Figure 1 to paragraph (h) of this AD – Compliance Times

Airplanes (all models)	Compliance Time (after the effective date of this AD)
A318, A319, and A321	Within 24 months
Model A320 series airplanes	Within 36 months

(i) Parts Installation Prohibition

(1) For airplanes with ELAC units listed in table 1 to paragraphs (g), (h), and (i) of this AD: After modification of an airplane as required by paragraph (h) of this AD, do not install any affected ELAC on that airplane.

(2) For airplanes with ELAC units not listed in table 1 to paragraphs (g), (h), and (i) of this AD: From the effective date of this AD, do not install any affected ELAC on that airplane.

(j) Installation of Later Software Versions

Installation of an ELAC unit with a software standard above L99 is equal to compliance with the requirements of paragraph (h) of this AD, provided the conditions specified in paragraphs (j)(1) through (j)(2) of this AD are met.

(1) The ELAC unit part number is approved by the Manager, International Section,

Transport Standards Branch, FAA; or EASA; or Airbus's EASA DOA.

(2) The installation is accomplished in accordance with modification instructions approved by the Manager, International Section, Transport Standards Branch, FAA; or EASA; or Airbus's EASA DOA; or in accordance with modification instructions approved by an EASA DOA (other than the Airbus EASA DOA), provided the conditions in paragraphs (j)(2)(i) through (j)(2)(iv) of this AD are met.

(i) Absence of ECAM warning or maintenance message related to ELAC, before the data-loadable ELAC unit is removed and software is loaded.

(ii) The data-loadable ELAC unit is removed as specified in Airbus AMM Task 27-93-34-000-001-A.

(iii) The data-loadable ELAC unit is checked by two different means, either LRU

identification and label call up, or Alpha Call Up ELA 1 and ELA 2.

(iv) After the software is loaded, the data-loadable ELAC unit is re-installed as specified in Airbus AMM Task 27-93-34-400-001-A.

(k) Airplanes Not Affected by the Requirements of Paragraph (h) of This AD

(1) An airplane on which any modification (mod) specified in paragraphs (k)(1)(i) and (k)(1)(ii) of this AD was embodied in production is not affected by the requirements of paragraph (h) of this AD, provided it is determined that no affected ELAC is installed as of the effective date of this AD.

(i) Airbus mod 161843 (installation of data-loadable ELAC P/N 3945129100 unit with L99 software P/N 3945129111) or mod

159979 (installation of non-data-loadable ELAC L99 P/N 3945128217).

(ii) Airbus mod 160577 (installation of data-loadable ELAC P/N 3945129100 unit with L101 software P/N 3945129112) or mod 162042 (installation of non-data-loadable ELAC L101 P/N 3945128218).

(2) An airplane on which any modification specified in paragraphs (k)(2)(i), (k)(2)(ii), or (k)(2)(iii) of this AD was done is not affected by the requirements of paragraph (h) of this AD, provided it is determined that no affected ELAC is installed as of the effective date of this AD.

(i) A modification specified in Airbus Service Bulletin A320–27–1267, Revision 00, dated September 27, 2017 (ELAC L101 P/N 3945128218 non-data-loadable).

(ii) A modification specified in Airbus Service Bulletin A320–27–1268, Revision 00, dated September 27, 2017 (ELAC P/N 3945129100 data-loadable with L101 software P/N 3945129112 for A320 NEO).

(iii) A modification specified in Airbus Service Bulletin A320–27–1269, Revision 00, dated September 27, 2017 (ELAC P/N 3945129100 data-loadable with L101 software P/N 3945129112).

(l) Terminating Action for AD 2016–17–03

Accomplishing the actions required by paragraph (h) of this AD or complying with any method of compliance specified in paragraph (k) of this AD terminates all requirements of AD 2016–17–03.

(m) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Section, Transport Standards Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Section, send it to the attention of the person identified in paragraph (n)(2) of this AD. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Section, Transport Standards Branch, FAA; or EASA; or Airbus's EASA DOA. If approved by the DOA, the approval must include the DOA-authorized signature.

(n) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA AD 2018–0007R1, dated January 19, 2018, for related information. This MCAI may be found in the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2018–0556.

(2) For more information about this AD, contact Sanjay Ralhan, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone 206–231–3223; fax 206–231–3398.

(3) For service information identified in this AD, contact Airbus, Airworthiness Office—ELAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; internet <http://www.airbus.com>. You may view this service information at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. You may view this service information at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

Issued in Des Moines, Washington, on June 7, 2018.

Michael Kaszycki,

Acting Director, System Oversight Division, Aircraft Certification Service.

[FR Doc. 2018–12885 Filed 6–19–18; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF HOUSING AND URBAN DEVELOPMENT

24 CFR Part 100

[Docket No. FR–6111–A–01]

RIN 2529–ZA01

Reconsideration of HUD's Implementation of the Fair Housing Act's Disparate Impact Standard

AGENCY: Office of the Assistant Secretary for Fair Housing and Equal Opportunity, HUD.

ACTION: Advance notice of proposed rulemaking.

SUMMARY: This advance notice of proposed rulemaking (ANPR) invites public comment on possible amendments to HUD's 2013 final rule implementing the Fair Housing Act's disparate impact standard, as well as the 2016 supplement to HUD's responses to certain insurance industry comments made during the rulemaking. HUD is reviewing the final rule and supplement to determine what changes, if any, are appropriate following the Supreme Court's 2015 ruling in *Texas Department of Housing and Community Affairs v. Inclusive Communities Project, Inc.*, which held that disparate impact claims were cognizable under the Fair Housing Act and discussed standards for, and the constitutional limitations on, such claims. As HUD conducts its review, it is soliciting

public comment on the disparate impact standard set forth in the final rule and supplement, the burden-shifting approach, the relevant definitions, the causation standard, and whether changes to these or other provisions of the rule would be appropriate. HUD is also issuing this ANPR in response to public comments submitted on its May 15, 2017, **Federal Register** document seeking input on ineffective regulations and an October 26, 2017, recommendation from the Department of the Treasury.

DATES: *Comment Due Date:* August 20, 2018.

ADDRESSES: Interested persons are invited to submit comments to the Office of the General Counsel, Rules Docket Clerk, Department of Housing and Urban Development, 451 Seventh Street SW, Room 10276, Washington, DC 20410–0001. Communications should refer to the above docket number and title and should contain the information specified in the “Request for Comments” section. There are two methods for submitting public comments.

1. *Submission of Comments by Mail.* Comments may be submitted by mail to the Regulations Division, Office of General Counsel, Department of Housing and Urban Development, 451 7th Street SW, Room 10276, Washington, DC 20410–0500. Due to security measures at all federal agencies, however, submission of comments by mail often results in delayed delivery. To ensure timely receipt of comments, HUD recommends that comments submitted by mail be submitted at least two weeks in advance of the public comment deadline.

2. *Electronic Submission of Comments.* Interested persons may submit comments electronically through the Federal eRulemaking Portal at <http://www.regulations.gov>. HUD strongly encourages commenters to submit comments electronically. Electronic submission of comments allows the commenter maximum time to prepare and submit a comment, ensures timely receipt by HUD, and enables HUD to make comments immediately available to the public. Comments submitted electronically through the <http://www.regulations.gov> website can be viewed by other commenters and interested members of the public. Commenters should follow instructions provided on that site to submit comments electronically.

Note: To receive consideration as public comments, comments must be submitted through one of the two methods specified